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## U.S. DEPARTMENT OF COMMERCE National Technical Information Service

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F100 MULTIVARIABLE CONTROL SYSTEM ENGINE MODELS/DESIGN CRITERIA

PRATT AND WHITNEY AIRCRAFT GROUP WEST PALM BEACH, FLORIDA

NOVEMBER 1976

# FIOO MULTIVARIABLE CONTROL SYSTEM ENGINE MODELS /DESIGN CRITERIA

PRATT & WHITNEY AIRCRAFT GROUP GOVERNMENT PRODUCTS DIVISION UNITED TECHNOLOGIES CORPORATION P.O. BOX 2691 WEST PALM BEACH, FLORIDA 33402

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This report has been reviewed by the Information Office (ASD/OIP) and is releasable to the National Technical Information Service (NTIS). At NTIS, it will available to the general public, including foreign nations.

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This technical report has been reviewed and is approved for publication.

LESTER L. SMALL, GS-13

Project Engineer

FOR THE COMMANDER

Tech Area Manager, Controls

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A	facility at NASA Lewis Research Center. The F100 engine computer simulations, the control criteria for defining the basic requirements of a F100 control system, and a brief evaluation of the resulting LQR engine control system are presented in this report.
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## SECTION I

## INTRODUCTION AND SUMMARY

#### INTRODUCTION

Aircraft turbine engines have increased in complexity in recent years, and the engine cycles now being evaluated for future applications are even more complex. The older generation of engines in use today are basically fixed-geometry designs with only fuel flow and perhaps nozzle area as variables. Newer engines, like the Pratt & Whitney Aircraft F100, use variable geometry in the fan and/or compressor stators. Future engines, the so-called variable cycle engines, may also have variable turbine geometry.

The trend toward more complex engines has resulted in additional requirements for the control system. Control systems for older engines required the measurement of three or four parameters and the control of only one or two variables. Controls for future engines will require the measurement of many parameters (perhaps 10 to 20) to control a significant number of engine variables, (perhaps 6 to 10). Operational requirements have also become more stringent, necessitating increased control system capability in terms of accuracy and response. The use of closed-loop control to give better operation of the engine in accordance with performance and limiting requirements is replacing the scheduling type (open loop) controls used on older engines.

Classical control design techniques, which involve the evaluation of each controlling loop individually, have worked well for older, simpler engines. However, such techniques are cumbersome and time-consuming when applied to variable cycle-type engines; therefore, optimal control techniques using modern control theory are now being investigated. The linear quadratic regulator (LQR) is one specific area of modern control theory that appears to be suited to the engine control problem due to the emphasis on maintaining "optimum" engine performance in the presence of a wide variety of external disturbances (i.e., aircraft maneuvers, horsepower, and bleed extractions, etc.) and the requirement for fast engine power transients.

The objective of the F100 multivariable control research program is to extend the LQR theory to develop a "practical" control system that can operate a state-of-the-art gas turbine engine over its entire flight envelope. The engine selected for this program is a Pratt & Whitney Aircraft F100 afterburning turbofan. To determine the adequacy of the control synthesis effort, the resulting control logic will be incorporated into a digital computer/controller. This controller will then be evaluated in conjunction with a hybrid engine simulation. If successful, F100 engine tests will be conducted at a NASA Lewis Research Center altitude test facility.

## SUMMARY

The Air Force Aero Propulsion Laboratory and NASA Lewis Research Center have jointly sponsored this control development and evaluation effort. NASA Lewis will provide the digital computer/controller, engine hardware interfaces, the hybrid computer real-time engine simulation, engine altitude test facilities, and engineering manpower to fully evaluate the LQR control system. The Air Force provided contracts for two supporting contractors: Pratt & Whitney Aircraft, an engine manufacturer, and Systems Control, Inc. (Vt), a control research organization. These contractors, although separately funded, have integrated their efforts as outlined in Figure 1 to develop and evaluate a control system for the F100 engine using a modern control approach.

Pratt & Whitney Aircraft had prime responsibility to define the F100 engine via digital computer simulations, to assist NASA Lewis in establishing the necessary engine/computer interface hardware, to establish control criteria, to define the basic requirements of a F100 control system, and, in conjunction with the control contractor, to support the Government in their evaluation of the LQR engine control system.

Systems Control, Inc., had prime responsibility to develop the LQR control logic and demonstrate that this logic can: (1) successfully control the F100 dynamic digital engine simulation over its operating range, (2) be adequately programed on NASA's digital computer/controller, and (3) control both the hybrid computer real-time engine simulation and the F100 engine in an altitude test facility.

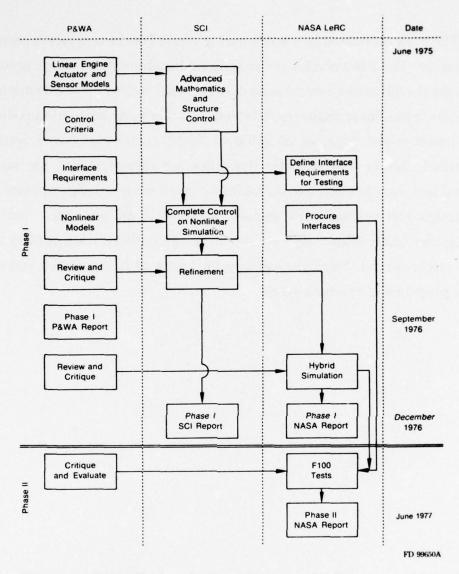


Figure 1. Total Engine Support Program Outline

The total program, as shown in Figure 1, is a 19-month Phase I effort, culminating in a hybrid computer evaluation of the control system, and a 6-month Phase II effort, concluding in a F100 engine test and evaluation in a NASA Lewis altitude test facility. At the conclusion of the Phase I effort, P&WA, SCI, and NASA will each report on the work conducted. Only NASA will report on the Phase II tests.

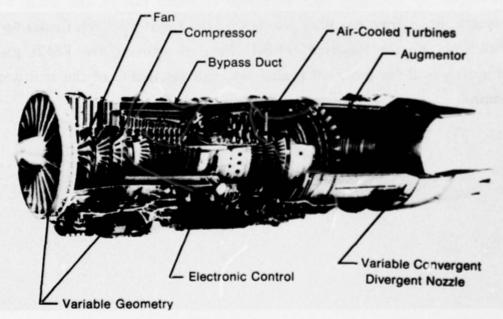
The work conducted by P&WA during Phase I of the contract is presented in this report. The F100 nonlinear dynamic engine simulation, which provides the basis for the advanced control logic development, is described. The method used to generate the linear engine models from this nonlinear simulation is discussed, and linear model data for 35 different flight points and power settings are presented. Sensor and actuator dynamics are discussed and the significant control features characterized for inclusion in the control design process. Criteria for design and evaluation of engine control logic are also presented. A brief description and preliminary evaluation of the multivariable control research logic are presented. Finally, directions for future control research and development programs are recommended.

## SECTION II

#### **F100 ENGINE DESCRIPTION**

The engine selected for this research program is a Pratt & Whitney Aircraft F100 afterburning turbofan, representative of current high-technology engines as illustrated in Figure 2. The F100-PW-100 is a low-bypass-ratio, twin-spool, axial-flow turbofan engine, consisting of the following components:

- · Three-stage fan driven by a two-stage turbine
- Ten-stage compressor driven by an aircooled two-stage turbine
- Main burner with an annular chamber
- Annular fan duct that surrounds the basic gas generator and discharges air in the mixed-flow augmentor
- Variable area nozzle



FD 39150B

Figure 2. F100 High Technology Engine

An inlet guide vane with a movable trailing edge to achieve variable airfoil camber is used ahead of the fan to improve inlet distortion tolerance and fan efficiency. The first three stators of the high compressor are variable to improve starting and high Mach number characteristics. Airflow bleed is extracted at the compressor exit for installation requirements and starting. The exhaust nozzle for the engine is a balance beam design with actuated divergent flap. The variable geometry of the balanced-beam nozzle enables all three nozzle performance parameters (nozzle area, expansion ratio, and boat-tail drag) to be simultaneously near optimum through the operating range. The fuel control is basically hydromechanical with an engine mounted digital electronic supervisory control. The engine has great potential for being adapted to an all-digital electronic control, which would then allow more versatile control and permit fully optimized engine operation at all flight conditions. In addition, an all-digital electronic control could accommodate the multivariable control logic to be developed in this program. This engine, being one of the most modern operational engines, has been selected by the Lewis Research Center for their Full-Scale Engine Research (FSER) Program. Among the FSER program objectives will be the F100 engine test and evaluation of the multivariable control.

#### SECTION III

#### F100 ENGINE DYNAMIC SIMULATION

#### **NONLINEAR DECK**

The F100 dynamic simulation used for this program consists of individual representations of the major engine components, such as the fan, compressor, burner, turbines, duct, and augmentor; with gas law relationships governing the component performance and interactions. The dynamic or time-varying relationships are based on natural laws of conservation of mass and energy. Calculations are made using the rotor inertial effects, enclosed volume capacitive effects, and transient heat transfer effects. Dynamic elements, such as integrators and first order time lags, are modeled with recursion formulas. The relationships and calculations combine to form a set of simultaneous, nonlinear differential equations. Figure 3 illustrates the F100 dynamic simulation gas path equations. The simulation also includes an option for using the current F100 engine control. The two primary components for controlling engine operation are the engine electronic control (EEC) and the unified fuel control (UFC). The UFC provides basic scheduling for primary and augmentor fuel flow and distribution, high compressor variable geometry settings (RCVV), compressor bleeds (BLC), and nozzle area setting (A<sub>j</sub>). The EEC provides trim to the unified control for aircraft/engine coordination. This is accomplished by trimming engine fuel flow (WFMB), fan inlet guide vanes (CIVV), and nozzle area based on feedback signals from the engine to maximize performance. Engine station designations are identified in Figure 4.

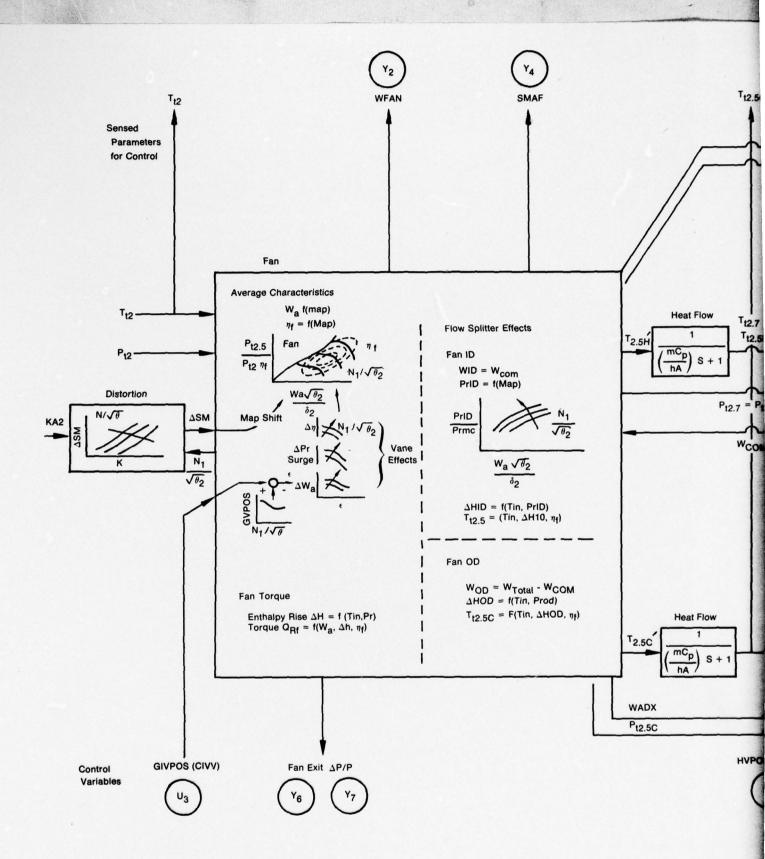
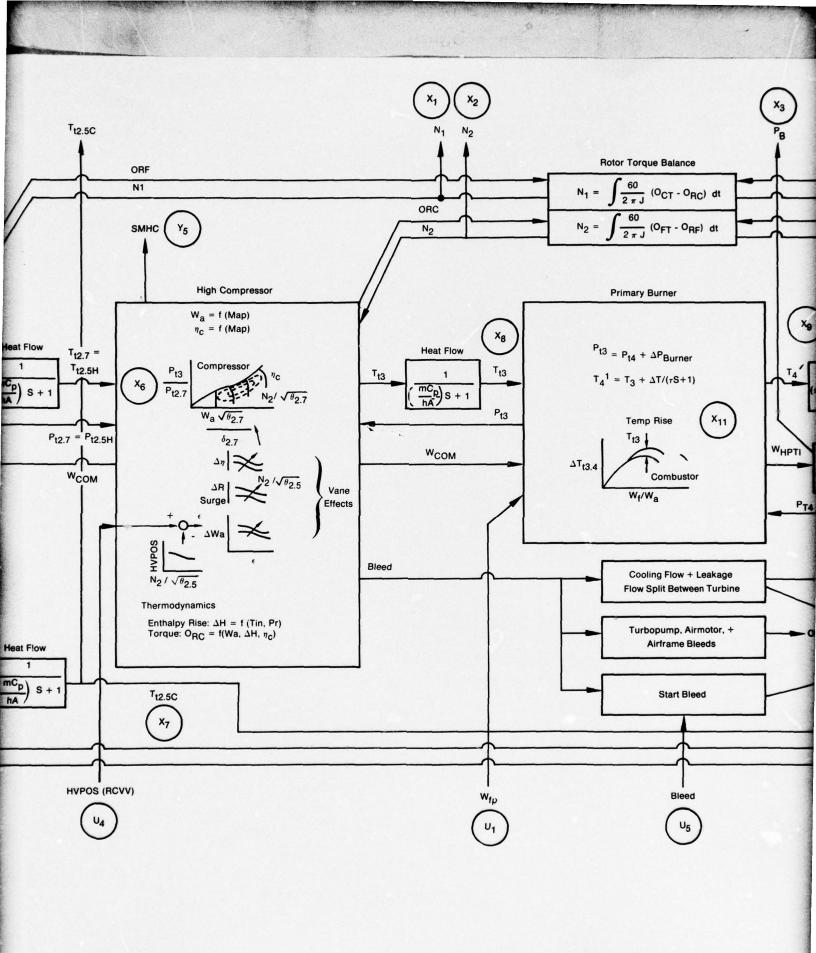
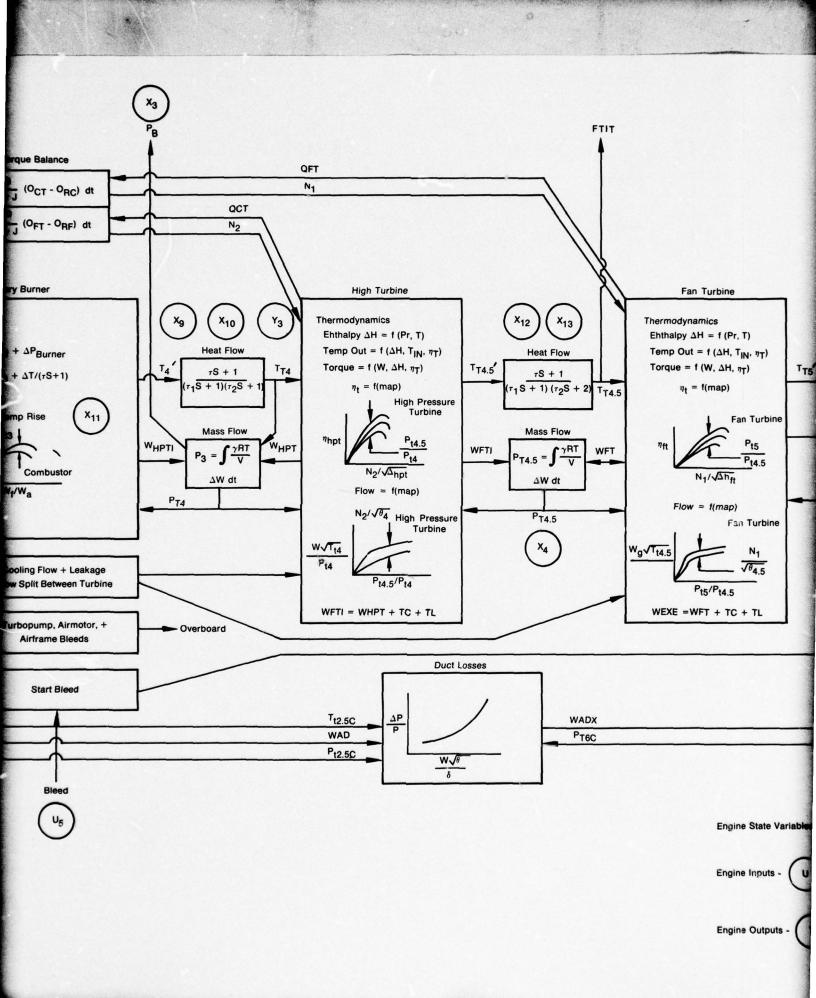
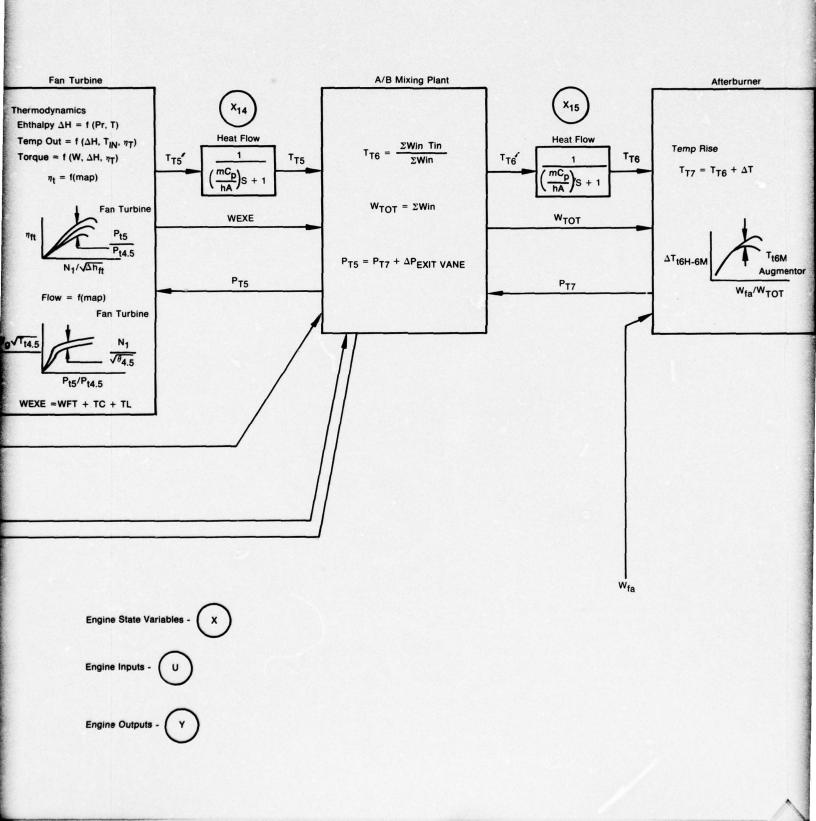
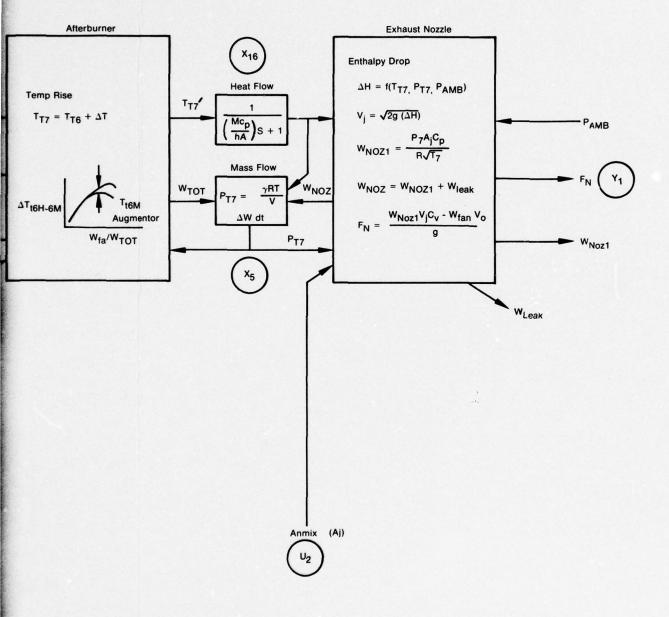


Figure 3. F100 Engine Dynamic Gas Path Equations









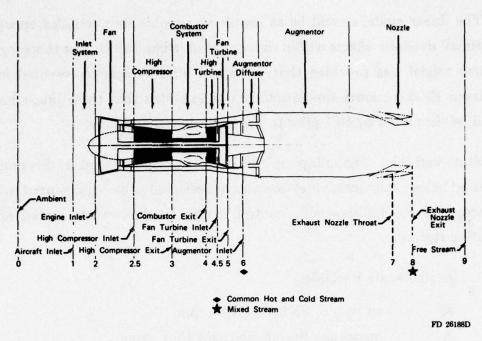
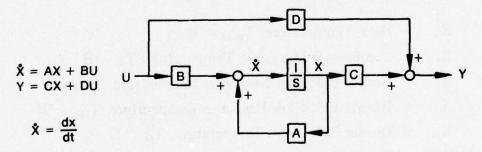


Figure 4. F100 Component and Station Identification

## LINEAR MODEL

The linear F100 model provided is of the form:



where X is the vector of state variables, such as pressures and rotor speeds,  $\dot{X}$  is the time derivative of the state vector, U is the vector of control inputs, such as fuel flow and exhaust nozzle area, and Y is the vector of observed parameters, such as thrust or airflow. "A" is the plant matrix and its elements are the partials from each state variable to each state variable time derivative. Elements of the output matrix "C" define the effect of each state variable on each observed variable. The control matrix "B" and the direct couple matrix "D" define the effect of each control variable on each state variable time derivative and each observed parameter, respectively.

The linear model should be as simple as possible, but complex enough to contain all dynamic effects within the controllable bandwidth. For this program, a linear model was provided that contained all 16 states represented in the nonlinear F100 dynamic simulation. It was expected that these linear models would be simplified by SCI prior to the control system design.

State variables, engine inputs, and engine outputs utilized in this program are listed below. Augmentor fuel flow was not selected as an engine input to limit the scope of the multivariable control design to the nonaugmented engine operating regimes.

## 1. Engine State Variables

 $X_1 = Fan Speed, SNFAN (N_1) - rpm$ 

 $X_2$  = Compressor Speed, SNCOM  $(N_2)$  - rpm

X<sub>3</sub> = Compressor Discharge Pressure, P<sub>t3</sub> - psia

X<sub>4</sub> = Interturbine Volume Pressure, P<sub>t4.5</sub> - psia

 $X_5$  = Augmentor Pressure,  $P_{t7m}$  - psia

 $X_6$  = Fan Inside Diameter Discharge Temperature,  $T_{t2.5h}$  -  ${}^{\circ}R$ 

 $X_7$  = Duct Temperature,  $T_{t2.5c}$  -  ${}^{\circ}R$ 

 $X_8$  = Compressor Discharge Temperature,  $T_{t3}$  -  ${}^{\circ}R$ 

X<sub>9</sub> = Burner Exit Fast Response Temperature, T<sub>t4hi</sub> - °R

 $X_{10}$  = Burner Exit Slow Response Temperature,  $T_{t4lo}$  -  ${}^{\circ}R$ 

 $X_{11}$  = Burner Exit Total Temperature,  $T_{t4}$  -  ${}^{\circ}R$ 

 $X_{12}$  = Fan Turbine Inlet Fast Response Temperature,  $T_{t4.5hi}$  -  ${}^{\circ}R$ 

 $X_{13}$  = Fan Turbine Inlet Slow Response Temperature,  $T_{t4.5lo}$  - °R

 $X_{14}$  = Fan Turbine Exit Temperature,  $T_{t5}$  -  ${}^{\circ}R$ 

 $X_{15}$  = Duct Exit Temperature,  $T_{t6c}$  -  ${}^{\circ}R$ 

 $X_{16}$  = Duct Exit Temperature,  $T_{t7m}$  -  ${}^{\circ}R$ 

## 2. Engine Inputs

U<sub>1</sub> = Main Burner Fuel Flow, WFMB - tb/hr

 $U_2$  = Nozzle Jet Area,  $A_j$  -  $ft^2$ 

U<sub>3</sub> = Inlet Guide Vane Position, CIVV - deg

U<sub>4</sub> = High Compressor Variable Vane Position, RCVV - deg

U<sub>5</sub> = Customer Compressor Bleed Flow, BLC - %

## 3. Engine Outputs

Y<sub>1</sub> = Engine Net Thrust Level, FN - tb

Y<sub>2</sub> = Total Engine Airflow, WFAN - tb/sec

Y<sub>3</sub> = Turbine Inlet Temperature, T<sub>t4</sub> - °R

Y<sub>4</sub> = Fan Stall Margin, SMAF

Y<sub>5</sub> = Compressor Stall Margin, SMHC

 $Y_6$  = Fan Exit  $\Delta P/P$ ,  $(P_{t2.5} - P_{s2.5})/P_{s2.5}$ , based on test data

 $Y_7$  = Fan Exit  $\Delta P/P$ ,  $(P_{t2.5} - P_{s2.5})/P_{s2.5}$ , theoretical function of area and airflow.

These states, inputs, and outputs can be noted on Figure 3. The 16 states are the outputs of the 16 integrations. In the heat transfer cases, the integrations are implied by the transfer functions.

The matrix coefficients for this program were generated by Pratt & Whitney Aircraft using an offset derivative technique with a forced steady-state match. This technique was computerized on the F100 dynamic simulation and operated as follows.

 Each X was perturbed one at a time while holding all other X's and all U's constant. This allows calculations of the A and C matrix components.

$$\dot{\mathbf{X}} = \mathbf{A} \cdot \mathbf{X} + \mathbf{B} \cdot \mathbf{0}$$

$$\mathbf{Y} = \mathbf{C} \cdot \mathbf{X} + \mathbf{D} \cdot \mathbf{0}$$

For a simple 2×2 A, B, C, D example:

 Each U is then perturbed, one at a time, with the simulation operating in the steady-state mode. This forces the B and D matrixes into a steady-state match for X and Y with perturbations in U.

$$AX + BU = 0$$
  
 $CX + DU = Y$   
 $A, C = known$ 

For a simple  $2\times2$  A, B, C, D example:

$$\begin{split} B_{11} &= -(A_{11}X_1 + A_{12}X_2)/U_1 \\ B_{12} &= -(A_{11}X_1 + A_{12}X_2)/U_2 \\ B_{21} &= -(A_{21}X_1 + A_{22}X_2)/U_1 \\ B_{22} &= -(A_{21}X_1 + A_{22}X_2)/U_2 \\ D_{11} &= (Y_1 - C_{11}X_1 - C_{12}X_2)/U_1 \\ D_{12} &= (Y_1 - C_{11}X_1 - C_{12}X_2)/U_2 \\ D_{21} &= (Y_2 - C_{21}X_1 - C_{22}X_2)/U_1 \\ D_{22} &= (Y_2 - C_{21}X_1 - C_{22}X_2)/U_2 \\ \text{or} \quad Bij &= -(Ai_1X_1 + Ai_2X_2 + ...)/Uj \\ Dij &= (Y_1 - C_1X_1 - C_1X_2 - ...)/Uj \end{split}$$

The complete set of A, B, C, and D matrix coefficients is provided in Appendix B

Several different levels of perturbations on the states (X's) and inputs (U's) were evaluated at sea level static (SLS) idle. For other programs this has proved to be the most difficult point to obtain good linear models. This was also true for this program, so idle was used for model and perturbation comparisons.

Statistical error indexes were calculated to mathematically compare the different perturbation combinations. The error index is an indicator of how closely the response of the linear and nonlinear models match as a function of time. The error index was defined as:

Error Index for State 
$$X = \sqrt{\sum_{i=1}^{n} \left(\frac{\Delta X_{1i} - \Delta X_{n1i}}{\Delta X_{n1i}}\right)^2}$$

where:  $\Delta X_1$  = Linear model state variable change in response to a change in U

 $\Delta X_{nl}$  = Nonlinear model state variable change in response to a change in U

i = Points in time selected for comparison

n = Number of points selected.

For these studies, every sample time (0.007 sec) was used in the error index for 8-sec transients.

Figure 5 illustrates error index differences for each of the 16 states with various levels of input steps. Error indexes of less than 0.001 provided a good transient match to the nonlinear deck, as illustrated by Figure 6, which compares linear and nonlinear responses at SLS idle for a fuel flow step. Studies at other power settings and flight conditions showed the "best" overall agreement between linear and nonlinear model occurred when the states were perturbed 0.5% and the inputs were perturbed as follows:

WFMB	3%
$\mathbf{A}_{\mathbf{j}}$	3%
CIVV	5 deg
RCVV	1 deg
BLC	0.2%

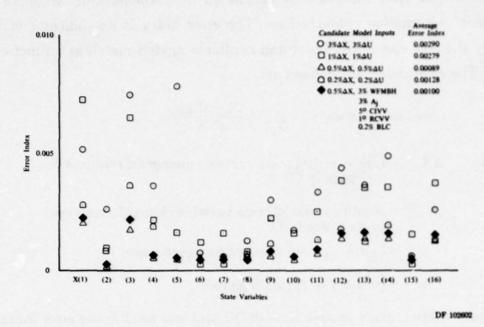


Figure 5. Error Index Differences

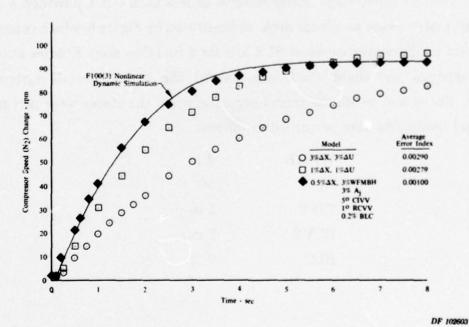


Figure 6. Sixteen-State Linear to Nonlinear Model Comparison - Idle Power

The same perturbation steps were used for all linear model development. Figure 7 compares the linear and nonlinear model responses at intermediate power for other parameters. Illustrated are fuel flow, burner pressure, turbine inlet temperature, and fan stall margin changes. The slight steady-state differences that occur are attributed to deck tolerances, and the models are considered to be in good agreement.

Advantages of the offset derivative method are short computer run time and improved steady-state matching. The dynamic characteristics (system eigenvalues and time response) of the linear models generated by this technique have been found to be similar to those generated using the least square coefficient matching technique in system output data. One disadvantage is that the offset derivative technique does not have the potential of being applied directly to test data.

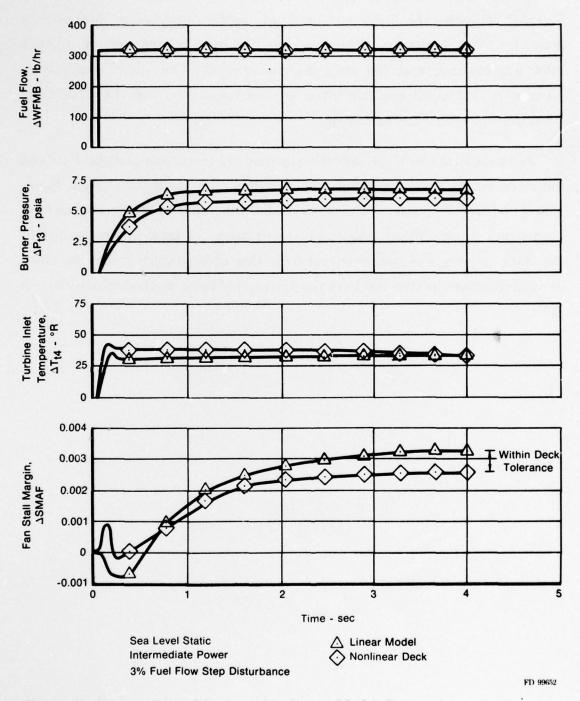


Figure 7. Sixteen-State Linear to Nonlinear Model Comparison - Intermediate Power

#### SECTION IV

#### SENSOR AND ACTUATOR CHARACTERISTICS

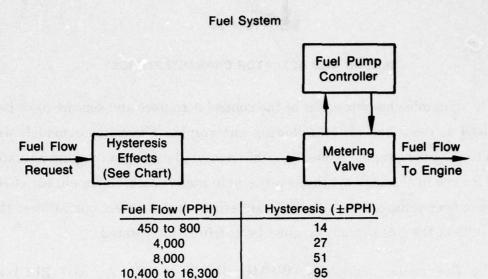
The dynamic characteristics of the control actuators and sensors have been linearized as described in the following paragraphs. These linear models were added to the linear engine model to obtain a total plant (sensor-engine-actuator) model for use in the LQR synthesis process. In many cases, the nonlinear effects that have been removed have a significant effect on the system operation so that the results of the linear analysis must be carefully interpreted.

The five control variables (WFMB, RCVV, CIVV, A<sub>j</sub>, and BLC) are positioned by actuators with limits and maximum slew rates, as shown in Table I.

Table I. Actuator Limits and Maximum Rates

Actuator	Maximum	Minimum	Rate Limits
WFMB	16,300 tb/hr	450 tb/hr	15,800 tb/hr/sec
RCVV	4 deg (Axial)	-40 deg (Cambered)	40 deg/sec
CIVV	0 deg (Axial)	-40 deg (Cambered)	48 deg/sec
Aj	6.4 ft <sup>2</sup>	2.8 ft <sup>2</sup>	1.8 to 3.6 ft <sup>2</sup> /sec
BLC	6%	0	Instantaneous

The dynamics of the main burner fuel flow (WFMB) path are dominated by the fuel metering valve and the pump controller, which also regulates the system pressure. The hysteresis represents an accumulation of mechanical backlash in the linkages and servos of the hydromechanical control system. The fuel flow hysteresis effects are relatively small and can be removed with no serious consequences. The fuel flow dynamics can then be represented by a series of two simple first order elements (Figure 8).



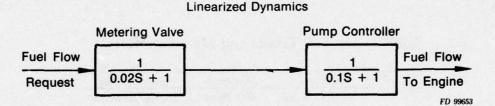


Figure 8. Main Burner Fuel Flow Dynamics

The rear compressor variable vane (RCVV) actuation system (Figure 9) is characterized by nonlinear actuation rates and vane linkage hysteresis. This loop can be linearized by replacing the control rate curve with a gain of 40 deg/sec and removing the linkage hysteresis function. These nonlinear items are significant and must be considered as the control analysis proceeds.

The compressor inlet variable vane (CIVV) actuation system includes a stepper motor interface (Figure 10). This loop is linearized by assuming the stepper motor is continuously variable and removing the nonlinear control rate and position limits. There is no significant hysteresis in the CIVV system.

The exhaust nozzle jet area  $(A_j)$  control synamics, (Figure 11) are highly dependent on the nozzle pressure loading conditions and the pneumatic power supply pressures and temperatures. The linearized representation is derived from a nonlinear simulation model and is valid for input steps of  $\pm 3\%$ . The damping

(0.27 to 0.56) and natural frequency (3 to 6 Hz) ranges include the effects of both flight condition and power setting variations.

The customer bleed (BLC) actuator is not part of the F100 system. Test data show the response of this valve to be virtually instantaneous. For a practical system, a servo time constant of 0.02 sec can be assumed.

With the exception of the FTIT and  $T_{t2.5}$  sensors, the engine sensors can be represented by first order time constants, as presented in Table II. The fan speed  $(N_1)$  sensor dynamics represent an electronic blade counting device. The compressor speed  $(N_2)$  sensor is a hydromechanical device mounted on the main fuel pump. The  $T_{t2}$  sensor response is typical of a shielded thermocouple placed directly in the airstream. The pressure sensor  $(P_b)$  represents a hydromechanical device used with the current control. A typical electrical pressure sensor that has been set up to avoid excessive line dynamics would be faster with a time constant on the order of 0.01 to 0.02 sec.

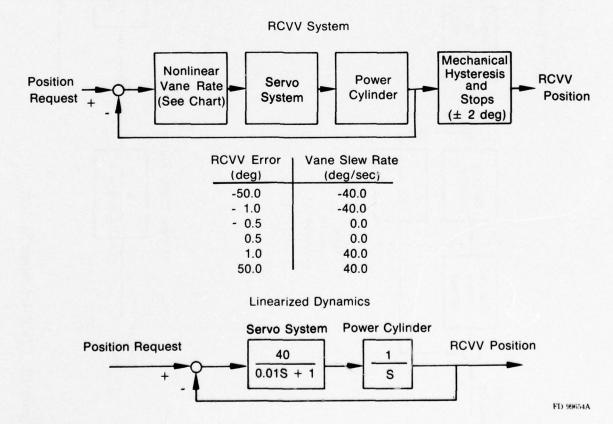
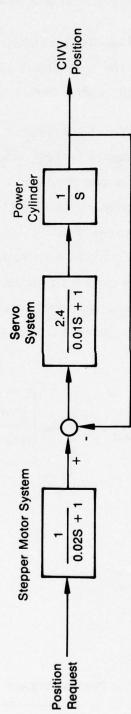


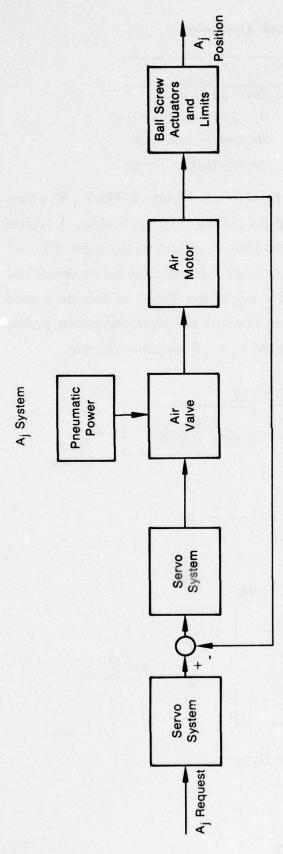
Figure 9. RCVV Actuation System



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FD 99655

Figure 10. CIVV Actuation System





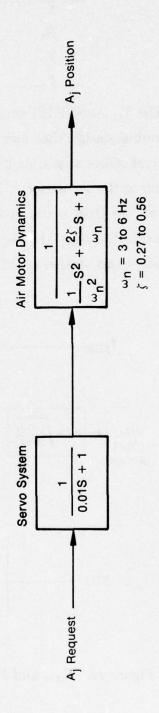


Figure 11. Exhaust Nozzle Actuation System

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Table II. Sensor Dynamics

Parameter	Time Constant (sec)	Comments
N <sub>1</sub>	0.03	Electronic Device
$N_2$	0.05	Hydromechanical Device
$T_{t2}$	1.5	Shielded Thermocouple
P <sub>b</sub>	0.05	Hydromechanical Device

The  $T_{t2.5C}$  and FTIT sensor models are shown in Figure 12. The  $T_{t2}.5C$  sensor is a unique design that has been configured for the F100 application. A typical electrical sensor at station 2.5 would have a time constant on the order of 1.5 sec (similar to the  $T_{t2}$  sensor). The FTIT sensor has been set up to match sensor test data at sea level static conditions. This model was found to provide a good representation over the flight envelope. Limited life instrumentation probes located at this station could be represented by a 1.0-sec time constant.

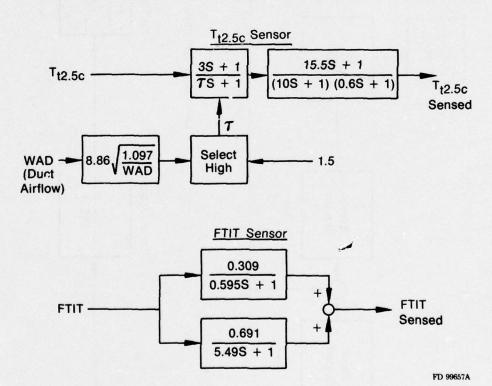


Figure 12. Tt2.5c and FTIT Sensor Dynamics

#### SECTION V

## **F100 ENGINE CONTROL CRITERIA**

Advanced propulsion systems operate at or near design limits with tight control of speed, pressure, temperature, and airflow to achieve maximum performance while maintaining engine durability. An accurate and reliable control system is required to ensure high engine performance and operational stability throughout the flight envelope. The control system must sense pilot commands, airframe requirements, and critical engine parameters; compute the necessary schedules; and actuate system variables for total engine control over the full range of operation. Mission, airframe, and engine requirements are combined to generate a control criteria list as shown on Table III. These criteria may differ in details for more complex propulsion systems, but the general character will be the same. Table III represents an approximate priority list of the design criteria for the F100 system. Engine protection is at the top of the list because of concern for aircraft safety. In cases where safety of the aircraft depends on engine dynamics (for example, V/STOL), transient response would move to the top of the list, along with engine protection.

### **ENGINE PROTECTION**

Limiting values that must not be exceeded to ensure adequate protection of the engine are described in the following paragraphs.

## **Temperature Limits**

The actual temperatures that must be maintained within limits to protect the engine are  $T_{t3}$  (compressor discharge temperature) and  $T_{t4}$  (turbine inlet temperature). The  $T_{t3}$  limit is based on maximum allowable metal temperature and the fact that turbine cooling effectiveness decreases rapidly above this value. The  $T_{t4}$  limit is based on maximum allowable metal temperature and turbine life considerations.

## Table III. Outline of Control Criteria

## **Engine Protection**

Temperature limits
Speed Limits
Pressure Limit
Structural Stability

## **Engine Stability**

Engine Fluctuations
Fan and Compressor Stall Margins
Augmentor Spikes

## Compatibility With Inlet/Aircraft

Airflow Corridor Minimum Burner Pressure

## Steady-State Performance and Accuracy

Thrust Modulation
Thrust and Fuel Consumption Requirements
Control Sensitivity

Deterioration
Installation
Inlet Conditions
Augmentor Ignition

## Repeatability

## Transient Requirements

Thrust Monotonic Function of Time Acceleration and Deceleration Times Combustion Stability

Trim Capability/Procedure Start/Transition Capability

The current control limits FTIT (fan turbine inlet temperature) to maintain  $T_{ts}$  and  $T_{t4}$ , as described above. Engine performance simulations (based on test data) are used to obtain correlations of  $T_{ts}$  and  $T_{t4}$  with FTIT over a range of engine operating conditions. The correlations are then combined into a schedule of FTIT as a function of  $T_{t2}$  (engine inlet total temperature).

To help meet transient response requirements, it is allowable to exceed the FTIT limit for a period of not more than 0.5 sec.

# **Speed Limits**

Fan speed  $(N_1)$  is limited as a function of  $T_{t2}$  (engine inlet total temperature). Under normal conditions,  $N_1$  will be scheduled below the limit because of performance and stability considerations. Compressor physical speed  $(N_2)$  is limited to an absolute maximum value. Usually  $N_2$  is maintained safely below this level. Any overspeed requires at least a visual inspection of the high rotor.

### **Pressure Limit**

To assure structural integrity, burner pressure is limited to a maximum value. This limit is normally encountered only at low-altitude, high Mach number conditions.

# **Structural Stability**

Engine fan and compressor variable geometry must be scheduled within the flutter boundaries.

### **ENGINE STABILITY**

# **Engine Fluctuations**

Under steady-state operating conditions, engine thrust fluctuations between ground idle and maximum continuous thrust must not exceed  $\pm 1\%$  of intermediate thrust or  $\pm 5\%$  of the thrust available at the power lever position and flight condition, whichever is less. During operation above intermediate thrust, fluctuations must not exceed  $\pm 1\%$  of the thrust available at the

condition. During engine transients, the variation of engine airflow from the corresponding steady-state values of the power setting selected must not cause propulsion system instability.

# **Fan and Compressor Stall Margins**

The ground rule for fan and compressor stability is to not allow fan stall margin to go below 0.15 or compressor stall margin to go below 0.05. Stall margin is defined by:

$$SM = \left[ \begin{array}{c} \underline{\text{(Pressure Ratio Stall - Pressure Ratio Operating)}} \\ \underline{\text{Pressure Ratio Stall}} \end{array} \right] \\ \underline{\text{At constant airflow}}$$

# **Augmentor Pressure Spikes**

Jet area and augmentor fuel flow must be coordinated during augmentor transients so that the resulting pressure spikes are within the limit. Pressure upspikes are caused by A<sub>j</sub> being too small and can result in fan stalls. Downspikes are caused by A<sub>j</sub> being too large and can result in augmentor blowouts. (Augmented engine operation will not be investigated in this program. However, pressure spikes resulting from augmentor initiation will be considered.)

### **COMPATIBILITY WITH INLET/AIRCRAFT**

### **Airflow Corridor**

Engine airflow limits are set by inlet constraints. Airflow variation is restricted for supersonic operation to help maintain the inlet shock at a desirable location.

### **Minimum Burner Pressure**

To provide accessory air for various aircraft subsystems, it is required that engine burner pressure be maintained above a minimum level of 50 psia.

### STEADY-STATE PERFORMANCE AND ACCURACY

### **Thrust Modulation**

The relationship between thrust and power lever is of the fully modulated type, free of abrupt changes and essentially linear with a thrust step of not more than 4% of intermediate rated thrust when augmentation is initiated or terminated.

# **Thrust and Fuel Consumption Requirements**

Engine thrust and fuel consumption specifications are given in engine specification tables.

# **Control Sensitivity**

### **Deterioration**

The engine must be controlled to maintain the required thrust, regardless of engine deterioration, except where the level of deterioration is such that the engine limits would be exceeded. Fan and compressor efficiency decrements of 1 to 2% are typical. Turbine efficiency losses up to 3% also have been experienced.

### Installation

The engine must be controlled so that thrust is insensitive to variations in horsepower extraction and customer bleed flow up to the point where engine limits would be exceeded.

### **Inlet Conditions**

The engine must operate satisfactorily in the face of inlet pressure  $(P_{t2})$  and temperature  $(T_{t2})$  variations and rates of change of inlet conditions common to the operation of highly maneuverable aircraft. Steady-state inlet variations are less than 1% of  $P_{t2}$  and  $T_{t2}$ . For aircraft accelerations,  $P_{t2}$  and  $T_{t2}$  rates of change of 0.15 psia/sec and 2°F/sec are representative. For aircraft decelerations, the rates may be as much as -0.5 psia/sec and -7°F/sec.

# **Augmentor Ignition**

The control must be insensitive to augmentor ignition pulse.

# Repeatability

Stabilized thrust at any power lever position must be repeatable.

### TRANSIENT REQUIREMENTS

# **Thrust Monotonic Function of Time**

For increases or decreases in power lever angle, engine thrust must be a monotonically increasing or decreasing function of time, respectively.

# **Acceleration and Deceleration Times**

Transient thrust response requirements are specified for sea level static, standard day, uninstalled conditions (Table IV). With maximum horsepower extraction and bleed flow, the response times typically cannot exceed 125% of these values. Idle thrust is defined as the lowest attainable engine thrust, intermediate as the highest nonaugmented thrust, and maximum is defined as the highest augmented engine thrust.

Table IV. F100 Transient Thrust Requirements

Thrust Change (%)	Idle- Intermediate	30% Intermediate- Intermediate	Idle- Maximum	Intermediate- Maximum	Maximum- Intermediate	Intermediate- Idle
90	4	3.5	8	4.5	2	3
98	15	15	15	12	7	20

Note: All times are given in seconds to achieve the indicated % of thrust change for sea level-static, standard day, with no bleed flow or horsepower extraction.

For thrust increments of  $\pm 3000$  fb, starting from a stabilized thrust between 25 and 45% of intermediate thrust available, the time for 90% of the thrust response shall not exceed 1.2 sec.

# **Combustion Stability**

Primary burner fuel flow must be maintained at a sufficiently high level during engine decelerations so that no blowout of the burner can occur.

# TRIM CAPABILITY.

External adjustments to the control must have sufficient range to permit the engine to produce, under standard sea level conditions, rated thrust or higher above the idle power level position; rated thrust or lower at the idle power lever position; rated specific fuel consumption or lower above the idle power lever position, and rated fuel flow or lower at the idle power lever position within the limits of the measured gas temperature, rotor speeds, airflows and engine pressure ratio associated with the ratings.

### START/TRANSITION

For this program, the engine will be started using the current F100 control system. Provision must be made for transition to the multivariable control mode at idle power.

# **SECTION VI**

# **FLIGHT POINT SELECTION**

The flight points at which the control is to be evaluated were selected jointly by P&WA/SCI/NASA/AFAPL.

Sixteen points were chosen to fully describe the operational flight envelope, with emphasis on regions of extreme conditions of pressure, temperature and control system limitations. Basic design points were also included for system evaluation. Table V shows the selected flight condition points and the tests that will be run on each. Figures 13 and 14 show plots of each of the points as altitude/Mach number and inlet condition maps respectively.

Table V. Selected Flight Test Points

Point	Mach No.	Altitude [ft (Thousands)]	$P_{\rm t2}$ (atm) $T_{\rm t2}$ (°R)	T <sub>t2</sub> (°R)	Criterion	Linear Model Points	SCI Design Evaluation Points	NASA Hybrid NASA Engine Test Points Test Points	NASA Engine Test Points
В	0	0	1.00	519	Basic Design Point	×	×	×	
p	6.0	10	1.16	299	Basic Design Point	×	×	×	×
	0.3	20	0.49	456	Low Mn, Low Altitude	×			
p	9.0	10	0.88	519	NASA Test Point	×		×	×
е	9.0	30	0.38	445	Low Mn, Medium Altitude	×			
J	1.2	0	2.40	899	P <sub>b</sub> Limit Point	×	×	×	
pr	2.2	40	1.79	768	High Mn, Medium Altitude	×		×	×
h	6.0	45	0.25	454	Low Mn, Medium Altitude	×		×	×
j	6.0	65	0.10	454	Low Pt2	×		×	×
¥	2.5	65	0.84	928	High Mn	×		×	
-	6.0	30	0.50	479	Basic Design Point	×	×	×	×
ш	1.8	75	2.05	652	High Altitude	×		×	×
п	1.8	20	2.50	737	High Dynamic Pressure	×		×	×
d	1.8	40	1.01	643	Low Supersonic Point	×			
ъ	2.15	58.5	69.0	750	NASA Test Point	×		×	×
1	1.2	10	1.65	622	NASA Test Point			×	×

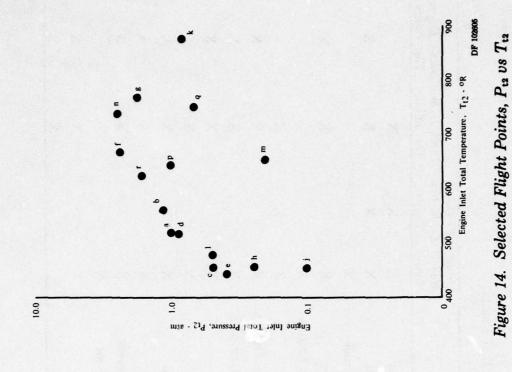


Figure 13. Selected Flight Points, Altitude vs Mach Number

DF 102604

Mach Number

Altitude - ft (Thousands)

### **SECTION VII**

# **CONTROL SYSTEM DESCRIPTION AND EVALUATION**

A simplified schematic of the F100 multivariable control algorithm, developed by Systems Control, Inc. (Vt), is shown in Figure 15. This algorithm was developed using reduced order linear models derived from the sixteenth order linear models provided by P&WA. Reference schedules and limits, based on the requested power setting and flight condition, were provided for the following engine state, control, and trim variables. These were obtained by exercising the nonlinear simulation.

# 1. Engine State Variables

Low rotor speed (N<sub>1</sub>)

High rotor speed (N<sub>2</sub>)

Burner Pressure (Pb)

Augmentor pressure (Pts)

Main burner fuel flow (WFMB)

# 2. Control Variables

Main burner fuel flow (WFMB)

Exhaust nozzle area (A<sub>i</sub>)

Compressor inlet variable vanes (CIVV)

Rear compressor variable vanes (RCVV)

Compressor bleeds (BLC)

### 3. Trim Variables

Fan turbine inlet temperature (FTIT)

Low rotor speed (N<sub>1</sub>)

Fan exit  $\Delta P/P$ 

Burner pressure (P<sub>b</sub>)

The transition control logic operates on the reference schedules to provide a transient reference for the controller to follow for any magnitude input.

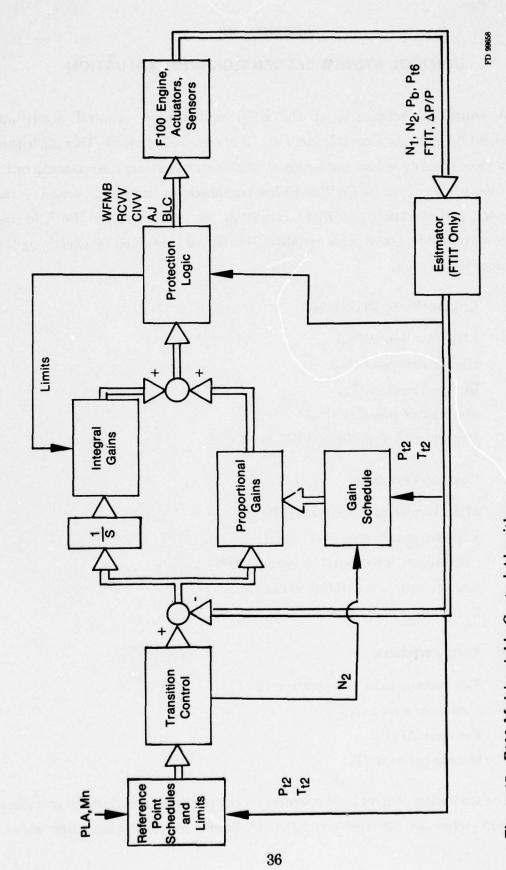


Figure 15. F100 Multivariable Control Algorithm

Transition commands, designed to avoid excessive error terms, which can lead to erroneous control action and/or saturation of controls, are calculated from an analysis of linearized engine data to provide desired rates of change on engine output variables, such as surge margin, thrust, and turbine inlet temperature.

The control mode is basically proportional plus integral with the outputs of each of these paths added to a scheduled value of each control variable. A proportional gain matrix, which forms the linear quadratic regulator (LQR), operates on error terms from all five state variables to drive all five control variables. These gains are calculated by solving the matrix Riccati equation for a performance index with weightings on both the states and the control inputs, sized to obtain the desired closed loop response. Control gains required at a given flight condition and power setting are scheduled as a function of air inlet density  $(P_{t2}/T_{t2})$  and corrected high rotor speed  $(N_2/\sqrt{\theta t_2})$ . (A two-dimensional linear interpolation.)

Integral gain terms are calculated separately for trim action roughly on the order of a 1-sec time constant response. The integrators on fuel flow and nozzle area are driven by errors on engine output variables. When driven by errors on fan  $\Delta P/P$  and low rotor speed, the fuel flow and nozzle area integrators operate to set the steady-state match. The CIVV, RCVV, and BLC integrators are driven to assure that the geometry returns to the proper steady-state schedule. These integrators are only allowed to operate in this fashion when the system is near steady state, which is determined by the magnitude of a high rotor speed error calculated in the "transition control" block. If FTIT or P<sub>b</sub> limits are exceeded, the "error select logic" selects the appropriate error terms to input to the fuel flow and nozzle area integrators, and allows the fuel flow integrator to trim regardless of the magnitude of the fan speed error. The integrators are also allowed to wind down if the error terms are of the appropriate sign. The integrator logic is controlled with inputs from the engine "protection" block wherein amplitude limits for each control variable are checked and flags set for clamping appropriate integrators. Finally, deadband is applied to all integrator inputs to avoid limit cycling due to downstream hysteresis and other error sources.

The sensed value of fan turbine inlet temperature (FTIT) (output from a slow sensor), the steady-state reference value of FTIT, and a function of fuel flow are combined in the estimator block to predict whether the current combination of inputs will cause a temperature overshoot at a later point in the transient. This predicted value of steady-state temperature is then compared with the FTIT limit and, if required, the fuel flow integrator downtrims to reduce fuel flow before an overshoot can occur. In the protection block, hard limits on fuel flow and geometry excursions are provided in case of control malfunction or as part of the designated transient excursion.

A preliminary control evaluation was performed by P&WA in accordance with the 10 test items listed in Table VI. These test items include large and small power transients, inlet, and augmentor disturbances, control operation on engine limits, and with the area saturated, and steady-state operation in the presence of deterioration, power extraction and bleed extraction. This preliminary evaluation was established to test the multivariable control on the nonlinear digital simulation prior to the NASA-LeRC evaluation on the hybrid simulation and the test engine. The NASA hybrid simulation plans include more extensive testing over the entire engine envelope.

In general, the large amplitude acceleration and deceleration transients met the specific time response requirements. The surge margins for these transients occasionally dipped slightly below the F100 Bill-of-Material (BOM) control margins, but not significantly. In fact, as illustrated in Figure 16, during most of the transient there is stall margin available for achieving even more rapid responses. The turbine temperature was well controlled. A typical idle to intermediate transient is presented in Figure 16. The control responded well to the large amplitude "Bodie" Power lever command.

Small perturbation response requirements are not specified in the control criteria but the guideline of 90% thrust in 1.2 sec was used as a measure of goodness. The small amplitude responses that were obtained initially showed that considerable potential existed for achieving faster part power transients.

Table VI. Control Evaluation

	Test Item	Input Specification	Flig SLS	Flight Condition 0.9/10K	0.9/30K
<i>≟</i>	Large Transient Response (5 to 10 sec each)	<ul> <li>Idle to Intermediate</li> <li>Intermediate to Idle</li> <li>"Bodie" Transient</li> <li>APLA 3000 tb below</li> <li>Intermediate to Intermediate</li> </ul>	××××	××II	××II
çi	Small Amp Transient Response (2 to 3 sec)	+3 deg PLA	At PLA of 20, 52, 80 deg	80 deg 80 deg (Intermediate) (Intermediate)	80 deg (Intermediate)
က်	No Holds Barred PLA (10 sec)	2 Runs	×		
4	Disturbances (2 to 3 sec)	$\begin{array}{l} \Delta P_{t_7} = 4\% \\ \Delta P_{t_2} = 1\% \end{array}$	83 deg (Internediate) 83 deg (Internediate)	1 1	11
ıć	Transient Limit Check  • P <sub>b</sub> Limited (FTIT = Limit Is Addressed in Item 1) (2 to 3 sec)	<ul> <li>Set P<sub>b</sub> 10 psi Below Intermediate Setting</li> <li>APLA = 10 deg</li> </ul>	10 deg PLA Input at 73 deg	I	ı
9	Transient Saturation Check - $A_1 = Constant$	A, Limited	Near Intermediate	1	ı
t-	Idle Die Out Check	Maximum HPX + BLC Specified	Idle $\Delta$ PLA = 20 to 35 deg	1	1
œ.	Deterioration (See Table VII)	0 to ½ Nominal - Nominal Specified	83 deg Steady State	1	1
6	Power Extraction (See Table VII)	Power Extraction (See Table VII) 0 to 1/2 Nominal - Nominal Specified	83 deg Steady State	I	I
10.		Bleed Extraction (See Table VII) 0 to ½ Nominal - Nominal Specified	83 deg Steady State	1	ı
Note:	te: 20 deg PLA = Idle thrust				

Idle thrust
 Intermediate (maximum nonaugmented) thrust
 Maximum augmented thrust.

20 deg PLA 83 deg PLA 130 deg PLA

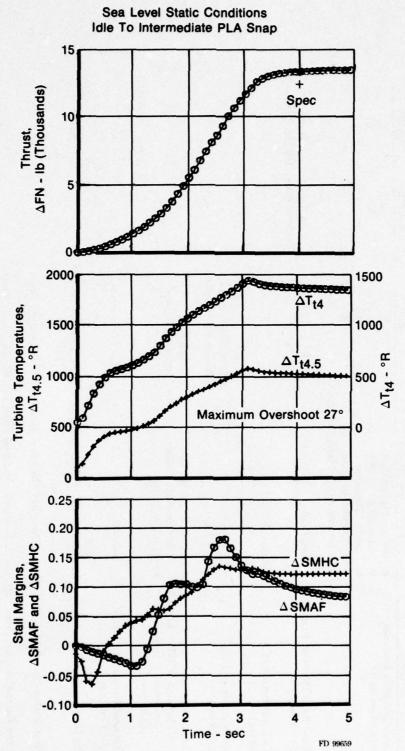


Figure 16. Typical Idle-to-Intermediate Transient

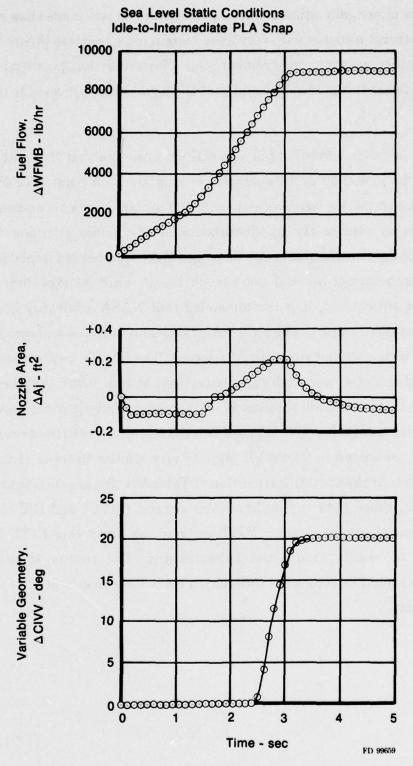


Figure 16. Typical Idle-to-Intermediate Transient (Continued)

With a few minor gain adjustments, a part power run was made showing a much improved thrust response with very little margin reduction (see Figure 17). While not a specified program requirement, this illustrated that potential exists for achieving faster thrust response and that the logic can be adjusted to obtain this response.

The "no holds barred" input was defined to ensure that the switching logic could not be fooled by cyclic operation around a switch point. The Pt2 and Pt7 disturbance responses were both smooth and stable. It was recommended that NASA further explore the P<sub>t7</sub> disturbance at the higher atltitude conditions where large  $P_{t7}$  disturbances occur. The high Mach number sea level data showed good thrust response but did not run on the Pb limit as expected. While no problem is anticipated, it is recommended that NASA arbitrarily lower the P<sub>b</sub> limit to test its operation. The locked A, transient was stable and compared very favorably with a similar run with A<sub>i</sub> released. The idle die out test showed that the control operated well with high extractions at idle power. Another test was performed to demonstrate that the W<sub>f</sub>/P<sub>b</sub> acceleration limit could provide surge protection in transients. The effect of deterioration, power extraction and bleed extraction, presented in Table VII, appear very similar to those of the current F100 control. At the operating condition of Table VII, the engine is running on an FTIT limit. Since both the multivariable control (MVC) and Bill-of-Material (BOM) control are integrating FTIT error to maintain this FTIT limit, the similarity of results should not be surprising. The results at other flight conditions should be somewhat different. This will be investigated in the NASA hybrid tests.

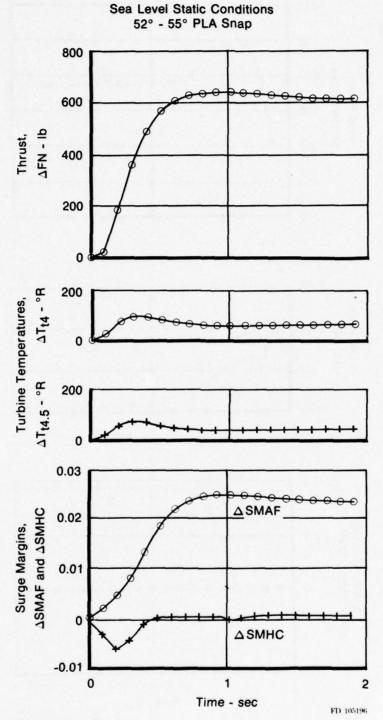


Figure 17. Typical Small Amplitude Part Power Transient

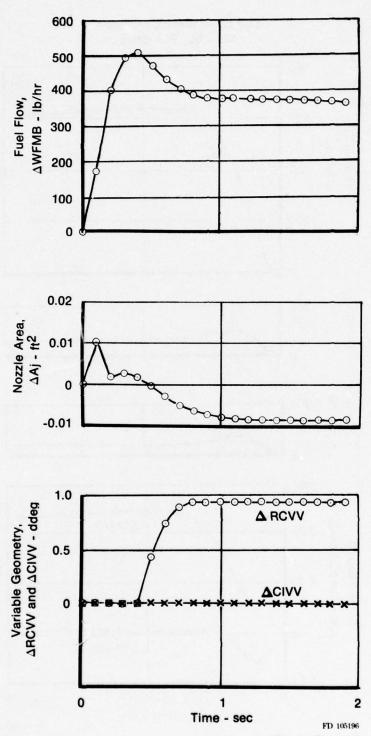


Figure 17. Typical small Amplitude Part Power Transient (Continued)

Table VII. Deterioration/Power Extraction/Bleed Effects

Flight Point: Alternate = OK PLA = 83 (Intermediate)

See Table VI) eterioration	Case 0	MVC	BOM	MVC	BOM
eterioration	0				
	•	0	0	0	0
	1/2 Nominal	- 4.27	- 2.71	+0.30	+0.26
	Nominal	- 8.94	- 7.22	+0.97	+0.92
ower Extraction	0	0	0	0	0
	1/2 Nominal	- 0.11	+ 0.12	+0.04	+0.04
	Nominal	- 0.12	+ 0.15	+0.09	+0.09
leed Extraction	0	0	0	0	0
	1/2 Nominal	- 6.86	- 7.01	+2.91	+2.89
	Nominal	-14.51	-16.93	+6.30	+6.65
ll Effects		-25.20	-24.85	+8.95	+9.07
		1/2 Nominal Nominal  leed Extraction 0 1/2 Nominal Nominal	1/2 Nominal	1/2 Nominal	1/2 Nominal

**Nominal Effects** 

- Deterioration:

 $\begin{array}{l} \Delta\eta_{\rm fan} \,=\, -1\%, \; \Delta\eta_{\rm comp} \,=\, -2\%, \\ \Delta\eta_{\rm High\ Turbine} \,=\, -2.5\% \end{array}$ 

Power Extraction: HPX = 65 hp

Bleed Extraction: 6% of Compressor Airflow

# **SECTION VIII**

# **CONCLUSIONS AND RECOMMENDATIONS**

### CONCLUSIONS

It is concluded that the statement of work for phase one of the contract has been satisfied by delivering the data necessary to support both NASA Lewis Research Center and Systems Control, Inc. (Contract No. F33615-75-C-2053). Work accomplished in meeting these requirements included the following items:

- Dynamic Simulation Delivered NASA and SCI each have been provided an identical operational F100 dynamic simulation. Support has been provided to assure proper installation of this simulation on the NASA and SCI computers. The NASA simulation will be used as a basis for development of the hybrid engine simulation. The SCI simulation will be used as a model for control design and verification. The simulation was updated for multivariable control interfacing and includes conventional F100 control logic for comparative purposes.
- Linear Models Generated Linear models of the form

$$\dot{\mathbf{X}} = \mathbf{A}\mathbf{X} + \mathbf{B}\mathbf{U}$$
  
 $\mathbf{Y} = \mathbf{C}\mathbf{X} + \mathbf{D}\mathbf{U}$ 

were generated and verified against the F100 dynamic simulation. The models include all the dynamic terms (16 states) currently represented in the F100 dynamic simulation. Matrix coefficients were generated using an offset derivative method with a forced steady-state match. This technique was selected over two other candidate techniques because of inherent steady-state accuracy and reduced computer run time. Models of the sensors and actuators were also provided, both in a nonlinear and linear format.

 Controls Criteria Provided - Detailed control criteria and goals for protection, stability, compatibility, performance, accuracy, transients, and trim were provided. The criteria proved adequate for all areas of control except the small amplitude response. In cases where aircraft operations depend on engine response, such as V/STOL, rapid transient responses would be more critical.

The expected variations in engine characteristics due to production differences and component degradation over the operating life of the engine were also provided.

SCI and NASA Lewis Research Center Supported - Systems
Control, Inc., has been supported throughout the control
synthesis process. This support included such areas as definition of control variables, selection of state variables, and
determination of significant dynamic elements.

NASA Lewis Research Center has been supported in developing the realtime, hybrid computer engine simulation. Assistance has been provided in defining the sensor and actuator interface requirements needed to run the F100 engine with the NASA computer/controller.

 Control Evaluated - A preliminary control evaluation was performed on the nonlinear simulation using the control criteria as a guide. Large amplitude transient time and protection criteria were met. There appears to be margin for more responsive control action if required.

# RECOMMENDATIONS

This program is directed toward the evaluation of advanced engine control techniques on a test cell engine at the NASA Lewis Research Center. It is recommended that future work include flight testing of these concepts on a vehicle such as the F-15 at the NASA Dryden Flight Research Center.

This program addressed the application of advanced control techniques to a nonaugmented engine control system. It is recommended that future work extend

these techniques to include (1) the entire propulsion system, i.e., the engine (including augmentation) and the inlet, and (2) the combined flight and propulsion control systems.

The linear models provided for this program were generated based on the dynamics of the nonlinear F100 simulation. It is recommended that future work include development of techniques for generation of dynamic models directly from engine data.

# SECTION IX

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### APPENDIX A

### **NOMENCLATURE**

# **ENGINE/CONTROL PARAMETERS**

A - Area, in.<sup>2</sup>

A<sub>1</sub> - Exhaust Nozzle Area, ft<sup>2</sup>

BLC - Customer Compressor Bleed Flow Fraction

CIVV - Compressor Inlet Variable Vanes, deg

Cp - Specific Heat, Btu/tb °R

FN - Thrust, tb

FTIT - Fan Turbine Inlet Temperature, °R
GVIPOS - Compressor Inlet Variable Vanes, deg

H - Enthalpy, Btu/fb

h - Film Coefficient, Btu/(°R in.2 sec)

HVPOS - Rear Compressor Variable Vane, deg

J - Polar Moment of Inertia, ft-tb-sec<sup>2</sup>

KA2 - Distortion Constant

M - Mass, tb

Mn - Mach Number

P<sub>b</sub> - Burner Pressure

N<sub>1</sub> - Fan Rotor Speed, rpm

N<sub>2</sub> - Compressor Rotor Speed, rpm

P<sub>amb</sub> - Ambient Pressure, psia

PLA - Power Lever Angle

PR - Pressure Ratio

PRMA - Mass Average Pressure Ratio

 $P_{t2}$  - Engine Inlet Total Pressure, psia

P<sub>t2.5</sub> - Fan Discharge Pressure, psia

P<sub>t3</sub> - Compressor Discharge Pressure, psia P<sub>t4.5</sub> - Interturbine Volume Pressure, psia

P<sub>t7m</sub> - Augmentor Pressure, psia

Q - Torque, ft-fb

R - Gas Constant, ft/°R

RCVV - Rear Compressor Variable Vanes, deg

S - Laplace Operator SMAF - Fan Stall Margin

SMHC - Compressor Stall Margin

SNCOM - Compressor Rotor Speed, rpm

SNFAN - Fan Rotor Speed, rpm

TC - Turbine Cooling Airflow, tb<sub>m</sub>/sec

T<sub>in</sub> - Inlet Temperature, °R
TL - Leakage Airflow, †b/sec

T<sub>t2</sub> - Engine Inlet Total Temperature, °R

T<sub>t2.5c</sub> - Fan Outer Diameter Discharge Temperature, °R T<sub>t2.5h</sub> - Fan Inner Diameter Discharge Temperature, °R

Tts - Compressor Discharge Temperature, °R

Ttahi - Burner Exit Fast Response Temperature, °R

T<sub>t4lo</sub> - Burner Exit Slow Temperature, °R

T<sub>t4</sub> - Burner Exit Total Temperature, °R

T<sub>t4.5hi</sub>
 Fan Turbine Inlet Fast Response Temperature, °R
 T<sub>t4.5ho</sub>
 Fan Turbine Inlet Slow Response Temperature, °R

T<sub>ts</sub> - Fan Turbine Exit Temperature, °R

Ttec - Duct Exit Temperature, °R

T<sub>tem</sub> - Augmentor Mixed Inlet Temperature, °R

T<sub>t7m</sub> - Augmentor Exit Temperature, °R

V - Volume, ft<sup>3</sup>

Vj - Jet Velocity, ft/sec
W - Mass Flow, tb/sec

Wa - Airflow, tb<sub>m</sub>/sec

WAD - Duct Airflow, tb/sec

Wf - Fuel Flow, tb/hr

W<sub>fa</sub> - Augmentor Fuel Flow, tb/hr

WFAN - Total Engine Airflow, tb/sec

WFMB - Fuel Flow Main Burner, tb/hr

W<sub>fp</sub> - Primary Fuel Flow

WFTI - Fan Turbine Inlet Flow, tb/sec

Wg - Gas Flow, tb/sec

WHPT - High Pressure Turbine Flow, tb/sec

### **SYMBOLS**

γ - Ratio of Specific Heats

δ - Corrected Pressure

 $\Delta$  - Change

e - Small Shift

ζ - Damping Ratio

η - Efficiency

 $\theta$  - Corrected Temperature

 $\tau$  - Time Constant

 $\omega_n$  - Natural Frequency

### **MISCELLANEOUS**

ALT - Altitude

AFAPL - Air Force Aero Propulsion Laboratory

C - Cold (Fan OD)

COM - Compressor

CT - Compressor Turbine

EEC - Engine Electronic Control

F - Fan

FT - Fan Turbine H - Hot (Fan ID)

ID - Inside Diameter

LQR - Linear Quadratic Regulator

MVC - Multivariable Control

NASA - National Aeronautics and Space Administration

OD - Outside Diameter

P&WA - Pratt & Whitney Aircraft

RC - Rear Compressor

SCI - Systems Control, Inc. (Vt)

SLS - Sea Level Static

SM - Stall Margin

TOT - Total

UFC - Unified Fuel Control

V/STOL - Vertical/Short Takeoff and Landing

# UNITS OF MEASURE

Btu - British Thermal Unit

Deg - Degrees

°R - Degrees Rankine

hr - Hours

tb - Pounds

ft<sup>2</sup> - Square Feet

psia - Pounds per Square Inch, Absolute

rpm - Revolutions Per Minute

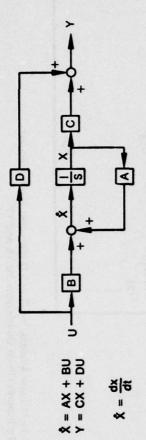
sec - Seconds

APPENDIX B

LINEAR MODEL DATA POINTS

Code	Point No.	Mach No.	Altitude ft (Thousands)	PLA (deg)	Comments
Basic Set	1	0	0	20	SLS Idle
Dusic Set	2	0	Ö	36	OLO Idie
	3	0	Ö	52	
	4	0	0	67	
	5	0	0	83	SLS Intermediate
	·	·	•	00	SES interintediate
Group I	1	0	0	24	
	2	0.9	10	83	
	3	0.3	20	24	
	4	0.6	10	20	
	5	0.6	30	24	
Croup II	c	1.2	0	83	
Group II	6				
	7	2.2	40	83	36 . 4
	8	0.9	45	130	Maximum Augmentation
	9	0.9	65	83	
	10	2.5	65	130	Maximum Augmentation
Extra	1	0.9	10	36	
	2	0.9	10	52	
	3	0.9	10	67	Added $\Delta P/P$ to Output for A the Following Runs
	4	0.9	10	83	
	5	0.9	30	36	
	6	0.9	30	52	
	7	0.9	30	67	
		0.0		•	
Group III	1	0	0	20	Deletd
	2	0	0	20	With BLD/HPX Extraction
	3	0	0	83	Deleted
	4	0.9	10	20	
	5	0.9	30	20	
	6	0.9	30	83	
Crous IV		0.0	45	99	
Group IV	1	0.9	45	83	
	2 3	0.9	45	52	
	3	0.9	45	40	Low P <sub>b</sub>
	4	1.8	75	83	
	5	1.8	20	83	
	6	0.3	20	83	
	7	1.8	40	83	
	8	2.5	65	83	High Mn
	9	2.15	58.5	83	

The linear F100 model provided is of the form:



- Engine State Variables
- Fan Speed, SNFAN (N,) rpm
- = Compressor Speed, SNCOM (N2) rpm
- = Compressor Discharge Pressure, Pt3 psia
- = Interturbine Volume Pressure, Pt4.5 psia
- = Augmentor Pressure, Ptrm psia

Customer Compressor Bleed Flow, BLC - % = High Variable Stator Position, RCVV - deg

= Main Burner Fuel Flow, WFMB - 1b/hr

2. Engine Inputs

= Inlet Guide Vane Position, CIVV - deg

U, 'n U

= Nozzle Jet Area, A<sub>j</sub> - ft<sup>2</sup>

- Fan Inside Diameter Discharge Temperature,
- = Duct Temperature, T<sub>t2.5c</sub> °R X,
- = Compressor Discharge Temperature, T<sub>13</sub> °R
- = Burner Exit Slow Response Temperature, Ttalo oR = Burner Exit Fast Response Temperature, Tuhi - °R XIO
- = Burner Exit Total Temperature, Tu °R
- Fan Turbine Inlet Fast Response Temperature,
- = Fan Turbine Inlet Slow Response Temperature, T4.510 - oR X13

Lu. shi - oR

- = Fan Turbine Exit Temperature, Tts °R XII
- = Duct Exit Temperature, Ttec oR
- = Duct Exit Temperature, Ttm oR

- Engine Outputs 3
- = Engine Net Thrust Level, FN tb
- = Total Engine Airflow, WFAN tb/sec
- = Turbine Inlet Temperature, Tt. °R
- Fan Stall Margin, SMAF
- Compressor Stall Margin, SMHC
- Fan Exit  $\Delta P/P$ ,  $(P_{t2.6} P_{a2.6})/P_{a2.6}$ , based on test data
- Fan Exit  $\Delta P/P$ ,  $(P_{t2.5} P_{s2.5})/P_{s2.5}$ , theoretical function of area and airflow.

First Page of Each Data Set The A Matrix

A31 A16,1A16,8	A16,8
A1,9 A1,10	A1,16

Second Page of Each Data Set The B, C and D Matricies

1	
	B <sub>11</sub> B <sub>12</sub> B <sub>15</sub>
	B <sub>16,1</sub> B <sub>16,5</sub>
	C <sub>11</sub> C <sub>12</sub> C <sub>13</sub>
	C <sub>51</sub>
	C <sub>1,9</sub> C <sub>1,10</sub>
	C <sub>59</sub>
	D <sub>11</sub> D <sub>12</sub> D <sub>15</sub>
	D <sub>51</sub> D <sub>55</sub>

Note: In Later Models, the C and D Matricies Are Expanded To Include Linear Coefficients for Y6 and Y7

FD 105197

(it

BASIC

P100 HODEL-HW=0.0, ALT=0.0, PLA=20- .5PC PERTX, 3PC PERTU, 8/28/75 PAR POINT NO. 1

	-1.595	-1.219	2.077	0709	11195E-01	.2988	.3688	-20.29	6.107	.9046E-01	37.62	9.508	. 4225	4.226	.8690F-01	.3709E-01	• • • •	-1.441	3850	.8400	1.190	28801E-02	.3208	.4002	-,3185	11458E-01	12333R-03	-3.780	1 .4083E-01	. 1750E-02	13033F-01	.5833E-02	-46°47
	-3,313	-1.020	2.197	1.814	1904E-01	.8144	-18.99	8109	1744E-01	022906E-03	-9.617	. 1308	.5755E-02	014534E-01	19.93	2.245		-2.643	6780	1.469	1.947	02 .2400E-02	.5633	-6902	5494	011304E-01	032318F-03	-6.572	01 .9997E-01	-02 .4346E-02	-6085E-01	05-19.99	27.49
	-15.22	27.39	-51.92	-1.003	.1159	-16.73	4.882	34.62	.4452	01 .6570E-	220.9	-2.382	1060	-1902E-	3838	-38.37		-1.097	2755	.5946	.9401	-9779F-	.2335	.2868	2265	1312E-	-022040E-	-2.700	.4372E-0	.1924E-	-19.99	-01 .3498E-	24.84
	-1184.	-148.7	326.3	6.999	-23.79	120.1	148.0	-123.4	-3.035	-024654E-	-1377.	20.03	.8862	295.5	3.824	12.29		-7.487	-2.685	5.859	8.300	-026503E-	2.238	2.767	-2.222	-011017	-03 1628E-	-26.43	-2848	-1.988	-1.595	-02 .8138E-	3.947
	636.3	-266.6	38.23	-177.1	01 2.509	9.255	11.51	-9.255	3259	025793E-	-110.9	640.7	28.47	67.08	.1738	4.809		1.510	3029	.6510	.9268	017689E-	.2532	.3074	2468	1130E-	1808E-	-2.952	-49.95	-1.998	-1.421	01 .9042E-	.4318
	-20.30	140.5	-147.6	119.0	01 .4554E-	5.677	7.654	50.43	013674	035789E-	126.4	-193.2	01-8.587	-86.48	012405	-11.12		-1.892	-1.538	2.008	2.333	011675E-	1767	.2190	1738	01 39.67	.5877	-50.08	3% • 03	1.734	17.51	01 .1607E-	-2.184
RIX	1.001	-3.838	5.722	.3985	12 1497E-	2302	3872	6762	12285E-	33400E-	-22.87	•	12 .1776E-		12 .4495E-	4.196	3	-7.419	1.769	4-470	5-464	125028B-	1.707	2.111	-1.695	7761E-	6679	-20.16	-9.717	4321	-4-601	12 .3104E-	3.601
THE A MAT	-2.324	5473	1.200	4044	.6532E-0	.4243	.3760	4527	1007E-0	1510E-C	-5-070	.7235E-0	.3196E-02	-6442E-0	6291E-0	-2.501		5/75	3.370	8684	1069	2975E-(	.1863	.2346	1849	-50.00	6667	-2.213	-9.329	9717-	-4.216	-3449E-	.9210

THE B MATRIX				
4.757 -70.04	-10.21	1816.	-5967.	
.1020 -83.18		-49.43	1259.	
9829 124.8	5.464	32.03	-9682.	
-3.437 75.77		2.728	4199.	
.2560E-01-20.11		.3921E-015871E-01-19.39	01-19.39	
-1.115 20.82	.9273	.2518	1369.	
	1.524	-4.088	1158.	
1.435 13.15	-2.335	17.53	-2993.	
.8773E-01-3.846	1	.4909E-011489	-3328.	
.1365E-025907E-017528E-032307E-02-49.43	E-017528E-	032307E-	02-49.43	
34.21 -678.6	-22.48	-80.44	.5013E+05	5
-1.572 -90.05		34.78	2962.	
6969E-01-4.001		.3106E-01.1.545	131.2	
639853.85		18.00	1853.	
10-3	•	•	9.111	
-1.061 98.66	•		-220.6	
THE C MATRIX				
.13172901	1,353	50.23	295.7	2.304:
.1736E-01 .2246E-025469E-019097E-01-1.165	E-025469E-	019097E-	01-1-165	3177E-01

-.3301E-01-.7333 -.3141 -.1177 -1.131 .2143 .9153E-01-.1711 -.1822E-02-.1661E-01-.1709E-02-.2427E-02-.2179E-01-.2219E-02-.5395E-02-.3117E-02 3240 --2034 7913E-02--2878E-02 1.000

BASIC

POINT NO. 2

P100 HODEL-HN ... 0, ALT=0.0, PLA=36- .5PC FERTX, 3PC PERTU, 8/28/75 PAA

	80 10	-1.408	2.114	.4707		6276E-01	1330	-19.86	0	.90	41.02	0	•	3	25	.2		2512	.2463	6749	1.798	.1665E-01	17587E-01	1557	.1636	.9853E-02	.1314E-03	1.180	•	99.	2168E	956E	10
	-2.787	14.	3.599	3.151	.4289B-02	.4510	-19.01	-1.021	1519E-01	022531R-03	-7.415	01 .5695E-01	02 .2582E-02	1048	9.6	9		5821	.2419	6123	2.790	01 .5941E-01	÷	1285	.1315	2 .0	0. 40	.9563	1 .226	3.	87E-0	02-20-00	~
	-9.752	.3	-56.14	-1.878	38	-18.34	77	.2	.4429	01 .6512R-	12.	8707E-	013786E-	590	N	8.1		6952	.1832	5084	1.562	92	5383E-0	1114	.1241	01 .7476E-0	3	9688*	-	.5233E-0	-19.81	01 .1495E-	-
	-756.2	-95.93	290.8	325.7	609.6-	37.29	81.89	-79.21	-1.841	022743E-	-585.2	1.299	-6063F-	26.	2.122	2		4.164	9	-4.421	11.66	_	1	-1.051	1.064	2	#	7.720	.9610E-0	-1.996	-2.883	2-,320	-6.583
	439.1	-228.6	5.4	-528.4	2.193	-1.756	-3.491	3.738	. 1549	12 .2065E-	27.63	318.8	14.17	32.49	1033	164.2		5.688	.1886	S	1.303	1.1	1583	1116	_	.5339E-0	-7119E-0	.9148	-50.00	-2.000	-2.795	113559E-0	3
	9919	113.2	-159.0	121.5	5 .2009	.5469	1.276	43.44	2 .2	3 .4	9	-87.10	1-3	-39.	115	-	- 1	S	.5	2.896	e.	1 1443E-0	3455E-0	7182E-0	.7546E-0	1 39.69	.5880	-47.42	39.29	1.746	17.91	12	-
RIX	.7209	•	4.596	.2148	17504B-0	1500	3196	2846	17146E	3 1072E-0	.559	2	3-	11799	7	-		5694	8.873	-3.416	6.880	-	13	-	-	.3559E-C	6662	6.038	-10.36	4598	8 4		4.2
THE A BATRIX	-3.348	-1.007	2.986	.7145	.1765B-0	.4782	.2452	7945	2247E-0	3333E-0	-5.933	.7594E-0	.3417E-0	8024B-01	8860E-0	-2.663	1	.5167	1.77.7	3796	8488	.1488E-0	4323E-0	8646B-0	.9121E-0	-50.00	9999*-	.6735	-10.38	4615	-4.750	2636E-C	-5667E-01

																4.108 9670 3006 3678 98648-02 131111-02		1.228 1.0281167	E-02 . 1314E-02	0. 0.	.2041E-04 .2349E-04 .2859E-04	925E-042110E-042573E-04					
.5485 -1865.	6.72 4835.	- 503	12.15 .1166E+05	.1484E-01 88.23	1	•	5.61 2459.	1022E-01-4218.	21498E-03-62.59	-16.36 .7064E+05	.5649 -5151.	E-01-		B-01	8068	66.64 166.9 4.	270E-031421E-01- 910E-03 .1251E-01-	7005E-016919	E-01	0.	187E-04 . 1879E-03	1687E-03-		1	854E-02	0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0	1
-50.84	-9.824	28.72	9.927	.3109	3.519	7.240	-7.516	2147	013210E-02	-57.35	.8349E-01	.3852E-02	.3394	1848	-42.56 -	3.590	E-042079E-03 E-036621E-02-	-,8779	E-02 .7060E-03	1.000	E-03 .1282E-04	IE-03 1207E-04		9.211	4550	-0-4989F -10-4	E-02
THE B HATRIX -4855 -170.4	93.3	-221.	-2.426 816.5	2224E-01-47.03	.6749E-01-27.93	.1582 -57.06	4216 42.34	.4434E-01 2.901	4039	9.978 376.0	.1026 2.145	.4501E-02 .1074	.9296E-01 2.927	.6414E-02-1.359	1.518 -353.5	THE C HATRIX3506	177E-04 .5556 261E-03 .5161	.1594 2885	501E-03 .7621	0.	.1541E-04 .1380	7	THE D MATRIX	'	480B-02	3046F-04 1180	.6918E-047215

POINT NO. 3 BASIC PIOO HODEL-NN=0.0, ALT=0.0, PLA=52- .5PC PERTX, 3PC PERTU, 8/28/75 PAA THE A MATRIX

4592	1.527	-,3176	1093E-01	0.1	7 19E	-20.11	6.250	.9260E-01	38.01	9.445	.4198		19	.2698	.3262	1.631	-2.154	-	4888E-02	.5541E-01		1287	.2234E-02	.2979E-04	-1.137	.9608E-01	.4290E-02	•	.5362E-02	-49.15
-1.772	2.651	4	3013E-01	m	-18.79	-1.366	5402B-01	028163E-03	-12.08	.4682E-01	01 .2161E-02	2017E-01	19.93	4.412	.3225	2.458	-2.917	1.949	-	-	-	3053	021792E-01	042389E-03	-2.705	_	2 .401	3E-0	•	25.29
-7.453	74.09-	-3.536	1174	-18.39	3.223	33.76	.4672	01 .6947E-	94.31	0069	01 .3076E-	.9487	2487	-15.49	3287	1.247	-1.606	.9555	01 .4258E-0	-4540E-0	-9081E-0	1003	011694E-	032259E-	9047	-4066E-0	-1830E-0	-19.74	01 .5421E-	27.71
-668.4	300.1	131.0	-7.997	35.63	84.69	-77.91	-3,358	4949E-	-689-3	7953	3535E-	217.4	01 2.580	192.6	9.431	10.55	-13.99	7.424	2-	-	-	1.	02 .1433E-	•	996.9-	.6163	-1.972	9	3	5.432
424.3	.2	-513.4	1 1.642	.9481	1.833	-2.023	0.	2 .0	-18.09	268.2	11.92	26.01	1 .9481E-(	161.7	7.316	1.169	-1.532	.8536	1	-	_	19173E-0	.1593E-0	-2123E-0	8058	n6.6h-	-1.997	0	8	•6036
4.230	-161.5	123.7	2 .7313E-0	1718.	1.593	27.97	2 .1857	3 .2751E-0	28.81	-72.14	1-3.206	-32.77	16190E-0	2.084	-2.837	-2.387	2.941	4.794	12875E-0	.2733E-0	.5467E-0	6038E-0	1 39.61	.5868	-48.43	39.05	1.735		1 .1632E-0	-1.910
.6312	4-336	. 1027	19503E-0	1080	2096	.2882	17574E-0	31122E-0	-7.098	13255	·	1897	2 .1944E-0	1.346	2.347	16.70	-10.28	2	2	1 .2645	_	Ţ	-1067E-0	99			#3	-4-3		4.555
5.	3.334		.1438E-0	.5691	.2215	6955	3012E-0	4462E-0	-6.150	2761E-0	1227E-02	1274	74 74 E-0	-2.242	8	9.759	.15	-1.527	9E-0	.2892E-0	.5926E-0	6590E-0	-50.00	6667	5796	-10-11	4493	5	0-344	

	1466.	.1341E+05	5224E+05	9190.	01-14.83	348.2	6.908	215.0	01-4986.	02-73.80	6.66 .5963E+05	-5377.	01-239.0	1	1 3	5532.
	1.257	-40.77	-10.98	650.6	6074E-	.3875	.8476	26.20	7300E-	-021072E-	-16.66	.6 180	•	1.084	.5025E-0	8.010
	-69-24	-11.52	46.63	14.81	.4032	3.436	7.824	-12.43	54 12	.8022E	-111.4	1285	5521E-02	.5439	2647	-56.15
THE B MATRIX	•	-1.722 261.7	2.547 -292.3	-1.191 241.0	0E-05-	E-01	1239 69.23	.5200E-01-49.02	562E-0167	.8242E-039465E-02-	10.88 -448.9	.7953E-01 17.80	.3519E-02 .7894	.8675E-02 37.31	5820E-02 3.943	8480 406.2

	5.72173411726	2537E-019849E-027724E-03	0. 0. 0.	.6674E-032589E-032032E-04	.3715E-02 .2227E-03 .1757E-04	2.035 1.852 .1587	.7243E-032195E-029049E-03	0. 0. 0.	.1908E-045758E-042389E-04	.1654E-04 .4983E-04 .2065E-04	
				.4335E-043321E-033825E-031482E-016674E-032589E-032032E-04	-034442E-02 .3325E-03 .1271E-01-	.9899 -1.086 .1080 .9769 2.035 1.852 .1587	4667E-034320E-024335E-036529E-035572E-027243E-032195E-029049E-03	1.000	1226E-041141E-031142E-041711E-041467E-031908E-045758E-042389E-04	-04 .9891E-05 .1484E-04 .1271E-03 .	
THE C HATRIX	.67333825	-3095E-01 .1651E-	0.	.2016E-03 .4335E-(	.1134E-03 .4019E-	.3804 .9899	4667E-034320E-	0.	1226E-041141E-	.1061E-04 .9870E-	

# THE D MATRIX -.1576 -223.4 20.19 .3842 1186. .1253E-02-1.166 .7331 -.7311E-02-6.208 .0 .0 .0 .0 .0 .3749E-04 .9215E-02 .1813E-03-.2197E-03-.1806 -.6068E-05 .4369E-02 .2033E-02-.7828E-02-.1086

BASIC

POINT NO. 4

P100 HODEL-HN=0.0, ALT=0.0, PLA=67- .5PC PERTX, 3PC PERTU, 8/28/75 FAA

3874 -1.628 3.013 -2.131 2941E-01 7102	.5527 .6096 -1.142 6055 .3168E-01 3333E-01 .7478E-01 .7478E-01 .2167E-03 .2167E-03 .2167E-03 .2167E-01 .2167E-01
7558 9276 2.735 5359 -4044E-01 .2652 4810 2061E-01 23206E-03 -5.504 5153E-01 19.89	-6694 -6967 -1.134 -9358 1.85218-01 1-25968-02 1-25968-01 2-68318-02 4-13668-03 1.34158-02 2-268318-02 2-28698-01
-6.646 25.91 -68.58 -7.600 -2.233 -18.90 2.222 29.02 29.02 1.232 1.232 -3131 -21.22	2480 8308 8308 3906 24808-0 24808-0 53948-0 13748-0 15508-0 0341338-0 27908-0 12408-0
-726.1 -227.1 552.1 87.71 -7.382 39.35 78.53 -75.12 01-4.445 036618E- -841.3 -9.260 -4119 188.2	11.64 3.889 -7.281 -3.862 -01.2041 8-01.2140 8-01.2140 8-01.2140 8-01.2140 1305 -1.994 -1.994 -2.892 -1.994 -2.892 -01.5872
428.3 -223.7 -20.21 -566.5 2.232 -6931 -1.553 1.476 -3209E- 15.63 234.7 10.43 21.56	8.502 -4379 -8206 -4234 1-25521 1-25521 1-53351 -3866 -1-999 -3.091 -1-23201
5.880 124.8 -160.5 1 121.1 2 .2287 1 .1685 31.62 -62.97 1-2.799 -28.66 1-9773E-	-3.188 -3.237 4.052 4.596 1514E-0 3129E-0 .3017E-0 .3017E-0 .3017E-0 139.52 -47.51 17.84 17.84 17.84
**************************************	3.629 12.78 -5.501 -5.638 11717 13863 13863 13590 661 3.794 -9.735 -4327 -4327 -4327 -4328
THE A MATRIX  4.132 -1.243 -5.243 5.002 5.780 -1.243 5.780 -3466 -4025 -2535E-0125328E-022328E-021569 -2109	1.166 10.20 -2.766 -2.766 -42078-0 -42058-0 -42058-0 -4335 -9.864 -4335 -17348-0 -17348-0

2	TRIX				
111	38.0	-106.3	3.619	2016.	,
4616	-137.0	-23.33	-77.02	6038.	
1,355	408.6	.19	.52	5727E+05	
.6243				-2131.	
2991E-01	Ţ			222.3	
.220 1E-0	_	8.338	.1616	271.5	
.5017E-01	122.8		2908	206.3	
1779	-75.42	-7.548	32.68	2466.	
.4516E-0	1.794	5636 -	2229	-5634.	
.6831E-03	.2378E-0	18480E-02-	1	-02-83.55	
7.200	-653.8			.7812E+05	
.1219	8765			-6134.	
.5432E-02	4241	E-01	4874E-02-273.0	-273.0	
.5094E-01		.4157	.3855E-01-2439.	-2439.	
.2444E-02	3.554	5694	.8113E-01-32.60	-32.60	
8299B-02	351.6	-89.02	8.981	1128.	
THE C MATRIX					
9	5442	5.667	00-68	120-0 8-0902001	1891 - 1681
.2601E-01	1 .3519E-03	F-036213E-03	48E-02	59 4643E-02	
0.	0.	0.	0.		
234E-0	.2592	E-044714E-04	.2244E-03-	222E-01-3417E-03-	56E-04 .1002E-04
.6857E-04	.3171	E-033571E-022378E-03	2378E-03	.1280E-012798E-02 .83	.8381E-041047E-04
.5140	1.614	-1.337	.1726	1.562 2.762 2.355	55 .2490
*8806E-04	.7925E-03	.6169E-04	.1178B-03	E-03 .1085E-03	E-04
0.	0		0.	0	
.6313E-05		.4538E-05	.8347E-05	289E-04 .7931E-05	.2635E-05 .1144E-04
\$657E-05-	5830	E-044729E-058736E-05	8736E-05	8227E-05-	1
THE D MATRIX	111				
2484	-355.2	33.28	7662	2130.	
7176E-04	1-1.581	979	1128E-02-1.128	1.128	
0.	0.		0.	0.	
8166E-052304	2304E-01		.4397E-065882E-01	58821-01	
.2165F-04	. 1297E-01		.1330F-028858E-022342	2342	

BASIC POINT NO. 5

F100 HODEL-HH=0.0, ALT=0.0, PLA=83- .5PC PERTX,3PC PERTU,8/28/75 FAA

5.727 -1745. -1392 -24.30 -6172E-02-1.082 -6777E-01 16.60 -1880E-02 9.147 -1677 435.8 -1677 435.8 -48666741	9.608 8.571 . 8215 . 6359E 8.940 . 2736 . 3980 . 3980 . 8241 . 89.94	653 940 160 989 989 624 632E-02 452E-01 984E-01	22 2 2 3 4 4 0 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5	10.52		# # 5 T
1 .2789 .0 4 .5496	E-05.0.0.0 .0.0.0 E-05.4790E-05.1478E-03	.0 .0 .1478E-03-	.0 .0 .1504E-01 .1618E-01	5545E-04 .0 6503E-04 1071E-02-	.4722E-04 .0 .8820E-04 9561E-04-	.0 .0 .4999E-05 5503E-05
.0 .0 .0 .0 .0 .0 .3434E-05 .2727E-3732E-05-, 2996E-	-1.863 .0 1.000 E-04 .1128E-05	1.863 .5709E-01 .0 .0 1.000 .0 .1128E-05 .4002E-05	.4815 .0 .0 .3673E-04 4024E-04-	3.428 .0 .0 .4290E-05-	2.161 .0 .0 .4958E-05 .5324E-05-	.7687E-01 .0 .5609E-05 .6103E-05
THE D HATRIX6777E-01-420.5 -1282E-03 .3353 -0 .0 -1030E-051193E-8109E-05 .2328E-	32.97 -1 .6804 .0 .0 .E-015806E-02 .	-1.824 5605E-04 .0 .6015E-04 5538E-02	1245. 1199E-01 .0 .4463E-01			

10/30/75 PAA P100 HODEL-HN=0.0, ALT=0.0, PLA=24- GROUP 1 POINTS |

-1.902 1-5826 -3	90		522.5	-891.2	-13.51	-2.617	7219
1.271	5.0	0.2	-	287.3	-48.05	2.826	2.219
.9850B-01	.12	-28	-	366.9	-3.652	6104	9787
10E-0	10	548	•	-15.70	. 1206	.6570E-03	.2048E-02
-2979	2366	286		86.04	-16.89	*8094	.1308
8	4267	844	•	117.4	4.838	-18.88	.1793
.4201	5151	-47	•	-94.50	33.59	8947	-20.14
.8664E-02	10	. 1406	4244	-2.242	.3396	2509B-01	5.909
1277E-0	1531	.2186E-02	S.	3363E-01	-5025E-0	23345E-03	.8754F-01
3.715	4.2	46.69	3	-836.2	144.7	-7.927	90-04
.4378E-01	94	-132.4	448.5	646.6	-1.586	.1087	6.097
O	4223	-5.887	9	.4428	7056E-0	1 .4850E-02	.4043
.1076E-01	998	-59.57	S	544.9	.2733	4515E-01	# 00° #
4560E-0		6559E-01	.6790E-01	2.747	2882	6.	.7946F-01
1.742	4	4.263	348.4	109.0	-22.22	3.056	9691.
5	-1.536	•	3,382	0.4	-,3331	-1.217	-,3978
.92	~	-1.985	-2159	-1.905	1860	5417	2785
.3156	.0	2.408	8444	00	.3893	1.192	.5854
	.0	2.269	-,3585	-3.146	2855	4583	0644-
94E-0	.3789E-0	11259E-01	.5485E-02	.4794E-0	1 -2889E-0	1 .3221E-01	.6687F-02
.7535E-01	-7304	.8406E-01	.1088	979	.9170E-0	1 .3050	.1432
8	-	1911.	.1563	1.407	.1289	.4350	.2057
5	8173	9408E-01	1227	.10	1029	3450	16 14
00		39.77	2159E-02	1943E-0	9	28333E-02	9
299	6999	. 5892	0	2591E-03	8651E-0	1111E-0	3789E-04
11			-1.099	.87	9235	-3.092	444.
.15		39.39	-49.97	.5830E-0	9	•	B-0
13	#		-1.999	.99	730E-0	3 .0000E-03	9
597	•	-	-1.961	-2-037	6.67	•	273E
		.1280E-01	.6046E-02	388	1 .6056E-0	2-19.99	9
0	4.159	8	2	4.761		27.30	-49.31

	-1954.	.2008.	1195E+05	-1185.	0		841.1	-2187.	-3406.	-02		-4161.	-185.1	-1677.	35.05	3470.	2.88229073434E-019015E-02-	0. 0. 0.	25988-019364E-03- .1982E-015016E-02	10 a 20 b	2005- 201-1041E-02-3505E-02		E-043044E-032840E-049548E-044451E-04	.1836E-04 .1731E-03 .1991E-04 .2594E-04 .2326E-03 .2185E-04 .7328E-04 .3401E-04		-117.6	.1715E-01-7.955		
	6.693	-48.98	49.76	1.899	·	-1.206	-3.331	9.431	01-,3334	034733	-138.2	2.518					58.58 012769	0.	037572R-03	7256	1230	0	04-,3393	04 .25941		-3.216	.1715	•	17841
	969.6-	-3.461	8.652	1597	.9223E-01	.1317	1.178	-3.046	64 16E-01-	E-019223E-034733E	-24.63	.2894	_		4938E-01	-14-64	2.427 58.58 251.0 E-022548E-012769E-019525	٥.	E-046946E-03- E-039006E-02	9904 - 10-4	02- 9534P-	1.000	-032603E-	-03 .1991E-		1.949	.1428	•	COLTUNE COLTANDEC -COLTCOC
	-18.75	1-95.98	66.19	-53.43	1-26.70	7.355	21.17	33.76	1-1.911	22880E-	-200.1	-1.063	24239E-0	1-6-475	578E-014221	90.18	2993 2772	•	.5957	44.93	8296	0	42262E-	4 .1731E-	414	1-58.47	21093		
THE B HATRIX	1.124	3672E-0	-1.003	1.010	150 1E-0	3907	5160	.7454 33.76	.8762E-0	.1277E-0	23.61	=	8	8	-		THE C MATRIX -2231 1786E-01 .	0.	.3223E-04 .8807E-04	0-27030	- A795P-03		2407E-0	. 1836E-0	THE D MATERIA		0-3080 h	•	44068

10/30/75 FAA P100 HODEL-BH=0.9, ALT=10K, PLA=83- GROUP 1 POINT 2

	.2338	-2.423	5.023	-4.811	71 E	.3096E-01	20	20.05	6.463	.9576E-01	36.21	9.587	.4261	4.438	12	36	.8745	.1503	.1196	.2.890	.4782E-02	.3143E-01	.5602E-01	.5124E	.1196E-0		.6781	.1196E-01	199h	.5397E-01	.4099E-02	49.12
	1.731 -	- 5953	3		-02	-0-		÷	-05	#	3675	0	-05	.1118	19.85	1.346	1.303	5430E-01	•	- 610	E-01-	.1062	.1822	1762 -	2413E-01-	3620E-03-	-2.211 -	1810E-01	39E	.5912E-01	9	- 61.27
	-4.635	34.58		.51	.2219	-19.11		28.71	.7044	1 .1045E-01	107.2	2.162	.9612E-01	1.727	43	-25.95	7442E-01	.1302	7973E-01	-2.174	1 .5971E-01	.2551E-01	.4359E-01	0	9	ō		9	595E-03		18	30.83
	-722.1	-306.0	761.4	39.44	-9.116	37.67	71.98	-76.29	11-6.556	129681E-0	8.606-	-18.10	8041	152.0	1 3.879	252.8	15.36	1.190	18082		•						-4.456	7857E-0	3	-2.761	12 .2694E-0	5.848
	425.6	3	046-4-	-617.6	11.764	.2517	.4847	4474	019321E-0	021554E-0	-5.476	192.5	8.558	8	72	3.	10.20	.1073	8980E-0	-2.110	013681E-02	_	-	013741E-0	3741E-0	4989E-0	4951	5	•	-2.970	3	9649
	7.498	127.0	-161.5	_	_	-	-	_	_	E	15.99	-51.44	1-2.286	-23.28	5	1.690	-3.698	-4.077	5.747	689. 4	14015E-	. 1838E-0	.3321E-0	3256E-	1 39.52	.5855	-48.21	38.58	1.715	17.70	1 .1967E-	-1.935
RIX	.6313	-8-077	698.6	541	22	0	9553E-0	23	-	3-,6507	0	15	22	2968	2 .4	2.920	5.492	12.52	+	-17.56	2-	1.1555		ī	59 14 E-0	6676	-3.380	-9.589	4262	•	~	0
THE A MATRIX	-4.236	5463	1.383	-2035	.7205E-0	.9160	.5693	144	133	1970E-0	-1.723	3289E-0	14 59E-0	1588	7627E-0	-	5	11.86	8448E-0	609-7-	-7040E-0	.1652E-0	.2966B-0	292	-50.00	9	3483	9	42	3	-2253E-0	0

		7.601 118.0 -7.334 14.73 1.1793184E-01 F-042305E-03.04416E-011041E-02.02764E-04 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	3.626 -1.978 .4474 3.973 4.400 3.615 .6078 41479E-031410E-042183E-041964E-032326E-041131E-032989E-04 0 1.000 .0 .0 .0 .0 54171E-044939E-056126E-055492E-046761E-052762E-048432E-05 5 .4311E-04 .5112E-05 .6345E-05 .5683E-04 .6988E-05 .2843E-04 .8715E-05	
5097. 4403. 6752E+05 1981E+05 -87.30 1311. 1725.	5053 -6694. 27564E-02-99.16 -63.92 .7168E+05 -1.041 -7062. 24607E-01-313.5 1674E-02-2409. .3525 88.27 30.87 .1342E+05	7.601 118.0 -7.334 14.73 F-042305E-03.04416E-011041E-02 .0 .0 .0 E-047214E-046668E-041144E-012710E-03- E-032355E-02.6850E-04.1172E-012248E-02	3.973 4 8-041964E-03 0 8-055492E-04	24.98 -2.859 4216. .58105876E-01-32.10 .0 .0 .0 E-025207E-024325E-032689 E-033627E-037274E-025178E-01
4.493 -121.0 47.27 -23.67 -2666 2.208 2.962 36.34	5053 5053 -0275641 -1.041 -0246071 -16741 -3525	118.0 -03.0 -04-66681	1.978 .4474 1410E-04-2183E-04 1.000 .0 4939E-05-6126E-05	24.98 -2.859 4216. .58105876E-01-32.10 .0 .0 .0 E-025207E-024325E-032689 E-033627E-037274E-025178
1.963 1.963 -10.41 -35.82 22.58 38.24	1190 1697E-0 2.750 .2022 .9326E-0 1.256 6518	7.601 F-042305E-03 .0 E-047214E-04 E-032355E-02	-1.978 E-031410E 1.000 E-044939E	24.98 .5810 .0 -025207E
THE B HATRIX6519 -576.08367E-01 603.12024E-01-1150. 2.271 14414431E-02-101.82508E-01 18.005665E-01 3.385		THE C HATRIX -1567 -1-290 -1309E-01 .7052F -0 .0 -4517E-04 .1827E -2218E-04 .2614E	.9066 3.626 1643E-041479E .0 .0 4368E-054171E	THE D HATRIX4437 -8.4939938E-049106 .0 .0 .6789E-05 .1545E

	1000	056	9949-	1846	3179	17005E-02	2586F-01	1724E-01	-19.98	1 5.997	3	41.68	1 9.182		_	.7652E-01	4769	1 .2602E-01	1 .6505E-02	ī	٠.	25794R-02	13122E-01	-	1 .2212E-01	1 .6505E-02	3 .8673F-04	.1886	2 .0		_	204E	-50.07
10/30/75 FAA	930	075-1-	0413	1.415	.1037	011584E-0	8789	-19.07	8393	28291-0	0.24401E-0	-7.478	-8016E-0	01 .3584F-0	1320E-0	19.92	2.391	8459E-0	3760E-0	_	_		_	Ϊ.	013948E-0	02 14 10 E-0	042506E-0	013102	01 .4700E-0	03 .1880E-0	•	05-20-00	27.03
e	2002	107-1-	17.04	-26.12	-2.051	-4766E-	-17.06	5.208	33.74	.2471	-01 .3605E-	140.6	-1.855	8260E-	1156	2610	-18.93	1375	0. 10	1473E-0	.7365E-0	-01 .8443P-0	1375E-0	1571E-0	.1277E-0	-012455E-0	-033273E-04	-9820E-0	.1719E-0	-6874F-0	-19.67	-01 .3528E-	23.83
GROUP 1 POINT	0 607	6000	-87.35	183.0	221.7	-13.45	87.40	119.6	-108.8	-2.886	-024233E	₩-896-	12.70	.5657	7.644	2.540	128.4	2.350	-02	-01-	0.	-023956E	-012089	7	-01 .1479	-02 .4351E		1.305	.2393	-1.990	-2.062	-05	-012004
PLA=24- 6	7 63 8	0.104	-5779-3	-1.642	-526.6	1-02 1.491	-1.768	-1.642	1.314	1642	3-05-4380E	11.33	785.7	34.92	19.46	.2628	459.2	2.185	.4835E	2417E	0.	E-014417E-0	S-012321E-0	2-011740E-0	-02	•	-6446E	.1450	16.64	-1.999	-2.172	E-01 .3868E	5560E-0
3, ALT=20K,	000	507.7	110.8	-125.7	E-01 97.64	-02	1.272	2.174	75.69	E-025743	E-0488361	83.15	-227.6	E-02-10.12	-102.0	E-011590	1.325	9037	-1.080	1.150	1.400	E-011146F-0	1049E-0	1295E-01	.9253E	R-01 39.72	.5885	-47.98	39.30	1.747	17.70	E-01 . 12951	-2.333
MODEL-MN=0.	MATRIX	• •	7	7	-01 .7393	-025295	•	•	2819	2 . 1835	1 .2447	7	E-01-	i	·	E-02 .2265			3.021	ç	•	P-03-	-01-	IE-011028	61.	•	•	•	-10.32			E-02 .25	.2571
P 100 M	THE A	2000	3010	.7156	.7305E	.3921E	.3639	.3112	4587	6601	9975E-04	-3.787	.5501E-0	.2447E-02	1144E-01	5105E	-1.632	.2571	2.996	1428	6891	8470P	1714E-0	1143E	.1214E	-50.00	6666	.1071	-10.41	4627	-4.683	.2856	.5659

											4E+05						
	-3.758	1953.	-7988.	218.9	2E-01-16.82	-19.88	91.95	-776.7	-3140.	-02-46.67	944	-4031.	1-10	-1698.	1-50.3	-138	
	3.774	-20.76	16.11	1.111	574	-1.718	-2.728	12.02	-011722	-032829E-	-80.19	1.576	-01 .6986E-	.4512	-01 .6150E-0	12.07	
	-4.147	-1.580	401.4	.4572	.2289E-01	4857	.8537	-2.848	5831E-	018836E-	-21.37	.3194	11 . 1409E	.2886	2	-12.79	
HATRIX	-01 8.327	-48.51	23.16	01 .8543	-01-14.56	02-1-206	-11.97	42.63	-1.454	-02 1874E-	-147.2	-2.176	-019751E-C	6207	-02 .7839	27.39	
THE B HA	6024E-	6773	5044.	-3715E-	-2263E-	-7895E-	.1165	.5516	.1255	-1930E-	36.48	.2846	-1283E-	11194	4270E-	.4198	

.6762E-05 .6115E-02 .7587E-04 .7230E-03 .6169E-04 .1088E-03 .9790E-03 .7672E-04-.2291F-03 .1382E-03 -.6563E-01 -.2397E-01-.5564E-02 .1212E-03 .9147E-04 .3515E-03-.1561E-01-.2887E-03 .2321E-01-.4906E-02 .1789E-03-.3868E-05 .9079E-04 .1123E-03-.5881E-03 .4868E-03-.4033E-01-.1354E-02-.3108E-03 .4483 -.1659 .6875E-01 .3945 -.1717 2.854 55.38 257.9 .1986E-02-.1049E-01 .8828E-02-.7216 .8171E-02 .7969E-01-.2680 1.000 .4168E-01-.1717 THE C MATRIX .1128E-01 .7537E-01

THE D HATRIX
-.2620E-01-44.62 .4012 -1.322 60.79
-.8102E-04 .4178F-01 .8156E-01 .1339E-01-.2272
-.0 .0 .0 .0 .0 .0
.6072E-04 .4127E-02-.3756E-02 .7574E-03-.1401E-01
.1753E-03 .1456E-01 .3869E-03-.1186E-01-.3047

10/31/75 PAA P100 HODEL-HW=0.6, ALT=10K, PLA=20- GROUF 1 FOINT 4

1111 111	01 102.2 022083E 4.047 63.41 021133 041510E -215.8 01-95.97 01-95.97 01-2567 -1.182 1.630 1.630	2.18 2.1406 2.1406 33.61 692.4 30.78 89.11 2.566 -9217E-0 -1892	1 1111 1 1 1	-35.12 -1.640 -17.99 7.122 34.52 34.52 -2577 -3836E-0 149.8 -3.337 -1482 -2113 -1435F-0 -9089F-0 -1626	- · · · · · · · · · · · · · · · · · · ·	
	5928B-0 5928B-0 5928B-0 5911 5911 5911 5911 5911 5911 5911 5911 5911 5911 5911 5911 5911 5928B-0 5928B-0 5928B-0 5928B-0 59311 593	1.9120E-01 1.9120E-01 1.1698E-01 2.587E-03 6.500 -1.999	1.6374 1.6374 1.1528 3.2329E-C 5.850 -1.992	-8706E-0 -8706E-0 -1674E-0 -1674E-0 -6123 -1914E-0 -19.89 -9568E-0	1-6822E-01 1-5598E-01 3-3499E-03 5-204 1-26248-01 3-1225E-02 5773F-01	11-1210 11-1210 11-2253E-01 13-3434E-03 13-3434E-03 11-1609E-01 11-1674E-01

											E+05								2283E-014060E-02	0.	35E-02-2421E-03	1	10-10-10-10-10-10-10-10-10-10-10-10-10-1	E-02 2990K-03 2624E-03	0-	1796E-04 1561V-04	2512E-042195E-04-					10-4		
	771.2	1869.	-9384	624.8	-35.86	-99.32	-331.3	-157.1	-2741.	-40.57	.4 187E	-3438.	-152.5	-1420.	-33.52	-393.4		286.2	1	0.	1774	.24931	1476						21 58				7160	20110
	18.33	-57.77	19.44	2.235	11497	-2.464	-9.041	26.38	13436	34921E-02		13.64	6909	5.319	1.1921	22.69		59.29	1 .1664E-01	0.			1640F-01				1		-2 851	2372E-01	0			34320E-02
	-8.297	-2.699	6.004	.9863	.2271E-01	3962	2.422	-2.838	1592E-013436	E-022350E-034921E	-25.29	.6547	1 .2917E-01	.5280	1937E-01	-11.56		4.026	123E-0	0.	E-036683E-03	E-022340E-01	2830	2 -2050E-03					- 8135	1140	0	P-02- 3251P-02		
RIX	3.121	-26.06	7.063	1 24.51	1-19.72	1-4-044	-7.527	16.91	6209- 1	.8280	-102.2	-1.168		2.256	1 1.228	-10.12	RIX	2878	.2135	0.	.1273	.1464	11.21.1	2128E-0	0		4 1768E-0	:	1-38 00	2 4642E-0	0	1677	2691	• • • • •
THE B DATRIX	7631	8525	1.161	E-0	.3518E-01-19.72	.0000E-01-4.044	.3526	1024	E-0	.4929E-03	28.81	9661.	.8731E-02	.1072	.1767E-0	9299"	THE C HATRIX	1298	.1588E-01	0.	3491E-04	.1819E-03	-9032E-01	.2364E-03	0	14062-04	1953E-041768		- 64338-01-38 00	1549E-02	0	5216P-04	20777-03	231115

10/31/75 PAA PTOO HODEL-EN=0.6, ALT=30K, PLA=24- GROUP 1 FOINT 5

-1.216 -2750 -2734 2 -3222E-01- -5795E-02- -5009 -6217 -9368E-02- -1399E-03-	×	7.895 103.8 124.0 87.67 .1213 .7895 1.854 101.8	8-03	-624.4 -75.12 209.1 96.38 -8.405 61.52 143.6 -166.4 -1.772	-3.992 9.883 -21.25 -2.973 .4969E-01 -18.40 4.452 32.68 .2455	7985 7421 2.039 -1.146 4079E-02 -1.601 2349E-01	.4683E-01 -6244 -9314 -1.056 -8697F-03 -0 -8326E-02 -19.99 6.088 -9019F-01
	632 253E-02 253E-02 5508 053E-01 520 688 688 688		99 '	יששחש שרדש	-1.320 5854E-01 2242 2242 -14.45 1448E-01	.1033 .1033 .4697E 19.96 4.387 .3370 .1685 -1.283	9.275 -4123 4.234 .6973E-01 .2264 -1940E-01 -1293E-01
.3923E-02 .6787E-03 .8823E-02 .8823E-02 .50.00 .5599E-01 .5599E-01 .5599E-01 .9909	.1861E-01 .0 .4887E-01 .8551E-01 .0 .0 .5039 -10.49 -4.661 -4.563	62938-02 58568-03 35148-02 76138-02 39.68 -47.95 -47.95 -1.748 -1.748 -1.878	.2498E-02 .9239E-03 .1201E-01 .0 .3080E-04 .7853E-01 .2.600	.2276E-01 .0 .1164 .0 .2772E-03 .2079E-01 -2.001	1931E-02 1931E-02 723E-02 3218E-04 3218E-01 7241E-02 2896E-63 -19.75	. 1888 E-01 . 5242 E-01 . 8425 E-01 . 0 . 0 . 0 . 0	.3506E-02 .1293E-02 .1552E-01 .0 .4312E-04 .1067 .9701E-02 .3880E-03

											E+05						2.4589573E-01 .1613E-01	70 71 10 10	)35E-02-,3794E-03	.6038 .7738 .1269	.24 14 E-044037 E-03	0.	.1450E-052363E-04	.1857E-041049E-05 .1780E-042590E-05		E-01
	1442.	h . 469	-7051.	-3273.	16.44	-22.87	-93.22	83.93	-3345.	2-49.63	.4296E+05	-4309.	-191.4	-1731.	8.947	3638	211.6		3936	.8165	.4157E-03	•	.2507E-04	1857	529.7	.2173E-01
	5.356	-25.41	94.6	-2.437	2766E-01	-1.430				3981E-02	-109.8		10-3	4551E-01-1731	2243	18.09	80.78 211.6			.9198E-01		0.	.2628E-05	1929E-05	-1.534 .1589E-01	
	-6.297	-2.301		-1.792	E-01		2.738	-4.855	9447E-012701	E-011486F-02	-35.29		.1727E-01		6084E-01	-14.80	5.068	0	E-04-4267E-03 .2237E-03 E-03-1722E-01-1686E-03	2795		1.000	1.1716E-05	.1312E-041222E-051929E-05-	.4392 .1066	3584E-022774E-02
IX	24.16	5.530	-8.447	-107.2	-13.48	7.340	12.71	-1.005	1.067	.1083	18.52	-8.324	•				ma		.9271	.6814				1	RIX -37.42 39282E-01	3584E-02
THE B MATRIX	33	.1123	5150E-01		-	2522E-01	.1286	3E-01	.1499	.2340E-02	40.83		1560E-01	7E-0	1522E-02		TBE C MATRIX .13818-01194		.8356E-04	.1536	.3181E-04	0.	.1930E-05	1375E-05	THE D MATRIX49394459E-03	716E-04

12/01/75 JHK P100 HODEL-HN=1.2, ALT= OK, FLA=83- GROUP 2 POINT 6 THE A HATRIX

F-01 F-02 F-02	10	000000000000000000000000000000000000000
-1.149 -3.241 -3.181 -3.998F -20.01 -20.01 -8949F	4.154 9.067 9.067 9.8808 9.2708 3.565 5.683	3099E-0 -2118E-0 -7059E-0 -7059E-0 -6706E-0 -6412E-0 -3318E-0
0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	010	
Del	M M M	ымими мым
-3.401 -2.813 9.808 -6704 -79061 -19.20 -9764 -57701 -6.256	8	- 4 - 4 4 4 A M
0-30		
24.87 24.87 -4.858 -7.858 -18.43 3.738 26.06 .3599 .3599	37521 37522 37522 37523 37523 37523 37523 37523 37523 37523 37523 37523 37523 37523 37523 37523 37523 37523 37523 37523 37523	
6	6 11 1 11	
-555.3 -58.70 193.7 18.57 -9.460 -9.460 14.43 -18.34 -1.146	1.681 103.8 103.8 4.586 34.05 17.43 5.876 6.206	1895 62748-0 62748-0 33618-0 14818-0 14817-0 1486 1210
-555 -58- -9-4- -1-1- -1-1- -1-1- -1-55		
400.2 -227.5 -8.815 -574.5 11 1.556 -1233E-01 -4315E-01 -4315E-01- 11-7705E-01- 2-1027E-02-	40 0 fee 6	2142E-01-1494E-02-6971E-02-3734E-02-4979E-04-4979E-04-4979E-01-4979E-01-49-99-99-99-99-99-99-99-99-99-99-99-99-
400.2 227.5 8.815 574.5 11.556 1233F 6781E 4315E		- 2 142 E E E E E E E E E E E E E E E E E E E
9 97	90	5000
1.608 117.2 165.9 132.7 2871E 3571 8742 13.56	-34.73 -15.71 -15.71 -3232E-4040E-5.972 -6.033 8.344	. 7268E-0 . 9694E-0 . 6786E-0 . 5888 . 5888 . 48.05 . 1325E-0 . 1325E-0
- 00	11.1. 1.1	NEEN
113 117 117 118 163E-0 151 183 183 183 188E-0	37 16 E 16 E 16 E 16 E 16 E 16 E 16 E 16 E	208 218-0 218-0 218-0 3308-0 564 564 565 564 565 564 565 564
24.8	12136	0 - 4 - 6 - 4 - 6 - 6 - 6 - 6 - 6 - 6 - 6
H - 0 - H - 0	F-01	E-02 E-03 E-03
12382386	7815E-01 3469E-02 1728 2783E-02 8694 1.761 18.12	-1508E-02 -7493E-03 -5245E-02 -3746E-02 -50.00 -6666 -3372E-01 -9.778 -4.468
5388		

																	12.67 -1.1485475	E-02 1484E-02		94231-04-	.1601E-02 .1196E-03 .5299E-06	6.326 5.894 .3660	1094E-03	0. 0.	.1089E-057625E-05-		
	3512.	.1218E+05	1076E+06	3863.	-246.6	-33.10	3.841	-195.8	-6007.	1-88.62	.7138E+05	-6501.	1-288.9	-2558.	15.53	.626.	123.7	2828E-01-	0.	1782F-02-	. 2260E-021601E-02	2.248	0.	0.	5142E-05-		.1148E-01
	47.14	-67.76	162.1	13.72	7647					10-		-1.458	E-0		.3212	24.21	110.0			45267E-05	2 .6998E-05	.2561	0.	0.	65713E-06	-22.03 .1340E-01	.0 .8868E-03 2.4514E-04
	-235.7	-84.82	261.7	22.15	.7506	5.542	17.94	-22.96	-1.348	E-011978E-0	-197.1	-2.320	1029	5991	1980	60.99-	6.678	E-031535E-02	0.	45634E-01	.2876E-031674E-02	-3.261	0.	1.000	55049E-06	-3.267	.0 .1309E-025808E-02 .808&E-02 .3202E-02
MATRIX	8-981-	-141.2	359.5	42.58	-	1 80.14	2 20.60	2-32.57	1-2.252	33453E-0	-332.4	1-2.634	2 1205	4398E-01-17.33	3 .3066	_	MATRIX	1 .6364	0.	#		2.668	0.	0.	63664E-05-	-334.0 -31.625	
THE B MAT	3316	2458	.5924	.1620	.1476E-0	.1899E-01	.2327E-02 20.60	8008E-0	-3050E-0	.4455E-0	400 4	.8116E-01	.3604E-021205	-4398E-0		9233E-0	THE C MAT	.3411E-0	0.	-4626E-0	.8816E-04	1.070	0.	0.	4070E-06	THE D MATRIX1618 -3	.0 .6408E-06 3639E-05

P100 HODEL-BN=2.2, ALT=40K, FLA= 83- GROUP 2 FOINT 7 12/04/15 JHK

. 6887		666.1-	3.842	-1.664	5110E-01	0.	.8639E-02	-20.00	1 5.854	3 .8674E-01	38.70	1 8.820	2 .3920	14.011	.8063E-01	3974	.13728-01	.1029E-01	1029E-01	15489E-01	15160E-01	1 .2745E-02	1 .1166E-01	12745E-02	.85778-02	.13721-03	3088E-01		3 .4803F-03	.6175P-02		66.64-
-2.562	•	00/-	•	.4322	021013	.4209	-19.06	-1.057	4374E-0	0-3	-7.690	5660R-0	012573E-0	8541E	19.92	2.081	.1281	-011230	-	-	÷	2	-01 .8508E-0	-018713E-0	-0.2.0-	0. 10	6202	-02 .768BE-0;	-04 .3075E-0	1066	-20.00	24.94
-5.80k	200	711.0	-37.20	-3.319	-6884E-	-19.19	2.042	27.15	.2957	01 .4384E-	47.16	.5021	-2233F-	.5373	1060	-7.043	8 183	2	01 .9121E-0	-4430F-0	.1542E-0	-02 .8860E-	-01 .2398E-	189E	-012606E-	-023475E-	1511	1303E-	212E	-19.72		01 25.54
4 474-	04		#		-10.99	18.97			-2.076	023104E-	-348.1	-3.460	1560	192.7			-	62	01-	01-	1	Э.	2 .6245E	21784E	2 .7806E	.1190E	12	.1227	-1.995	78	•	2
432.3	2000	-734.1	3.293	-599.8	1.085	.2839	.6927	6813	1136	021514E-	-4.883	235.5	10.47	19.40	01 .0	132.9	7.690	-7435E-	1487E-0	4213E-0	013701E-0	-02 .9913E-0	•	1	.6 195R-0	.9913E-04	1611E-0	66.64-	-2.000	-2.800	01 .4556E-0	
2 697	2001	7.001	-161.2	134.3	1	-	.9011	27.78		4 .2668E-	19.54	12.49-	1-2.856	-29.08	24446E-	*8004	-3.046	-3.426	01 4.750	4.956	5634E-	23232E	13878E-02	-	1 39	.5904	90.84-	39.06	1.736	17.76	.1164E-	-2.221
MATRIX	2000	-3.343	3.687	2896	2-	7043E-0	1728	.6111	_	3 .5425	-4.067	1.	- 1	•	2	.6388	.9926	9.707	ä	-3.079	Ť.	.672	.471	2	.5888E-	6658	111514	-9.623	428	-4.391	0.	.6729
THE A HAT		16401	.0	_	BE-C		.4413	8768	3770E-0	5596E-0	-6.382	5705E-0	2543E-02	1987	3217E-0	-1.872	.1757	0.670	-7477E-0	-2.548	1984E-0	.2617E-02	.8225E-02	5608E-0	-49.99	9999	3458E-0	-9.741	m	-4-430	0.	.5654

	1112.	4756.	5543E+05	2891.	-448.2	-14.81	87.01	-157.7	-5642.	-01-83.54	.6873E+05	-6394.	9.487-	-2726.	2.722	1870.	
	112.4	-461.5	713.0	60.42	-3.497	-15.62	-38.83	-1.921	-5.403	.7990E	9.006-	16h.8-	013777	-9.650	2.034	136.3	
	-93.80	-25.28	85.05	6.717	9661.	3.978	13.43	-14.56	6188	019185E-02-	-106.3	9116	4683E-	-1.052	2007	-52.40	
THE B MATRIX	5490E-01-252.5	.5930E-01-134.4	1354E-01 252.4	.9604E-01 27.09	.4316E-01-96.64	7963E-02 .9667	.8385E-02 40.54	.1638E-01-19.69	.4716E-01-1.427	.6957E-032144E-0	7.574 -335.3	.1630 -9.786	.7274E-024388	.1001.	1843E-03 1.366	.3634E-02 146.3	

	9510	12 .0	0.	34218E-07	13 .6327E-07	4759	3388E-041472E-03 .5757E-042168E-041952E-031401E-035958E-033002E-04	0.	053937E-06	3183E-06	
	-1.914	1456E-017232E-02 .0	0.	-031262E-0	-02 .1233E-0	3.054	-035958E-(	0.	-029848E-(	-05 .969BE-(	
	471.9	1456E-	0.	-022473F	-021716E-	3.560	3-03 1401E	0.	3-05-2348E	2-05 .2271E	
	129.5 9.744	E-023390	0.	E-045760E	E-04 .5595E	-3.080	E-04 19521	•	E-0616341	E-06 .1742E	
		5E-024578	0.	BE-037763	2E-02 .7679	93418	7E-042168	0.	7E-051815	6E-05 .1513	
	6295 4.665	12E-02610	0.	20E-04 106	10E-03323	74 -2.50	72E-03 .575	1.00	32E-05 . 105	14E-05102	
THE C MATRIX	845662	3033E-01 .1212E-026105E-024578E-023390	0.	.1471E-04 .2120E-041068E-037763E-045760E-022473F-031262E-034218E-07	1025E-03 .27	3203 -1.6	3388E-04 14	0.	5202E-061232E-05 .1057E-051815E-061634E-052348E-059848E-053937E-06	.4746E-06 .1314E-051026E-05 .1513E-06 .1742E-05 .2271E-05 .9698E-05 .3183E-06	
	i	•	•	•	•	•	1		•		

	-3838	4829	0.	.2209E-02	.3185E-01
	-149.2	.2736	0.	.4735E-02	.1585F-01
	-12.96	.7381	0.	.1337E-063631E-022065E-02 .4735E-02 .2209E-02	1702E-05 .5823E-02 .1704E-02 .1585F-01 .3185E-01
YT	216.5	12257	0.	3631E-0	.5823E-0
THE D DATKLA	.4252	1945E-032257	0.	.1337E-06	1702E-05

F100 HODEL-HN=0.9, ALT=45K, PLA=130- GROUP 2 POINT 8 12/04/75 JHK

	.3507 .1740 .4065 .2420 .9019E-02 .4975E-01 .1430 .1291E-01 .1290 .1291E-01 .1900 .3602E-03 .1604E-02
3481E-01- 7728 -1.873 3205E-02- 2117 20.49 6781 5918E-01 6819E-03 9.009 39.009 1476 1476 19.79	-3245 -1795 -9004 -3882E-02 -9389E-01 -1270 -1270 -3222E-03- -3222E-03- -1785
-2.212 -23.13 -1.538 .7028E-01 -19.70 30.11 .6771 .1002E-01 110.2 2.457 2.457 -53.83	-1608 -1298 -3922 1-74998-03- -2935E-01 -6546E-01 -4966E-01 -4966E-01 -19652E-01 -1552E-01 -1552E-01 -22230
-738.3 -421.0 1001. 179.0 -10.54 142.5 -354.9 -425.6 -4713. -4713. -4.543 446.7 22.85	1.451 -01-4280 .9512 3.472 -0211518-0 -01 .2045 -01 .2901 -022378F-0 -0431718-0 -4.055 -71348-0 -2.003 -3.453
387.2 -253.2 29.84 -529.7 01 1.363 6.701 10.88 -10.25 -1.282 -1.2	2.669 -4756E -1030 -3963 2-1415E 1-2272E 1-328E -5284E -5284E -5284E -5284E -5284E -73289 -5000 -3289
11.169 129.0 -157.2 1 136.8 2 .7127E- 1 .9158 1 .8282 56.59 72.73 -191.1 1-8.492 -86.79 -40.80	-1.105 -1.096 1.433 1.833 29063E-0 2315E-0 .2213E-0 2.39.26 .5816 -47.29 38.24 1.700 17.46 1.2076E-0
22548-01-4783E-02-4783E-02-4783E-02-4783E-02-22548-01-4965-01-4965-01-4965-01-3417E-	6815 2-798 .7532 1-883 275408-0 1 .2582 12331 8967E-0 6668 -3.192 -9.406 4182 -4.537 2 .1435E-0
-1.298 -1.298 -34.28 -2337 -3172E-02- -8278 -1207 -1026E-01- -1535E-03- -15107 -1591E-01- -1591E-02- -1591E-01- -1591E-01-	.1494 3.098 .1116 -4762 .2630E-01 .4025E-01 -366E-01 -50.00 -6667 -4891 -9.412 -4183 -4.335 -4.335 -4.335

	-5088-	328.0	1585E+05	5191.	01-17.39	542.7	1047.	4.26.6	-7026.	-02-104.2	.7594F+05	-7365.	01-327.5	-3301.	58.40	.1022F+05
	-1.043	-17.07	14.54	6.119	6020E-	.2000	1878.	96.01	9964-	3305E-027409E-	-67.84	-1.457	-016481F-01-327.	-1.874	.3818	40.78
	-34.63	-2.544	5.937	7.029	.1719	10.46	13.34	-2.053	2224	-013305E-	-26.76	5694	2558R-	1.914	7587	-132.9
ATRIX	23.54	-8.790	15.77	-70.00	-02-15.49	6E-01-19.83	-12.66	.1801E-01-2.816	2.701	-02 .3777E	-22.15	7.633	-01 .3498	17.33	-01 .9643	-259.5
THE B MATRIX	.5408	.2440	2194	-1.542	-5759F-	9816E	-,3211	1801E	.1960	-2900E-02	25.08	.5287	.2356E-	-5047	15 16 E-01	-5.608

4450	8853E-05	5334E-05	5576E-05	9348	59478-05	0	24 97 P-04	2663E-04
2737 - 8859	30461-04-	1182F-03-	1257E-03	4220	1726E-04-		2370F-04-	24831-04
	70-	.0 .0 .0 .0 .0 .0 .0 .0 .0 .0 .0 .0 .0 .	.2258E-04 .1347E-031001E-01 .1942E-02 .bbu0E-011272E-021257E-03 .5576E-05	1429 -1.2946817E-014220 .9348	6606E-055945E-04 .01726E-045947E-05	•	6591E-054374E-04 .3852E-056025E-055423E-04 .8515F-052370E-042497E-04	.6883E-05 .4571E-044114E-05 .6277E-05 .5649E-049314E-05 .2483E-04 .2663E-04
200 5 8 235 5 141 7 245 7	31850F-01-	.0 6215E-01-	2 .bbu0E-01-	-1.294	55945E-04	0.	55423F-04	. 5649E-04-
7.1.17	38010E-0	.0 31829E-02	1 -1542E-02	1429	6606E-0	•	56025E-0	15 .6277E-0
	51218E-0	.0 52438E-0	31001E-0	9335	0. 4.	1.000	4 .3852E-0	44114E-0
HE C MATRIX	12 .1716E-0	.0 13 .6665E-0	14 . 1347E-0	.102374899335	4981E-054484E-04 .0	0.	154374E-0	15 .4571E-0
THE C MATRIX	.4063E-0	.0 .1190E-0	.2258E-0	.1023	4981E-C	•	6591E-C	.6883E-C

THE D HATRIX
.5767 5.130 13.09 -6.040 -1498.
.3924E-03 .1097 .1752 -.3714E-04 .3233E-01
.0 .0 .0 .0 .0
.154E-03 .2897E-01-.4914E-02-.2414E-04-.1011
-.1491E-04 .1303E-02 .3766E-03-.1575E-02 .5228F-01

F100 HODEL-HN=0.9, ALT=65K, PLA= 83- GROUF 2 POINT 9 12/01/75 JHK

THE A HAT	HATRIX						
6944-	.3821E-0	_	430°B	9.769-	5713	9511.	7413E-01
4353E-0	115711	136.1	-232.4	'	5.014	. 1920	1836
1001	.5632	-171.5	-42.25			66 14	.2718
.4203E-01	135		-593.1	3		2702	3530
.6596E-03	153	12 .7112E	-02 1.575	'	10	20-3	3156F-02
.8495	4147E-0		-17.52				1066
.6598	8979E-0		-31.64	536.1			1927
8637E-0	-	174.1	68.74	-751.1			-19.56
7756E-02	·	-	5.411	-65.44	1.114	10-3	7.362
1134E-03	2521	m	-01 .7901E-	-019712	.1652E-01	. 1422E-02	1601.
-1.128	-6.983	1.644	1.888	1007E+C	175.0	13.40	90.04
3978E-01	15892	-539.9	2072.	-294.5	4.771	.3129	10.75
1761E-02	122619E-0	+	92.08	-13.09	.2121	.1394E-01	4779
1854	3128	-244.8	200.3	1572.	3.255	.2077	4.952
1271E-0	11 .2412E-0	11-1-692	-3.504	33.58	5913	19.80	.9602E-01
-2.017	1.779	<b>76.08-</b>	1139.	2472.	-43.81	-3.677	-2.159
.1108	.2751	3709	.9194	1.108	5077E-01	.7765E-01	.3228E-01
1.108	1.427	3530	.5275E	9474. 10-	.5077E-01	.1059	.7532E-01
1452	-1.307	.3969	1884	-1.718	1951	4306	2726
3229	8769	4984.	90 44E-	-01-,8139	9353E-01	2400	1291
2908E-03	39162E-0	24468E	-021467E	-051336E-(	11 .4511E-02	.7128E-04	2140E-02
5769P-0	115227	4717E		-016783	7269E-01	1468	1076
1043	1	7482E	-011357	-1.230	1384	4758	1951
.2407	2.173	.1971	.3175		.3324	.7807	.45 19
86.64-	.1891	39.11	.2261E	0-	.2940E-01	.5294F-01	30491-01
6663	6638	.5795	.3349E-	-03 .2864E-(	12 .4276E-03	.8000E-03	.4543E-03
3.139	28.25	06-44-	4.105		4.280	10.09	5.884
-8.910	804.8-	37.95	46.64-	.7346	.9754F-01	.1694	.1130
3960	3735	1.687	-1.997	-1.967	.4329E-02	.7623E-02	.5021E-02
94.07	-3.800	17.41	-2.910	-2.433	07.61-	.9035E-01	.8393E-01
8	1 10	.6181F	-021507E	3	1657E-01	-20.04	22224E-01
2694	-7.333	-3.339	-1.151	-10.42	28.20	18.78	-51.65

109.0 .1451r+06 6547. 290.8 2404. .1438r+05	.7716E-028987E-05 .8889E-05 .4413E-05 .0 .0 .0 .0 .0 .0 .0 .0 .0 .0 .0 .0 .12032197E-03 .1579E-03 .6828E-04 .14243642E-021960E-038222E-04	.2612 .3974 .3237 .3224E-01 .2826E-04 .3346E-05 .4412E-05 .4463E-05 .0 .0 .0 .0 .4366E-03 .5051E-04 .1189E-03 .6929E-04 .5249E-636070E-041430E-038327E-04	428.3 .307712-01 .0 .4341
F-02 .134E-02 12.37 -1608E-01 E-017217E-03 -2463E-02 -1020 -5.894		939 .2165E-01 .2 033E-05 .3140E-05 .2 000 .0 .0 012E-04 .4837E-04 .4 617E-045814E-045	3546923E-01 393E-01 .3347E-64 .0 91E-01 .4648E-03 723E-035931E-02-
-22-2327 -3249- -59-65 -01-2-679 -71-40 -71-40	MATRI 6E-01- 1E-02 4E-04 4E-04	.2001 .2149 .0 .3333	THE D HATRIX3435 -7.950 1.8 .1966E-03 .5060E-01 .6: .0 .0 .05280E-031454E-0110
	-022327 .2467E-02 .1334E-02- -3249. 35.96 12.37 -59.65 .31451608E-01- -01-2.679 .1441E-017217E-03- -71.40 .9185 .2463E-02- 11.5784581020 - 793.2 -104.9 -5.894	-22.2327 .2487F-02 .1334E-02-109.0 -3249. 35.96 12.37 .1451E+06 -59.65 .31451608E-01-6547. -01-2.679 .1441E-017217E-03-290.8 -71.40 .9185 .2463E-02-2404. 11.5784581020 -222.1 793.2 -104.9 -5.8941438E+05 -017246E-01 6.536 116.0 -17.59 1.714 -017246E-04 .2203E-02 .1047E-0112032197E-03 -04 .2442E-04 .2203E-02 .1047E-01 .14243042E-02-	5E-02-1337

P100 HODEL-HN=2.5, ALT=65K, PLA=350- GROUP 2 FOINT 10 12/04/75 JHK

2.311	-2.476	55(	-4.106	382.9	-503.1	-6.178	-1.740	44.27
-02-1530E-01-146.0 -541.7 163.81176 1.5232912 -21.137854E-01-14092912 -21.137854E-01-14092912 -21.137854E-01-14092912 -12.137854E-01-14091257 7.223 55.67 4.956 -18.358556 7.223 55.67 4.956 -18.36852613525246E-012926 7.1.70 -7.43157.24 24.31 -1.7083-1376E-03-3.515E-02-34146E-01 .4674E-027689E-0333789881E-01-2.904 31525246E-01154411.64	<b>D</b> (V	2.311	-160.1	3.820	86.69	-18.97	3.042	1.331
-02-1530E-01-1.094 -2.912 -21.137854E-011409 -1.257 1.570 2.222 16.92 -18.69 .5056 -18.50 5.721 7.222 16.92 -18.69 .5056 -18.25 71.70 7.431 -57.24 24.31 -1.708 -1.708 -0.19385E-0238358681E-01-2.904 .31525286E-01 -0.31376E-035715E-024146E-01 .4674E-02.7689E-03 -0.31376E-035715E-024146E-01 .4674E-02.7689E-03 -0.31376E-035715E-024146E-01 .4674E-02.7689E-03 -0.31379 -193.0 70.31 1.867 0.0 .4325E-01 .9110E-01 -8.577 31.25 .8296E-01 .9110E-01 .2018E-02 -0.13202 -87.15 91.92 .421.6 1.210 -4469 11.26 .013202 -87.15 91.92 .421.6 1.210 -4469 11.26 .22.97 -692.1 .6097 -21.49 11.26 .110 19.95 .2017 -22.97 -692.1 .6097 -21.49 11.26 .1102 .3074E-01 .2097 .3202E-01 .6230E-01 .3074E-01 .2097 .3202E-01 .6134E-01 .3074E-01 .2097 .7310 1.313 -2.315 -0.14535340E-01 -60497021E-01 -6134E-01 .1255 -12051 -2051	13	.7170E-01	146.0	-541.7	163.8	1176	1.523	.2622
1257 1.570 2.222 16.92 -16.69 .5056 -18.35 .67.423 55.67 4.956 -18.35 .67.23 55.67 4.956 -18.35 .65.67 4.956 -18.35 .65.67 4.956 -17.00 .7.431 -57.24 24.31 -1.708 .9.5.67 4.956 .9.5286.01 .9.528.02 .9.538.0 .9.63.1376.0.3-5715.0-0.2-34148.01 .4674.0-0.2-7689.03 .75.52 48.21 -51.22 -390.7 65.61 -11.64 .9.101.3379 -19.30 70.3.1 1.867 .0 .4325.01 .9.92 421.6 1.210 -44.69 .01 .3232 -1.558.01 .9.92 421.6 1.210 -44.69 .01 .3258.01 .45128.01 .1736 .8296.01 .9410E-04 .20188-02 .01 .3598.01 .45128.01 .1736 .8296.01 .9410E-04 .20188-02 .3057 -22.97 -692.1 2.60721.49 11.26 .1102 .3.452 -1.147 .30748-01 .2997 .3202E-01 .6230E-01 .102 .3.452 -1.147 .30748-01 .2997 .3202E-01 .6230E-01 .102 .3.452 -1.147 .30748-01 .2997 .3202E-01 .1255 -01 .3232 .5340E-01 .67318-01 .1034 .7044 .5.315 .1105 .1102 .3345 .1248 .7044 .5.315 .1105 .10348-01 .1103 .31958-04 .201 .2232 .2152E-01 .3275E-01 .2051 .1105 .1103 .2051 .20	62	15	-1-094	-2.912	-21.13	7854E-	011409	76 33 E-01
- 4255 5.721 7.223 55.67 4.956 -18.35 -17.0 -7.431 -57.24 24.31 -1.708 -1.708 -1.5724 31 -57.24 31 -1.708 -1.5529 71.70 -7.431 -57.24 31 -5.708 -1.5826-02-3835 -8681E-01-2.4146E-01 .4674E-027689E-03 -1.376E-035775E-024146E-01 .4674E-027689E-03 -7.552 48.21 -51.22 -390.7 65.61 -11.64 -11.64 -11.61 -11.64 -11.61 -11.64 -11.61 -11.61 -11.64 -11.61 -11.61 -11.64 -11.61 -11.	4	12	1.570	2.225	16.92	-18.69	9505.	.3609E-01
.6529 71.70 -7.431 -57.24 24.31 -1.708	7	.42	5.721	7.223	25.67	4.956	-18.35	.1173
-01-9385E-02-38358681E-01-2.904 .31525286E-01 -03-1376E-035715E-021158E-024148E-01.4674E-027689E-03 -7.552	3	.6529	71.70	-7.431	-57.24	24.31	-1.708	-20.12
1376E-035715E-021158E-024146E-01 .4674E-027689E-03 -7.552	9	93	3835	8681E-0	1-2.904	.3152	5286E-01	5.565
The contract of the contract	39E-03	13	5715E-0	21158E-0	24 148E-C	11 .4674E-	027689E-03	.8245E-01
E-01-3379 -193.0 703.1 1.867 .0 .4325E-01 E-03-1501E-01-8.577 31.25 .8296E-01 .9410E-04 .2018E-02 E-01-3202 -87.15 91.92 421.6 1.2104469 4469 44612E-01 .1736 .82961110 19.95 4469 3.057 -22.97 -692.1 2.60721.49 11.26 11.26 3.057 -22.97 -692.1 2.60721.49 11.26 2971 -22.97 -692.1 .2997 29352971 -21.49 11.26 3.041 2.134 3074E-01 .2997 3202E-01 .102 11.26 3.041 2.134 7044 6.317 7310 1.313 E-01-45535340E-01-6731E-01-6049 7021E-01 .102 11.31	13	.5	48.21	-51.22	-390.7	65.61	-11.64	39.06
E-031501E-01-8.577 31.25 .8296E-01 .9410E-04 .2018E-02 E-013202 -87.15 91.92 421.6 1.2104469 E-03 .1359E-014512E-01 .1736 .82961110 19.95 3.057 -22.97 -692.1 260721.49 11.26 3.057 -22.97 -692.1 260721.49 11.26 3.452 -1.147 .3074E-01 .2997 .3202E-01 .6230E-01 E-013836 1.658 .6148E-01 .5072 .6136E-01 .1102 3.441 2.134 .7044 6.317 .7318E-01 .1255 E-0145535340E-016731E-016049 .7021E-011255 E-012232 .2152E-01 .3275E-01 .2859 .3415E-01 .134E-01 E-0174637157E-01 .11779867 .116320518668 .59251708E-049221E-031067E-033195E-046668 .59251708E-049221E-031067E-033195E-046668 .59251708E-049221E-031067E-033195E-034293 1.748 -1.999 .5763E-01 .1334E-01 .1198E-014293 1.748 -1.999 -1.997 .6403E-02-19.994.757 17.74 -2.645 -3.177 -20.00 -2.473 -4.757 17.74 -2.645 -3.177 -20.00 -2.473 -3.179 -2.645 -3.177 -20.00 -2.473 -3.179 -3.17	29E-01	3	-193.0	703.1	1.867	0.	.4325E-01	8.502
E-013202 -87.15 91.92 421.6 1.2104469 E-03 .13598-014512E-01 .1736 .82961110 19.95 3.057 -22.97 -692.1 260721.49 11.26 3.057 -22.97 -692.1 260721.49 11.26 3.452 -1.147 .3074E-01 .2997 .3202E-01 .6230E-01 E-013836 1.658 .6148E-01 .5072 .6136E-01 .1102 3.441 2.134 .7044 6.317 .7312E-01 .1255 E-0145535340E-016731E-016049 .7021E-011255 E-012232 .2152E-01 .3279E-01 .2859 .3415E-01 .134E-01 E-01 .2232 .2152E-01 .3279E-01 .2859 .3415E-01 .134E-01 -01 .7254 .6995E-01 .1081 .9636 .1126 .1994 E-01 .7254 .6995E-01 .1081 .9636 .1126 .1994 -1.4493 .7197E-0111179867 .11632051 -4293 1.748 -1.999 .1.997 .6403E-01 .1198E-01 .4293 1.748 -1.999 .1.997 .6403E-02-19.99 -4.757 17.74 -2.645 -3.177 -20.00 -2.473 -4.757 17.74 -2.645 -3.177 -20.00 -2.473 -3.195E-02 .1744E-01 .1110E-01 .3074E-02 .2305E-01 .3202E-02-19.99	44E-03	.150	-8.577	31.25	-8296E-	5	04 .2018E-02	.3778
E-03 .1359E-014512E-01 .1736 .82961110 19.95 3.057 -22.97 -692.1 260721.49 11.26 3.057 -22.97 -692.1 260721.49 11.26 1.658 .6.148E-01 .2997 .3202E-01 .6230E-01 3.452 -1.147 .3074E-01 .2997 .3202E-01 .6230E-01 3.452 -1.147 .3074E-01 .2997 .3202E-01 .6230E-01 3.452 -1.147 .3074E-01 .2997 .3202E-01 .1302 3.541 2.134 .7044 6.317 .7310 1.313 E-0145535340E-016731E-016049 .7021E-011255 E-012232 .2152E-01 .3279E-01 .2859 .3415E-01 .1255 E-012232 .2152E-01 .3279E-01 .2859 .3415E-01 .1994 E-01724637197E-01 .11179867 .116320518718E-02 .59.591281E-026916E-018004E-022395E-026668 .59251708E-049221E-031067E-033195E-045.100 -48.797621 -6.7437937 -1.4214293 1.748 -1.999 .5763E-01 .1334E-01 .1198E-014293 1.748 -1.999 .5763E-01 .3202E-02-19.99 -4.757 17.74 -2.645 -3.177 -20.002473 -4.757 17.74 -2.645 -3.177 -20.00 -2473 -11.91	53E-01	320	-87.15	91.92	421.6	1.210	6944	3.761
E-01-1.343 -1.085 2.231 .404129352971 -3.452 -1.147 .3074E-01.2997 .3202E-01.6230E-01 3.452 -1.147 .3074E-01.2997 .3202E-01.6230E-01 3.452 -1.147 .3074E-01.2997 .3202E-01.6230E-01 3.941 2.134 .7044 6.317 .7310 1.313	11E-03	.135	4512B-0	1.1736	.8296	1110	19.95	.6824E-01
E-01-1.343 -1.085 2.231 .40412935297129352971 32.452 -1.147 .3074E-01 .2997 .3202E-01 .6230E-01 .3836 1.658 .6148E-01 .5072 .6136E-01 .1102	37	0.5	-22.97	-692.1	2607.	-21.49	11.26	2.759
3.452 -1.147 .3074E-01 .2997 .3202E-01 .6230E-01 .3836 1.658 .6148E-01 .5072 .6136E-01 .1102 3.941 2.134 .7044 6.317 .7310 1.3135340E-016731E-0160497621E-0112555340E-016731E-0160497621E-01125520517554 .6995E-01 .1081 .963612612619945978E-01 .72546995E-01 .1081 .9636126126199420512	06E-01	-1.3	-1.085	2.231	1484.	2935	2971	4330
E-01.3836 1.658 .6148E-01.5072 .6136E-01.1102 3.941 2.134 .7044 6.317 .7310 1.313 E-0145535340E-016731E-0160497021E-011255 2.01.2232 .2152E-01.3279E-01.2859 .3415E-01.6134E-01 E-01.7254 .6995E-01.1081 .9636 .1726 .199420518718E-02.35.991281E-026916E-018004E-02295120518718E-02.35.991281E-026916E-018004E-02295120516668 .59251708E-049221E-031067E-033195E-0495.100 -48.797621 -6.7437937 -1.421 -6.7429 1.748 -1.999 -1.997 .6403E-01.1198E-014293 1.748 -1.999 -1.997 .6403E-03.5751E-03 -4.757 17.74 -2.645 -3.177 -20.002473 -4.757 17.74 -2.645 -3.177 -20.002473 -4.757 17.74 -2.645 -3.177 -1.991 11.91 11.51 14.53 -	51	3.452	-1.147	.3074E-0	1 .2997		_	.1493F-01
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-EXTRA POINT 1-1/23/76 PAA P100 HODEL-0.9/10K, PLA=36,

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	-701.3	-73.86	251.4	286.7	-7.432	29.05	65.75	-71.46	-1.758	-022620D-	6.064-	1.345	-5793D-	208.5	1.572	44.27	9.950	2.880	-7.325	7.169	-01 .1381	-016449	-1.453		9	40		.93	-1.9	-2.5	-03626	-4.993
	473.9	-222.0	-30.54	-583.2	2.434	-2.856	-6.425	106.9	.1983	-02 .2644D	47.94	311.8	13.86	28.08	1388	168.0	6.059	.3165	8105	.8035	-01	-110	-01100	o						-2.7	0	5565
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4878E-04 .	1202E-02-	.1202E-021904E-02		-2092			
0E-044	.4610E-044100E-03	. 1025E-62-	. 1025E-62 4246E-02.	1992			
9815E-04 .	-1052E-01	.6298E-03	.4425E-03	.4183			
2377B-04 .	-2330E-02	.5000R-03		.1026			

F100 RODEL-0.9/10K, LLA=52,

## -EXTER FOINT 2-1/23/76 FAA

THE A MATRIX	RIX	-		200	1		6330
-4.343	000/	•	433.1	1.701-	9	-2.314	0
-1.361	-4.398	112.5	-235.2	-124.4	18.43	-1.792	-1.943
4.045	.252	-150.8	6.710	373.4	-52.26	5.569	3.634
1649	9	1 119.1	-536.5	83.06	-4.766	.5219	-1.219
.1915D-0	20D-0	2 .1394	1.901	-7.930	.1304	5800D-02	.9832D-03
.5470		.8103	.8525	34.51	-18.54	.5260	.4087D-01
.4935		1.621	1.705	80.89	3.030	-18.96	.8396D-01
8413	.4875	30.93	-1.958	-78.77	29.22	-1.207	-20.10
3102b-01	1	1 .1542	-3947D-0	1-2.596	.4171	3777B-01	6.095
4591D-03-		3 -2331D-02	.52630-0	33807D-0	1 .6199b-(	325493D-03	10-01506.
-6.603	-7.979	29.79	-15.83	-619-0	85.40	-9.500	38.57
106D-0	1-,3199	-73.35	268.8	-1.990	.3385	5493D-01	9.179
9447D-03	1	1-3.260	11.94	8998b-	1 .15040-	112472b-02	64079
1152	2021	-33,35	19.77	215.7	.7713	1071	4.197
0-0	2 .2003b-01	17815D-01	.6315D-0	1 2.146	2051	19.93	.8842D-01
-2.444		.2262	151.0	175.2	-13.41	2.932	1709
8099*	1058	-3.012	6.879	5.739	7378	5332	1942
8.792	7.	-3.181	5998D-0	16251	86800-	01-,3350	9269D-01
.2234	2.032	4.133	.2967		.3639	1.217	.4237
-2.135	8.	4.271	-4 10 4D-0	_	.1002	.1504	.61791-01
.1187D-0	1.51	I2248D-01	.6552D-0	-	-06#15. Ti	11 .5376D-01	9
-2916D-0	1 .26	.2473D-01	-3851D-0	_	-4207b-	_	.53850-01
-5455D-0	1 . 49	.5027D-01	.7324D-01	_	.8547b-0	-	.1024
6208D-0	159	5838D-01	8334D-0	Ī	1008	3021	-,1165
-50.00	1058D-01	1 39.69	.3157b-0	2-	1 .0	1025D-01	-44 14D-02
9999-	6668	.5880	.4209D-0	4-	3 .0	1367D-03	+0-07288.
5044	-4.635	-48.43	0429-	-6.180	7962	-2.403	0096
-10.17	-10.19	39.13	-50.01	.2841D-0	-97799- I	126836D-02	•
4518	4529	1.739	-2.000	-1.999	-2671b-0	034 10 1D-03	9
-4.654	-4-707	17.90	-2.938	-2.938	-19.77	2	11765D-02
8	.16	1.15810-01	-2525D-0		1 .2671b-(	05-19.99	9
.7584	2.127	-2.106	.2305	2.159	26.47		99.64-

780.8 2266. 3299E+05 2345. 55.76 323.8 885.1 4842. 71.66 5351E+05 5351E+05 257.6	### C MATRIX  -8332E-01-6537	.3492 .1693E-01-1.5054950E-014364 2.424 2.1196991E-015159E-034709E-024700E-036866E-036322E-028054E-032457E-029710E-0300 .0 .0 .0 .0 .0 .0 .0 .0 .0 .0 .0 .0	%
-780.8 2266. -3299E+05 2345. -01 55.76 323.8 654.7 -885.1 -4842. -02-71.66 .5351E+05 -5787. -01-257.6 -2565. 28.00	98.04 016375 031216 031218 032123	014364 036322 041203 04176 042086 045306	10.86 -459.6 .2994E-01-5.929 .0 .5738E-031204 .2631E-03 .1102 .3460E-032182
9.923 -780.8 -22.65 2266. 79.68 -3299. 7.712 23459193E-01 55.761.636 323.8 -3.431 654.7 -18.92 -885.1 -57.95 -484202-8567E-02-71.6613.1.8 .5351 -6635 -57871.247 -25653309 28.00	x.6537 5.294 94.55 98.04 .2168E-021463E-011613E-016375 .0 .0 .0 .0 .4131E-042789E-033076E-031216 .3745E-034515E-02 .2991E-03 .1181 .1161E-041211E-035412E-032123 .6475E-057094F-041386E-035344	4950E-014364 036866E-036322 .0 051306E-041203 051276E-041176 042320E-042086	
-81.44 -23.43 69.52 6.762 .4995 3.980 11.38 -14.41 5323 -02.78788- -113.9 -3751 3751	5.294 -021463E- 0.042789E- 034515E- 041211E- 057094E-	.4709E-01-1.505 .4709E-024700E-0 .8996E-048963E-8758E-04 .8720E-1567E-4030E-044014E-	
158.0 341.9 341.9 35.99 39.99 72.99 72.99 -69.91 -4902 -1.774 -1.774 -1.774 -1.774 -2.506	8332E-016537 8332E-01.2168E. .0 .0 .0 .4390E-04.4131E. .1259E-03.3745E. .7404E-041161E.	.1693E-03-4709E-05-8996E-05-8996E-04-1572E-05-4030E	1018
THE B MATRIX  1940 -1775 -4927 -2826E-01 3-2826E-01 3-2826E-01 -1204 7 -1367 -6185E-01 -1204 7 -1204 7 -1204 7 -1307 -6185E-01 -6185E-01 -1307 -6185E-01 -1307 -6185E-01 -6185E-01 -6185E-01 -6185E-01 -6185E-02 -6185E-02 -6185E-02 -6185E-02	THE C MATRIX8332E-013495E-01 .0 .4390E-04 .1259E-03 .7404E-04-	5159E 5159E 9829E 1718E	THE D HATRIX 1018 -2 1251E-02- 0 2191E-04- -1848E-04 -3897E-04- -9678E-05-

-EXTER POINT 3-1/23/76 FAA P100 HODEL-0.9/10K, PLE=67,

	+00+	.3429	6447	.8327	011125D-01	.7138b-01	.1260	-20.15	01 6.230	03 .9230D-01	37.06	9.465	02 .4207		.9797b-01	45	.4082	2.853	-4.013	2.752	013701b-02	.8164D-01	.1447	1697	012021D-01	032964D-03	-1.562	.1374	9	.8972D-01	.4041D-02	-48.95
	-1.303	5.596	-1.014	5.040	2435D-	.5652	-19.02	-1.126	6588D-	-059662D-	-10.48	.1285	-01.5797b-	-4875D-	19.92	3.867	.4260	4.365	-5.793	4.528	-01 .3630D-	01 .1887	.3316	3814	-054260D-	-046117D-	-3.562	.1835	-02 .8257D-	.1127	-02-19-99	25.20
0	007-1-	27.99	-70.02	-1.503	.1435	-18.50	2.711	28.31	.4924	-01 .7309b-	89.27	1.152	-5120D-	1.187	2534	-15.90	3222	2.154	-3.014	2.151	-01 .488BD-	-6629D-	.1123	1338	4603D-	-026138D-	-1.237	.1089	-4849D-	-19.70	-01 .3683D-	28.20
. 303	-000-3	-77.96	340.5	196.8	-8.749	41.13	70.60	-80.69	-4.727	-026981D-	-751.9	-2.473	1076	187.3	-01 2.502	216.5	10.55	18.34	-25.79	17.66	-02-,2387b-	-01 .5247	.9299	-1.091	-011299	-031905D-	-10.04	.8832	.0	-2.426	-02 .2598D-	6.741
	****	-187.8	-65.31	-513.4	-01 1.732	1.184	2.162	-2.522	2252	-023003D-	-22.68	237.8	10.57	23.01	772	150.2	8.217	0.	-2.872	1.971	-012524D-	-01 .5772b-	-01 . 1010	-011212	1876D-	2694D-	-1.113	-49.91	-1.996	-2.956	-01 .2886D-	.7490
	•	131.8	-172.0	135.8	-01 . 7910D-	01 .7346	1.331	22.67	-01 .1685	-03 .2471D-	20.18	-63.24	01-2-810	-28.68	-016740D-	3.008	-3.164	-2.181	2.720	6.154	-023227D-			80 99D-	-01 39.59	.5865	-48.65	38.97	1.732	17.88	63	-1.832
MATRIX	• 1003	-4.814	5.218	.2752	-011240D-	9815D-	1914	.7243	1-1	-032053D-	-7.581	013594	1403D-021597D-	2103	-02 .2200D-	1.579	2.895	23.58	-19.33	10.47	d	-	-	-01-,8159	4857D-	6673	-7.460	-9.237	4103	-4.103	_	5.702
-	27.7	8857	2.854	1.024	-1073D-	.5567	.2328	5410	2696D-0	3978b-0320	-5.033	3167b-	1403D-	1397	4188D-	-1.515	1.056	11.38	-2.156	9562	-2900D-0	.4317D-0	.7641b-0	8764D-	-50.01	6668	8116	-9.867	4385	-4.506	A	1.136

	######################################	.4560E-034104E-024296E-036295E-035666E-026886E-031995E-028815E-03  .0  .0  .1.000  .0  .0  .0  .0  .0  .1.313E-04  .195E-031324E-041935E-041755E-032150E-04629E-042730E-04  .1313E-04  .2313E-04  .33145E-04  .33145E-04	
11.60 3141. 1.261 .2757E+05 23.477865E+05 34.31 .2635E+05 .1871 -26.46 .9108 1120. 9.695 -167. .6237 -5585. .9079E-02-82.68 1499E-01-232.2 .1839 -1916. .2994 29.59	### C #ATRIX  -3660E-016903	2.041 3.2 035666E-0268 041755E-0321 042471E-0330 042471E-0330	7.734 -5.996 27206618 .2182E-02 4.874 .0 .0 .0 .0 .1796E-02 .1522E-031995 .1534E-027516E-03 .2518 .3552E-035987E-033440 .4448E-032562E-048093E-01
	*6903 6.804 107.8 65.76 •1043E-027730E-021279E-014302 •0 .0 .0 .0 .0 •3280E-042416E-034011E-031333 •3260E-033455E-02 .3676E-03 .1215 •1992E-041047E-035618E-031874 •5441E-056401E-041425E-034822	-036295E- -041935E- -04 .1778E- -042749E-	7.734 -5.996 2720. .6618 .2182E-02 4.874. .0 .0 .0 .0 .0 .0 .0 .0 .0 .0 .0 .0 .0 .
-66.24 23.39 23.39 29.41 .4162 3.075 8.116 -10.14 5600 8215E-02 -94.50 .1038E-01 2621 -47.04	6.804 8-027730E 8-042416E 8-033455E 8-041047E	-1.616 -0242968 1.000 -0313248 -0318838	
THE B HATRIX -3.658 -347.3 -2.623 -262.3 3.839 991.5 -2.521 319.8 -3.429E-02-80.73 -8475E-01 164.5 -1306 164.7 -1395 -220.4 -1395 -220.4 -1217E-01-15.89 -2821E-01-8.419 -2821E-01-8.419 -2928E-02-48.16 -3596E-02 8.451	THE C MATRIX -3660E-01-6903 -2352E-01 .1043E-0 -0 .0 .0 .0 -1624E-03 .3280E-8127E-04 .3260E-1992E-7772E-04 .5441E-1992E-	0E-03-41041 0E-04-12941 3B-04-12941 0E-04-18591 2E-05-47481	THE D HATRIX2849
7HE B 2.558 2.658 3.839 2.521 3.84759 3.643 3.648 3.648 3.648 3.648 3.648 3.28218 3.28218 3.28218 3.28218 3.28218 3.28288 3.2828 3.2828 3.2828 3.2828 3.2828 3.2828 3.2828 3.2828 3.2828	THE C - 366 - 235 0 0 0 0 162		1HE D - 2849 - 84961 - 25291 - 20931 - 36001 - 50881

7100 HODEL-0.9/10K, PLA=83,

-EXTRA POINT 4-1/23/76 FAA

.5054D-02 .4778D-01 .7281D-03 .6621D-02 -.2662D-01 .9643D-01 .9597D-01 .1632D-01 .3473 9.684 6.425 -19.50 6.508 .4304 8409 --.5338 .1485 -.6756 -.2475 42.75 4.487 -5.184 -.2683 -4.262 -.5613 2.092 2.051 -2.294 . 16 19 D-02 .1360D-01 .1648D-02 .1158D-01 3.915 12.31 10.35 1.027 -2.756 3.624 .3047 -.5184 -20.96 9908 2440 19.80 6969 .1115 .2623 416.6--4.146 .1093 3.491 3.256 -8.290 -3.973 -. 4220 -.9827 .8621 -20.04 .5843D-03 .5206D-02 -.2072b-01 .38510-01 .1223b .7594D 27.73 118.3 2,389 1.837 1.572 4.980 .1169 -19.67 4.019 .4191 .2568 -19.66 .4596 .8259 .1062 -1.703 -.2082 -. 4664 42.12 -100.5 -13.29 -.4815 -29.65 -2.682 -4786D-02 -.6063b -689.6 .3141 13.48 154.7 28.00 -4.114 -685.0 23.40 .1086 -3.975 3.508 42.26 .9870 -1.956 -2.176 6.009 58.68 -8.452 54.46 -57.34 -.5673 2.880 -34.10 -15.07 -1.763 .4985D-03 .3365D-01 .1173b .1117L 4.701 6.909-19.55 1.488 -3.794 .3933 68.64 -1.995 -204.9 -73.63 2.042 -3.816 -7.046 7.596 .7447 91.15 194.1 8.629 -. 3816 11.13 -1.673 -. 1954 -- 4422 .4909D-02 1.718 42.34 -.30121 17.81 3.357 -44.85 38.65 .5477 -1.919 -50.78 -22.98 2.594 -6.873 -6.873 -3.199 -. 1224 -.2784 39.56 .5860 -. 1447D-01 124.9 .1013b-01-.2058b-01 .1519 -- 9490D-01-.7904 .3314 -.5431D-03-.2200D-01-2.257 4.973 .2471 -3.123-.1906b-02-.3450b-01 --2858D-04-.5095D-03 .9070b -.1701 -.6098D-01 1.119 -9.781 -- 12 15 D-0 1-. 4949 -.2840 -25.67 -8.304 9.082 11.43 -1.327 .2364 -.6631 31.79 -.3978 -3.803 1-.1317 21.92 -14.42 -2.992 2.641 -8.950 THE A MATRIX -.1172D-01 .1866D -- 1463D -. 7310 -. 1480 4717 -.2743 4.059 .6760 .2920 .8750 -2.872 .2979 76.64-3.556 -9.578 --4257 -4.358 -.2873 12.90 -. 1486 -.6662 2.081 -4.291

	2.047 .4016 .5729E-03 .2703E-03 .0 .0 .1541E-03 .7437E-04 .1582E-037635F-04 .3086E-03 .7992F-04 .1014E-03 .2091E-04 4.461 1.087	.0 .1296E-03 .8054E-04 .1332E-038274E-04 .1396E-03 .8701E-04 .3619E-04 .2280E-04
<b>0.0.4.0</b>		.0.42E-04 .6242E-04 .6718E-04 .1761E-04
18.86 .1516E+05 33.36 .2137E+05 38501144E+06 13.911352E+05 5183E-01 122.8 2.396 -1313. 4.584 -3635. 13.53 5774. 1986 -5993. 2997E-02-88.77 7.474 .1390E+06 .1769E-01-245.6 .4711 -1598.	124.0 7.743 15.39 .4392E-023282E-014038E-03 .0 .0 .0 .1748E-028612E-021091E-03 .1771E-02 .8816E-022091E-02 .1229E-029218E-02 .4687E-03 .3200E-032426E-02 .3979E-04 .7964 7.125 4.776 .2134E-03 .1921E-02 .2274E-03	.0 .5890E-04 .5294E-03 .6359E-04 .5719E-03 .1666E-04 .1496E-03 .1666E-04 .1496E-03 .7477E-03 .5056 -3400E-02-8853 .5701E-03 .7169
		E-04 .5890E-04 E-04 .6051E-04 E-04 .6359E-04 E-04 .1666E-04 -4005E-02 -0 E-02 .7477E-03 E-02 .7477E-03 E-03 .5701E-03 E-03 .2362E-03
-157.8 24.85 -73.27 -73.27 -77.79 19.04 30.36 9.285 1.935 1.935 1.935 1.935 1.935 1.935 1.935 1.935	.1.203 9.369 .1088E-03 9.281E-03 .0 .0 .0 .2834E-04 .2459E-03 .2145E-032698E-02 .2290E-04 .3692E-03 .1137E-04 .6526E-04 5.966 -1.757	.0 .3985E-03 .3700E-04 .4094E-033798E-04 .4305E-03 .3998E-04 .1128E-03 .1048E-04 .177.1 30.66 1.007 .5808 .0 .0 .2895E-014069E-02 .2841E-011505E-02 .4872E-02 .7105E-03
THE B HATRIX -1.580 -10991.431 -104.4 4.044 581.7 1.773 12731238E-01-109.2 -2124 127.4 -4481 262.2 -4504 -114.4 -37128-02-3.137 -4748E-04-4851E-0 1.153 -809.1 -2824E-01-2.609 -1265E-02-9387E-0 -1770E-01-12.90 -2031E-01 5.426	E-01	**************************************
-1.580 -1.431 4.044 1238 1238 4504 4504 1753 1265 1770	. 1313] . 1313] . 0 . 5774] . 9426] . 9774]	-45831 -45831 -48171 -12651 -35011 -64151 -18241

P100 BODEL-0.9/30K, FLA=36,

-1.465

-- 4677 1.364 .2303 -4302

.2562 -.6762

## .50510-01-.1488D-01-.5550D-02 .2217D-01 .9209D-01 .7004D-01 -20.07 -,3502 -.5719 -- 4103D-01 6.217 1.503 39.39 -.8644 .4674D-02-.6018D-03 -EXTRA FOINT 5-1/23/76 FAA -9.930 .4547 -1.274 --7550 2,388 .1970 -18.74 .3180 -18.95 95.83 3.215 10.79 -25.20 -2.413 29.14 05-.5748b-0 171.5 -171-8 0.619--8.075 -3.880 -1333. -113.8 338.9 75.88 60.53 -2672D 1.479 2.084 5.330 -5.090 -236.6 5.210 1. 164--2004 -40.78 .4515D .5828D 2.657 110.9 1.031 3.136 -139.8 .3126 .3096p-01 109.0 68.54 61.34 -.8336D-01 -.1764D-01-.1155D-02 .6264D-02-.4771D-02 -.2609D-03-.1848D-04 .3013 .6624 -. 2555 -8.071 2.260 -2.177 THE A MATRIX

4264D-01 -3198D-01	.05330D-02 .1426D-013013D-02 .8662D-01 .2878D-01	.8954D-01 8954D-01	7008 2931D-01	.1279D-02 .1599D-01 3198D-02
35577355D-014264D-01 1199D-011634D-01 .3198D-01 7993D-01 .2697	1998b-01 .05330b-02 1702b-01 .1426b-013013b-02 3357b-01 .8662b-01 .2878b-01	9112b-01 .2321 9192b-012354 5995b-024086p-02	7993D-045448D-04 .0 7114 -1.8597008 1599D-01 .2860D-01 .2931D-01	.7194b-03 .1144b-02 .1279b-02 19.736538b-01 .1599b-01 .2398b-02-20.013198b-02 .5.65 26.35 -49.67
3557 73551 1199D-0116341		.9112b-01 .2321 9192b-012354 .5995b-024086	.7993D-04 7114 .1599D-01	.7194D-03 .1144F -19.736538F 12398D-02-20.01 25.65 26.35
3.208 2.962 .2648D-01 .2383	.8821135D-016808D-01 1296D-012092D-021895D-01 1060D-01 .2043D-01 .1838	.1 .5447 115447 .0	-4.425	-1.992 -2.730 )25447b-0 2.111
3.208 2.962 .2648D-01 .2383	.8821135D-01. 1296D-012092D-02.	.2650D-01 .6506D-01 .5447 .2747D-016052D-015447 39.62 .0 .0	-49.98	-1.999 -1.992 -2.865 -2.730 026052D-025447D- .2402 2.111
-1.318	1.882 111296D-0	.2650D-C 2747D-C 39.62	.5870 -48.12 39.23	1.744 17.87 1.9637D-0
4.306		01 -4180 014029	6667 -3.273 -10.24	4552 -4.628 024029D-0
.3245 .15 4.149 4.3 27980-01	9736 -1. 1552D-021	.4868D-01 . 4924D-01 -49.99	6666 3931 -10.33	-4591 - 45 -4.702 -4.6 1679D-0240

.7802D-01

4.298

11490-01

-.5055D-02

-. 1141

4.311

.1897

26.37

72.74

593.3

-162.1

.5774D-03-. 1252D-01-7.206

.1315D-01-.2817

-5.241

-. 1660

-. 1215

-.2405

.1738D-01-.1667

-.7570D-02

-1.941

2.448

428.5

.2935

-.2234 -13.90

16.61

-.1529

9.422

.3693D-01 .1641D-02

40.5	+0.5	### C MATRIX  -3783E-012915	2658E-032235E-021446E-033357E-033021E-01 1.106 1.096 .2132E-022658E-032235E-021446E-033357E-033021E-024846E-031256E-024730E-030 .0 .0 .0 .09583E-058028E-045158E-051218E-041085E-031752E-044523E-041707E-048873E-05 .7414E-04 .4788E-05 .1114E-049973E-04 .1613E-04 .4198E-04 .1581E-041762E-041488E-039723E-052256E-042012E-033198E-048411E-043153E-044412E-053682E-042388E-055550E-054978E-047972E-052059E-047807E-05	-01
597.2 1517. 1378E+05 540.6 01-18.01 112.2 435.6 -68.32	.44358-02-62.36 1014 .5082E+05 .2145 .5124 .9268E-02-227.8 .2159 .2245 .1143 .13.56	111.9 019029 033251E 035867E 031477E	02 .2451E 033021E 041085E 04 .9973E 042012E	3.249 2.347 3294E-01-11.57 0.5091E-031249 -1265E-02 .6224E-01 -3297E-031759
1.682 -2.714 25.69 2.278 5985E-0 -7300 -2.467		90.34 0.0 139921E- 11.9182E- 131818E-	3329E-02-03-3357E-03-05-1714E-04-05-2256E-04-05-2256E-04-05-2550E-05-05-05-05-05-05-05-05-05-05-05-05-05-	.6946 -3.249 2.347 .2255 .3294E-01-11.57 .0 .0 .0 .1907E-02 .5091E-031249 .1247E-021265E-02 .6224E-01 .4957E-033297E-031759
-22.92 -5.061 14.35 1.281 1.281 2.107 8.052 -6.932	2-2619B-02-56.04 -56.04 -2060 -9037E-02 -1938 -32.63	.2915 5.351 90.34 111.9 .1221E-021559E-012755E-019029 .0 0 0 .4404E-045615E-039921E-033251E-01- .4372E-031146E-01.9182E-03.2982E-01- .1688E-041969E-031818E-025887E-01- .4964E-051246E-034564E-031477E-01	6623 1446E-0 1.000 45158E-0 45158E-0 39723E-0	
THE B MATRIX -1193 -67.73 -5533E-02-39.59 -1244 172.0 -4767E-01 14.83 -7653E-02-24.68 -8184E-01 40.45 -2431 122.3 -2006 -113.0	101	THE C MATRIX3783E-012915 .1476E-01 .1221E-0 .0 .4667E-04 .4404E-0 .1172E-03 .4372E-0 .1142E-031688E-0	-2658E-032235E-0 0 9583E-058028E-0 8873E-05 .7414E-0 1762E-041488E-0	THE D MATRIX -1257E-01-112.0 .6946 -3.249 2.347 -9382E-03-1.138 .2255 .3294E-01-11.57 -0 .0 .0 .0 .0 .0 .0 .0 .0 .0 .0 .0 .0 .0

F100 HODEL-0.9/30K, PLA=52,

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	•		1.4022527		0-02	20530	9.293080D-01	7	9-010	.7578b-03 .9619b-01	9	5	05	397D-01 4	. 16.	·			٠		.34230-01 .77870-02	ī	1		1	5025D-049081D-04			1146D-01 .7174D-02		-20.009081D-03	-49.
	·	15.48	-35.02 -	5662	.8626D-01	-18.48	2.977	27.76	- 4567	-6794D-02-	94.32	1.121	10-01664.	1.155	-	2	6139D-01	1.269	-1.941	1.184	D-01 .3170D-01	1228D-01	0	-2660D-01-	D-01 .34110-02-	1-03 .4548D-04-	- 2746	.1228	.5457D-02	-19.67	D-0 16822D-03-	1
	.2 -693.	=	82	-	11			35 -145.2		0-02	•	'	1		6-4 10-0	.2 345.3			,		5418D-02 .4869I	-	į.		.02-	-04-	49 2.881		'	,	.02	
	42	-18	-80	•		•		43.73 1.735	•	0-02-					7055. 6151.	.9039 275	4	.040	1.065 -1.835	-	1159D-01.	-02-	-01-	10-0	•	.585063				6	.1346D-01 .1274D-	2.168 .27
BATRIX	.3666	0	. 0/	1	30-05	9	#3	91	9363D-02	90D-03	**	1-,3583	31593D-01-	•	2 .20850-01-	1.528	2	- 90	.32	5.966	2 .4045D-01-	1112	1668 -	1 .2010	070-01	7	2.117 -4	83	80	-4.063	034276D-02 .	2.491 -2
EA	-1.584	3888	1.159	.4927	4	.5185	.3419	4304	2215D-0	3281D-(	-4-401	1969D-0	8640D-0	1463	3828D-0	-1.402	.6510	996.5	-1.371	4205	-8003D-0	1235D-01	1806D-01	-2423D-0	-50.00	6667	-2423	₩06.6-	-	-4.535	503D-	9908*

	4.91610155086F-01 4742E-021184F-02 .5430F-04 .0 .0 .0 5120E-031280E-03 .6180E-05 2926E-02 .1447E-036835E-05 .1486E-03 .1072E-03 .9930F-05 1611E-04 .5993E-04 .2593F-05	1.754 1.690 .2205 .3837F-046595E-04 .6810E-04 .0 .0 .0 .4322E-056900E-05 .7294E-05 .4696E-05 .8096E-058014E-05 .6936E-051069E-04 .1198E-04	
6.368 2156. 2.686 .1333E+05 -1.0753859E+05 14.88 .1203E+05 .2273E-01 55.96 -1.116 98.00 -2.186 187.1 -2.779 210.6 -3.394 -5313. -02.5006E-02-78.75 -57.57 -4805. -7771 -4805. .7771 -4805. .7485 -1833. .2485 -1833. .1615 21.61	110.8 .2495r-022139 .0 .2854r-032310r-01 3133r-03 .2607k-01 .4561r-039470r-02	.1537 1.397 .4380E-04 .3942E-03 .0 .0 .5063E-05 .4557E-04 .5545E-054977E-04 .8019E-05 .7217E-04	6.207 -1.223 1555. .4493 .3166E-02 2.874 .0 .0 .0 .5341E-02 .3917E-031378 .1728E-022301E-02 .1455E-01 .5091E-033427E-041849E-01 .7172E-03 .9366E-042850E-02
-38.56 5.007 5.348 20.35 .3560 2.155 7.220 -9.593 5587 8584 -97.40 .3980E 1.216 3733	## C HATRIX	.2940E-03 .2294E-04 .0	
THE B MATRIX 8603606.0  -3.184285.6  2.024 777.5  -2.879 227.8 1390E-01-34.52  .1084 263.0  .3412E-01 245.1 9785E-01-260.9  .108 -12.00  .16.32E-021766  16.71 -2662 3026E-01-21.03 1395E-029519 4206E-01-30.62 2768E-02 7.633	THE C BATKIX5228E-023325 .1116E-01 .3064 .0 .1020E-03 .3295 .7719E-04 .3475 .6803E-041838	.3529 .3564E-04 .0 .3846E-05 -4246E-05 -6172E-05	THE D MATRIX3881 -163.4 .9023E-03 .5668 .01097E-054088E-014249E-055368E-01

P100 HODEL-0.9/30K, PLA=67,

-EXTRA POINT 7-1/23/76 FAA

				9					5								7				-01				10-0	-03		9	69	9	9		
- 4013	1000	3 127	-1.261	11385D	.1085	.1005	-20.26	6.623	J1186.	34.14	9.820	4364	4.527	.1173	.5921	2000	.36690	2487	.8275	3302	19101-	.1190	11011.	2960	12446D		-3.412		489	.163	2	-BR HO	
5		3,108	3733	-012174D-01	.4322	-19.43	8973	7179D-01	-051101b-02	4	25	3	8471D-01	3	3.105		1355	6135	1.919	5207	-	-01 .2768	_	6277	-013923D-01	-036182D-03	24	-056063D-01	2711D-0	3424	-01-19.97		
-3 902	30.00	-36.35	-4-546	.7554D-	-18.81	2.376	26.22	.5441	-8019D-	96.16	1.554	-6902D-	1.353	3064	-21.10		35/1	1970	-6465	2401	.01 .2193D	.9544D	.8928D	2284	1539D-	-052463D-	-2.635	9	9	_	-	26 88	
-713.3	2000	710.3	50.56	-9.184	39.47	128.6	-169.5	-11.70	021739	-1984-	-28.51	-1.269	330.1	6.724	461.1	•	•	7	41	7	1	_		1	ī	i	7	í	7	ï	9.	•	2000
0 200	200	20 59	-547.7	1 1.441	2.993	8.980	-7.562	14501	367520-	-88.11	417.1	18.54	42.04	1 .2026	270.3		4.7	1690	.5757	2320	17064D-	1 .8477D-	1 .7446D-	2051	1718D-	2673D-(	-2.386	-50.02	-2.001	-3.199	10-dfc01. TO	9304	2000
87.8	2000	-145.1				1.587	31.24	0-05465° I	3 .7927D-0	12.37	-112.5	1-5.002	-51.51	0-06811I	7.313																101		***
, 5	0007	3.280	0-0 99	8184D	6 180 D-0		1.046	~	2	-8.250	4	7	2496	.7	1.968		٥		3.922		7.	.5642	5	-1.403	1159	6685	-16.17	-9.874	3	864-4-	. 5024D-0	6 261	٠
THE A HATRIX	0030	9067-	2170	.2308D-02		.6313	1584	1000D-01	1505D-03	-1.849	2437D-01-	-02	1618	4822D-02	-1.052	2003	. 2303	5.483	.4272	-1.548	1009D-02	.6355D-01	.5668D-01	2	-50.01	6668	-1.780	-9.810	4360	-4.511	.7729D-02	1.231	

THE B MATRIX	RIX			
.2157E-01-223.5	1-223.5	-74.10	2.375	-277.4
.5678	-15.82	-8.139	-27.07	648.3
-1.516	69.59	22.03	32.91	2146B+05
.6232	114.2	-2.642	.4141	-1414.
.1868E-01	1	.1151	1367	-76.13
2184	33.66	21.02	84 17E-01	11 1553.
2910	82.60	25.30	5327E-01	11 1991.
.5137	-18.54	-6.101	3.687	-1842.
.1349	3.215	4624	5426	-6172.
-2029E-02		017080E-	.5796E-017080E-028187E-02-91.46	12-91-46
20.97	181.3	-72.98	-96.02	.5002E+05
-2956	-1-615	8980	-1-226	-6791.
.1316E-0	17040E-	-1316E-01-,7040E-01-,4024E-0	-	1-301.8
.1603	5.963	1501	9370	-2908
2450E-01-8056	1-8056	-2742	.3477	119.3
-2.152	-88-43	47.43	24.44	.1242E+05
THE C MATRIX	RIX			
.3693E-014391	14391	5.175	105.6	17.69 5.68127912998
-64 68 E-0	2 .5301E-	051301E-	034923E-(	E-029845E-043141E-04
0.	0.	0.	0.	0. 0.
-4551E-0	4 .2190E-	044438E-	031139E-C	81E-01-4171E-03-1352E-03-
2815E-04	1 2691E-	035550R-	02 1340E-C	2691F-03-5550F-02 1340F-02 3019F-01-2332F-02 1587F-03 4691F-04
.7026E-0	53220E-	043100E-	031545E-(	
.7236E-05	5 .5656E-	051492E-	034182E-	.5656E-051492E-034182E-039378E-021243E-04 .4169E-041459E-04
2669	3226	-1.059	- 4006-	- 7790R-02 838R-01 1 862 1 378 - 0131E-02
03030	016340	40400 -40	10611- AN	70 71616 - 10 41616 - 10 41616 - 60 40611 - 10
-3252E-	J-0433E	04-0043E	U31233E-CU	20E-03- 134/E-04- 312/E-04-
2	0.	1.000	٥.	0. 0. 0.
2307E-0	42098E-	032088E-	043094E-	
.2701E-0	4 -2458E-	03 .2448E-	04 .3623E-	-2701E-04 .2458E-03 .2448E-04 .3623E-04 .3299E-03 .4005E-04 .1103E-03 .5182E-04
3119E-0	42842E-	032821E-	044221E-C	
8365E-0	57675E-	047608E-	051136E-C	8365E-057675E-047608E-051136E-041035E-031253E-043448E-041619E-04
TOTAL U SH	114			
34 18F-01-36-14	1-36-14	5.643	-3,991	-63.00
1121E-032275	32275	.3113	-4912E-	.4912E-02-13.14
0.	0	0.	0.	0
.7793B-0	46176E-	023654R-	.7793E-04-,6176E-02-,3654E-02 ,5683E-06-,4671	064671
8364E-04	4 .7869E-	02 .1033E-	7869E-02 1033E-02-2745E-02 2704	22 2704
.1091E-03		025733E-	.6103E-025733E-036403E-034560	34560
-2857E-04		02 .3299E-	.1150E-02 .3299E-033642E-041298	041298

F100 HODEL-HN=0.0, ALT=0K, PLA=20-GHOUP3 POINT2 MAX HPEELD 1/21/76 FAA

	-1.802	-1.304	2.446	1180	122990D-02	.4217	.5182	-20.43	11 6.553	3	34.53		2	444 T	.8476D-01	1.183	-1.630	5318	1.207	.3831	115937b-03	.4415	.5478	4598	10-09171-10	326690-03	-5.936	-	2 .2059D	12287b	.12580-01	-48.17
	-3.626	-1.215	2.770	.6292	5206D-0	1.012	-18.76	-1.047	4369D-C	05-,640BD-0	-13.45	.6554D-0	01 .27960-0	0-01669-	19.96	4.312	-2.971	+168	2.056	.7580	01 .1624D-C	.7545	. 9235	7806	0126140-0	-034066D-C	-10.05	01 .4792b-0	-02 .2091D-0	-3833D-C	01-19.98	29.98
	-14.32	26.76	-45.39	-2.374	.1174	-16.77	4.638	28.11	.4593	-01.6817I-	216.8	6196	2756b-	.7348	3397	-35.46	-1.209	3694	8478	.3232	-02 .1638D-	.3131	.3887	3265	-1679D-	-022518D-	-4.189	-2308D-	A	-20.00	-07 00T. TO-	25.36
	5	-157	357	62	-24.			•	•	-056703D	-1691-	6.246	.2742	300.6	3.778	90.65	-8.662	-3.606	8.231	2.626	-032425D	3.026	3.755	-3.159	1-021372	-032090D	-40.51	.3724	-1.984	-1.497	-01 .6271D	12.64
	640.2	-250.9	54.16	-815.7	2.957	14.37	18.21	-15.12	4421	-057369D	-194.9	603.6	26.83	38.91	.4421	725.6	1.224	4050	.9185	.2918	-012094D	.34 14	.4181	3528	65331	_	-4.529	-49.95	•	-1.354	4	1.396
	-24.99	139.3	-149.2	112.2	-01 . 1748	7.693	10.15	52.86	-015235	-037911b	140.4	-200.2	-05-8-500	-89.59	-011955	9075	-1.881	-1.496	2.079	1.804	-031153D	.2309	.2866	2418	39.38	.5834	-50.89	38.54	1.713	17.29	-01 .1838D	-1.633
MATRIX	1.095	-3.933	5.238	.2381	-021291D	2251	3463	.1855	-012996D	-03-	-23.59	·	1		-02 .4183D	4.442	-8.309	.5419	6.322	1.264	-02 .6774D	2.324	2.848	-2.426	1054	6683	-31.14	-8.700	3865	-4-130	œ	10.22
EA	-2.339	5812	1.318	.2784	.8551D	.4137	.4395	4874	1482D-0	22340	-6.249	.2578D	.1134D-02	.4536D-01	9	-2.412	7024	3.007	1669.	5118	0	-2562	.3144	2669	-50.01	6668	-3.450	-6.933	3970	040-4-	59D	1.664

												**************************************	
	38.30 -58.74				140 -3958.	5.2 3537F+05	•	E-01	D74 -2121.		61.43 6209.	HE C MATRIX  -1321	.8186 -3.650 -461.2 .9834E-015636E-02-13.69 .0 .0 .0 -4276E-021450E-033699 .116E-02 .1307E-01 .6029 -2098E-038111E-032966 .7532E-042110E-036485E-01
	-2.839 -38.30	E-01-		75.448	8023E-0181403958.	-30.65 -315.2		8-02	.3032 -1.074	E-01	-15.13 61.	X	HE D HATRIX  .3936
BATR	.8863 -65.57	E-03-			24165-03-3743	44-46 -675-9		E-01	.2914 -3.246	3382E-01 .9608	-5.059 167.2	THE C HATRIX .13212508 .1692E-01 .1659E .0 .0 .3793E-04 .6687E .2245E-03 .1661E .2622E-041844E .1378E-043979E 3214E-016156 2461E-022230E .0 .0 7991E-047234E .1239E-03 .1119E .6749E-046091E	THE D HATRIX  .3936  .1331E-013276  .0  .0  .4030E-031232E-01  .3633E-03.2633E-01  .3256E-031051E-01

-GROUPS POINT4 1/23/76 FAA F100 HODEL-0.9/10K, FLA=20,

				-02					01					-01							-02				9	69			-02	9	9	
-1.578	-	2.081	30	19371D	1404	5	0		-	0	8.472	m	1 3.763	.6292b	.3509		9.	2439	8906	.2314	17833D	.1814	.6579	5166	19381D	312510	-3.630	.1001	2 .4503b	9	_	PL-61-
-6.282	-1.631	4.266	.3133	12466b-0	.6812	-17.42	-1.918	3213D-0	124820D-0	-13.40	.4780	.2121D-0	1446D-0	15.98	3.013		-2.557	8004	1.371	8004.	0-0808D-1	.2597	.9492	7456	121203D-0	141603D-0	-5.223	.1323	12 .5932b-0	9	3	
-19.35	18.99	-33.51	8634	-08999°	-18.30	7.549	27.89	.1609	02 .2355p-0	94.55	-3.214	1428	3406	1350	-9.281		3	2019	.7018	.3268	01 .1352b-C	.1394	.5037	3980	0172100-0	0396 13D-C	-2.797	9011.	-4803P-C	-19.87	01 .86520-0	21.25
-824.9	-60.75	160.5	372.8	-10.87	26.37	97.81	-73.10	6903	029204D-	-511.3	18.64	.8283	566.9	.24			-7.997	-1.677	6.234	1.634	-05459D-	1.247	4.514	-3.543	-056449D-	0485990-	-24.87	6489.	-1.969	-2.107	-06789. 20	5.826
454.8	-246.0	29.68	-610.3	01 1.799	660*9	21.27	-16.83	3050	-054066D-	-118.2	547.4	24.33	81.85	01 .3659	310.0		2.193	13	.6879	.1624	-016248D-	-01 -1385	8964	3927	7166D-	9554D-	-2.756	06.64-	-1.996	-2.038	-06658° 10	-6545
-17.45	117.0		02 99.35		2.309	9.234	47.59	02-,3556	045471D-	24.15	-162.0	01-7.202	-01-72.81	-01 .3282D-	9.245		-2.256	-1.809	2.317	2.286	-011775D-	-9183D-	.3223	2514	01 40.11	.5942	-50.21	39.98	1.777	17.96	-d1 +21 - 10	-1.467
.58	-3.383	3.895	-9563D-	0	1512	0609-	.2108	-02 .5285D-	7	-10.55	.4281	02 .	1 .6664D	-01071. EO-	1.212		-7.927	3.725	4.585	1273	-033761D-	.9233	3,305	-2.617	4776D-	5793	-18.34	-10.79	4795	-5.164	-02 .5731D-	4.967
THE A MA-2.693	5770	1-424	.1775	-d0069·	.3919	-6089	₽665*-	9	9	-4-178	-1485	-6605D-	2	3796D-	9495	24 SEE E-40 E	5094	4.896	₹209#	-1.079	40-	-1026	.3672	2908	-50.01	6667	-2-043	-11.45	5090	-5.183	-6367D-	1.038

	5	** 3.570 75.40 216.3 3.711370118605240E-024787E-01121652723442E-011382E-012862E-020000000000	-2092E-021882E-011820E-022827E-022550E-012878E-025366E-023717E-02 -0	
-8335. 477.7 -7253. 1622. -40.26 848.9 3437. -2739.	.1626E+05 .1626E+05 -3077. -136.6 -1620. 72.20	216.3 5272 .0 .021794E-0 023162E-0 027233E-0	016329 022550E-0 048682E-0 031533E-0 043491E-0	8.056 -370.0 -3371E-01-19.98 -0 -1306E-026344 -1034E-01 .4944 -2561E-03-1.118 -1496E-032569
-18.36 23.75 -4.629 -48.09 11.90 74.30 1.743 3.510 -2077 -1.988 5.688 -9.154 -5.223 -12.79	8385E-035845E-02-37.40 -36.19 -213.7 .1626 1.302 7.486 -30775794E-01 .3329 -136.6 .7187 .7280 -16201797E-01 .3565 72.20	75.40 -011216 -024132E-01 .4312E-027295E-031560E-03	55736043E-016329 2092E-021882E-011820E-022827E-022550 0	HE D MATRIX -2035 -140.6 -2.175 -8.056 -370.0 -1054E-014813 .1388 .3371E-01-19.98 -0 .0 .0 .0 .0 -3402E-031984E-013334E-02 .1306E-026344 -2627E-03 .3470E-01 .1272E-02 .1034E-01 .4944 -5941E-033515E-018763E-032561E-03-1.118 -1358E-037915E-022783E-041496E-032569
-18.36 -4.629 11.90 1.743 -2077 5.688 -5.223	-018385E-03 -36.19 1.302 .5794E-01 .7187 .1797E-01	3.570 -024787E041628F031194E042212E054815E-	5573 -011820E- 1.000 -036192E- 03 .6470E- 021100E- 032500E-	-2.175 .1388 .0 -013334E- -01 .1272E- -018763E-
4.585 -284.6 4.585 -284.6 4.596 -90.78 2.134 189.7 5.247 165.0 2.1382-01-28.04 4.909 30.81 1.837 93.96 1.390 -76.89	33.26 -641.0 1285 21.91 5809E-02 9758 1084 -4.539 3086E-01 1.840 2.269 57.53	## C HATRIX -26844124 3.570 75.40 -1820E-01 .2440E-024787E-011216 -0 .0 .0 .0 .0 -5802E-04 .9771E-041628E-024132 -1522E-03 .8903E-031194E-01 .4312 -5792E-05 .3643E-042212E-027295 -1006E-04 .3148E-054815E-031560	-2429 -021882E- 0 -046412E- -031136E- -042584E-	HE D HATRIX -2035 -140.6 -1054E-01-4813 0 0 -3402E-03-1984E- -2627E-03 -3470E- -5941E-03-3515E- -1358E-03-7915E-
THE B MATRIX 4.585 -2 .4596 -9 -2.134 1 -5247 1 -4909 3 -1.837 9	.1311E-02 33.26 1285 5809E-02 .1084 3086E-01	THE C HATRIX26841820E-0105802E-041522E-035792E-05.	.1081 -2092E 0 -7125E .7451E 1263E	THE D MATRIX 2035 -1 1054E-01- 0 3402E-03- -2627E-03 - 5941E-03- 1358E-03-

P100 HODEL-0.9/30K, PLA=20,

-GROUP3 POINTS 1/23/76 FAA

3547 6476 -9252 -5654D-01	2467D 3290D 9.97 -716 8467D 2.39 -808	. 6168D-01 -7325 1250 11316D-01 1974D-01	2444	
-1.713 7265 1.830 .7355		19.94 1.686 3667 -018944D-0	999999 999	
-5.331 11.40 -20.66 8371	-18.67 4.025 28.61 -1806 -01.2643D -1.983 -1.983	- 1815 - 1815 - 12.51 - 2701 - 01- 1000D	999999 99 9	
-673.0 -72.56 183.7 219.0 -8.050	51.63 128.7 -127.5 -1.926 -022997D -887.8 20.87	602	00000	
424.8 -227.6 6627 -478.5	-01-2.319 2.054 3313 -014418D 12.76 799.7	100.1 366.8 2.220 9653D 1448D	-01-4370-0 -01-23170-0 -01-30890-0 -01-32820-0 -48270-0 -64360-0 -2148 -50.00 -2.543	
3.790 102.9 -116.1 -01 103.6	00 00	3.101 3.101 -1.178 -1.105 -1.105	0-01-1155D -1411D -1964D -2087D -01 39.89 -48.08 39.61 1.761 17.99 0-02.7978D	
#ATRIX -4.550 -1.939 1.948 D-01 .5917]	0-02 6496 0-02 63760 0-04 95210 0-01-90540 0-02-40300	D-02 15811 1.090 3186 2.995 D-011274		
THE A H - 2267 - 5860 - 9727b - 4511b		4823D -1.418 -2336 3.030 1062D	32 4 1 1 6 6 2 2 6 6 8 6 8 6 8 6 8 6 8 6 8 6 8 6	

P100 HODEL-0.9/30K, PLA=83,

-GROUPS POINTS 1/23/76 PAA

	6736	7348	.9117	4422D-01	2165D-01	2830	1136	-19.53	6.810	.1009	41.62	10.04	1944.	4.595	.1014	-2.674	1001	.6829	-1.977	1.148	1709D-01	-,3039	1266	1861	.3754D-01	.5482D-03	5.797	0-0	0	9	7170	-52.32
	.8456	2.510	-6.832	1.826	12451D-01	6523	1-21.34	1.569	.1141	-1700D-02	18.29	.3154	.1396D-01	.1799	19.75	-6.379	3063	1.032	-3.130	2.211	11188D-01	#66h*-	11318	.8110	1.7325D-01	-11	9.472	1.1731	r.	30 D-0	1-20.05	•
	-1.834	15.63	-49.08	-2.714	.7899D-01	-20.19	.7039b-01	31.39	.7885	.1169D-01	119.1	2.514	7111.	1.730	5225	-33,37	4750	.5223	-1.528	1006.	. 1944D-0	2317	8445D-01	.3826	-	2 .44450-03		-7917D-01	.3500D-02	-19.73	211	29.08
	-824.5	-385.2	0.606	186.0	-10.33	71.97	199.7	-161.6	-13.91	112062	-2080-	-47.40	-2.105	283.0	9.257	522.9	4.476	4.383	-12.89	5	1111	-1.958	117833	3.212	.24	373	7.	-	96.	3	111725	-15.12
	403.0	-218.8	-56.54	-534.3	11 1.345	-8.435	069-1-	13.98	1.205	1792D-0	162.7	375.4	16.68	09- 11	7786	155.6	5.033	74947	-1.432	.8315	0112210-(	2202	19066D-C	.3569	.2720D-C	.3972b-C	4.200	-49.93	-1.997	-3.075		-1-690
	4.397	131.8	-175.5	-	N	-	-	35.21	-	3	84.53	-97.77	11-4.346	69* 11 1-	113979	-21.74	11-2.258	-1.856	2.049	3.656	12727D-(	1338	5507b-C	.2213	39.38	.5833	-45.09	38.41	1.707	17.64	-d600L.	-3.464
HATRIX	.1111	.85	2.887	42	28553D	3914D-O	5790D	.89	-	15(		-014538	-0320170-0	2471	11 .2696D-C	1.586	.7020D-C	9.477	.72	4-124	027898D-0	-1.492	-	2.446	.1843	19	28.46	3	901	23	111334	-10.84
	-2.329	1354	.3097	.4622	-2666D-0	-8065	.6132	50650-0	3652D-02	5519D-04	5827	1631D-0	0	1665	1656D-0	-2.016	.5050	6.524	-1.088	9184	4	1665	70590-0	.2718	86.64-	9	3.184	-9.558	4	-4.397	25	6678

	.4282 5.215 108.6 -77.55 6.624 .25774507 .83772-05 .3070E-03 .1622E-021586E-01 .5499E-04 .1468E-03 .5953E-04 .0 .0 .0 .0 .0 .0 .1330E-04 .4597E-03 .2032E-022636E-01 .4595E-04 .2270E-03 .6830E-04 .1756E-035560E-022219E-02 .2818E-011373E-022498E-037446E-04 .2067E-04 .6696E-03 .1963E-022503E-01 .5912E-03 .3473E-03 .6555E-04 .6537E-05 .9373E-04 .5006E-036472E-02 .7284E-04 .1120E-03 .1685E-04	24212696 -1.2638662E-017609 2.030 1.0591156 3412E-04 .3071E-03 .2086E-04 .4533E-04 .4080E-03 .4862E-04 .8323E-04 .6257E-04 0.	
1.276 -616.6 -25.61 7673.8 8.3365151E+05 10.78 980201-1694 -113.4 -1.803 -13623.540 -3147. 15.60 52523.118E-01-641802-3852E-03-95.09 -10.16 -59411516 -59412356 -19072356 -1769E+05	108.6 -77.55 6 .1622E-021586E-01.0 0.2032E-022636E-01. .2219E-022818E-011963E-022503E-01.	1.2638662E-017609 2.2086E-04 .4533E-04 .4080E-03 .0.3206E-04 .5256E-04 .4717E-03 .3521E-045731E-045155E-03 .3089E-04 .5066E-04 .4523E-03 .7977E-05 .1304E-04 .1154E-03	-217.6 -03 .4622 .0 -03 .5573 -027096 -03 .5805
1.276 -616.6 -25.61 7673.8 8.336 -5151 10.78 980201-1694 -113.4 -1.803 -13623.540 52523.118E-01-641802-3852E-03-95.09 -1516 -59411516 -59412356 -19072356 -1907.	108.6 3.1622E- 0 3.2032E- 22219E- 3.1963E- 4.5006E-	8662E-01- 4.4533E-04 0.5256E-04 4-5731E-04- 4.5066E-04 5.1304E-04	7.852 -4.273 -0.3533 0.0998F-03 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0
-60.21 5.762 -17.38 -14.94 -9722-01 7.650 8.861 4.624 .3187 -47322-02 55.26 .7203 .31822-01	5.215 .3070E-03 .0 .4597E-03 5560E-02 .6696E-03	.2086E-04 1.000 3206E-04 -3521E-04 .3089E-04	7.852 .3533 .0 .4735E-02 .7567E-03
02.2 58.7 58.7 10.0 13.3 13.3 13.3 15.0 15.0 10.0 10.0 10.0 10.0	## C HATRIX 7722E-01-4282 7789E-02 .83773-05 0 .0 1995E-03 .1330E-04 7360E-05 .1156E-03 2516E-04-2067E-04	2421 - 2696 34 12E-04 3071E-03 0 3983E-04 3563E-03 4349E-04-3886E-03 3822E-04 3427E-03	29.0 2647 0 2490E-01 1446E-01 2601E-02
THE B HATRIX  -1.020 -3.58.7 3.353 754.1 -1.723 -1.723 -3.58.7 2634E-01-44.94 4187 -2517 438.3 -2786E-01-7.590 4111E-031166 2.629 -2027 -1013 -9.522 4518E-024101 -6723E-01 92.41	THE C HATRIX -77228-01-4282 -77892-02 .8377 0 0 0 0 .19958-03 .1330 -79608-05 .1156 -25162-04-2067 -11722-04 .6537	.2421 .3412E-04 .3983E-04 -4349E-04 .3822E-04	1630 1335E-03 0 -1052E-03- 1314E-03 -2738E-04-

HATE							
			1.208	2175.			
B-01				1397.			
				1876E+05			
				-4897.			
E-02	4.02	. 1383	H	-01-1.751			
	-27.03	10.32	.2015	-181.8			
.4916 3	345.1	8.152		710.4			
4137 5	50.36	9029-	11.81	2053.			
.1598 9	- 771-6	8476E-013778	3778	-6801.			
.2391E-02 .	.1320	1286E-025708E	5708E-02	-02-100.8			
21.18 2	206.9	-6.023	-61.31	-9448E+05			
.4206	15.64	3345E-01-1.327		-6796.			
. 1869E-01 .		1393E-025875E	10-	-302.1			
.2038	60.91	.5754	6366	-3457.			
.1685E-01 .				-85.07			
T		-82.81		1735.			
THE C HATRIX							
.7697E-011988	1988	8.141	119.8	-49.34	3.835	- 1949.	2551E-01
.4325E-02 .	.1718E-05	0.		140 TE-01-	1331E-04	.3075F-04	0.
0.	0	0.	0.	0.		0.	0.
.1987E-03 .	.7321E-05	.6156E-04	.8045E-03	.8045E-036199E-018211E-04		.1196F-03	.8719E-05
	1350E-03-	-1069E-01-	-9059x-03	.1350E-031069E-019059k-03 .6805E-011355E-02	-1355E-02-	1321E-03-	9731E-05
	2823E-04	.5082E-03	.7624E-03	.7624E-035686E-01	.4907E-03	.2400E-03	-8079E-05
.1053E-04 .	4964E-05-		.1978E-03	.1978E-031458E-01	.4436E-04	.8523E-04	.2102E-05
. 2381	- 9265	-,5363	.1128	1.034	1.196	-9527	.1551
		0.	0.	0.	0	0	0
	0.	1.000	0.	0.	0	0.	0
.5957E-05	.4749E-04	-2879E-05	-7021E-05		.7545E-05	.1872E-05	.9647F-05
	54 14 P-04-	-3199E-05-	7956R-05	7404E-04-	.8559E-052145E-05	.2145E-05-	1088E-04
.5398E-05	.4491E-04	.2655E-05	*6608E-05	*6040E-04	.7124E-05	. 1855E-05	.9165F-05
.1415E-05 .	.1199E-04	.6647E-06	.1767E-05	.1561E-04	.1898E-05	.4085E-06	-2439E-05
THE D MATRIX							
4944 2	29.82	6.570	7046	1539.			
E-03	-2024	661	.3696E-03	1.544			
0.	0.	0.	0.	0.			
.1284E-05 .	-2310E-01	.5854E-03	.1698E-03	.5221			
1	1762E-02	-9896E-04-	.9896E-041677E-021824	1824			
	.1251E-01	.1359E-02-		.1285			
6789E-05	.4453E-02	.5857E-03	.4831E-04	.3095E-01			

-04-.3537D-04 -.5121D-01 6.745 -05-.7353D-03 -01-0171D-01 GROUP 4 POINT 2-2/11/76 PAA .7356D-1.071 2.726 .3214 -19.40 -. 7484 -8.205 -.2403 -. 1970 -1.481 -.5751 19.91 -.2429 -.8774 1.324 -.7012 .7431D 4747b .4678D -2.457 2,359 108.4 1.133 1.056 .4925 -18.78 .5624 -.2936 -.3601 6.154 -16.45 -2.575 29.07 -22.98 -279.6 4.057 125.5 226.2 -.2263 705.4 554.2 1.657 -7.949 -15.12 -2.971 613.4 46.90 -3055 -48.44 -2.160 9.257 J2707. 1.880 -2.652 39.58 2.086 805h. -266.3 35.80 9.095-5.622 .5364 61.93 81.37 .3182 492.8 -.3270 .7947D 2.875 .8835 117.9 .1227 .5259 1.446 -1.143 .3225 95.22 -.2862D-02-.1881D-01-10.81 -111.6 1-. 1963 -10.97 -.3755b-01 112.5 -243.1 P100 MODEL-0.9/45K, FLA=52, .3512D-02-.2675D-02 --2339D-01-.7122D-02 -.3449D-03-.1065D-03 -.9194D-01 -2080D-0 1.377 1904--.1766 -7.226 --6440D-01--4232 -.2404 .8696 3.035 -1.373 1.491 -. 1496 THE A MATKIX --6495D-02 2.368 -.1753 .3349 -1.165 -8046 44.744 -.3869 .8829 .1161 .4195 -.4284 -2.017 -.847 -.194 .331

4293	.3319D-02 .1374D-016714D-02 .1727D-02 .1647D-01 .1543D-01 .1541D-01 .2467D-02	2378D-011880D-013745D-01	4620D-011567D-016850D-01	-01 .7855D-01	39.36 . 7936u-02 .7143u-01 .3397u-027834u-02 .1142u-01	-03 .1522D-03	.8859	7535p-01	2446F-021050D-013375D-02	4658D-01	.5480D-02	-50.62	
6580	-01 .1541D	-011880D-	-011567D-	-01 .1253D-	-027834D-	-04 1045D-	.2115	5435D-012350	-021050D-	1285	02-20.00	23.01	
3057	-01.1543D-	2378D-	46200-	-06675°	-d795. 10	-03 .4529D-	.5978	5435D-	24461-	-19.79	.2850D-02 .2565D-01 .1424D-01 .3809D-02 .3428D-01 .2717D-02-20.00	26.86	
-2.714	-d7431. 20	012343	014343	4164. 10	-dE417. 20	03 .9523D-	5.543	5143	-2.023	-3.463	02 .3428b-	-3.857	
30 16	-07271- 2	1383D-012603D-012343	2562D-014825D-014343	1 .5460D-	-1936₽-	-1058D-	.6159	-50.07	-2.003	-3.147	- 13809D-	4286	
1.053	16714D-0	1383D-C	2562D-0	.3010b-0	1 35.36	.5831	-47.31	38.55	1.713	17.72	1 .1424D-0	-2.703	
-2.651	2 .1374D-0	11753	1-,3249	-4085D-01 .3677 .3010D-	.5344D-0	6660	4.211	-10.17	4523	-4.737	2 .2565D-0	-2.277	
8479	-3319D-0	1948D-011753	3895p-01-3249	.4085D-0	-50.00	9999-	.4786	-9.809	4360	-4.516	-2850D-0	-2945	

-.8220D-01

.6485

6hLn . -

.9496D-01

-1.031

-.3328D-01 -.6087D-01

-19.43

1.672

-.8157

-1.019

-,3125

.9991D-01

1 16.6

.4418 4.554

38.40

ATA	18 -163.0		.7511	-478.9			
-2.194	372.8	24.22		-8209.			
1.653	118.8			-1676.			
E-02			E-02	15.07			
.1287	113.9			108.5			
.2454	181.6			-105.9			
	-221.2			543.3			
.1810	-9.551			-5587.			
.2705E-021468	1468	7365E-02	3406E-0	2-82.78			
29.61	-2423.		-45.81	.7460E+05			
.8531	-21.88	-2.135		-6722.			
.3803E-019824	9824	9510E-01	E-01	1-299.1			
.5054	-30.87	6140		-2996.			
9389E-02	7.694			-19.16			
2.498		-82.61		-2142.			
THE C MATRIX	1X 1839	4.187	102.6	30.26	2.536	- 3466	- 1856
696AP-02	6259F-04	EU-4840E -10-46564		14929-02- 8485V-01- 8201V-03- 234VP-03	9201F-03-	23145-03	400000
10000		70000		2000	2000	0	0
12998-03		3164P-04-1412P-03		8786P-03-4342F-01-4247P-03-1164P-03	- 4247F-03-	-1164P-03	10991-04
8213F-04		3374F-03- 1484F-01- 1068F-02 528RF-01- 3268F-02	- 1068F-02	5288F-01-	3268P-02	14201-03-13498-04	
8835E-04		5331B-03	1372E-02	1372E-026731E-01	3326B-03	1297E-03	1762E-04
.2101E-04	.6689E-05		.3639E-03	.3639E-031758E-01	.2397E-04	.6582F-04	.4667E-05
.9417E-011441	1441	5043	3908E-013677	3677	.7985	.5964	5731E-01
.1188E-04	.1069E-03		.8897E-05 .1587E-04	.1429E-03	. 1486E-04	.4896E-05	.2283E-04
0.			0.		0.	0.	0.
.6651E-05	.5887E-04		.4910E-05 .8742E-05	.7868E-04	.8318E-05	.3022F-05	.1253E-04
8153E-05	8153E-057202E-04	60 13F-05 1070E-04	1070E-04	9577E-041023E-043519E-051529E-04	-1023E-04-	-3519F-05	1529E-04
. 1032E-04	.9321E-04	.7557E-05	1384E-04	.1246E-03	1322E-04	.4648F-05	-2001E-04
-2760E-05				.3299E-04		. 1262P-05	.5273E-05
BATE	n						
- 8592E-04- 2085	-85.22	03380	-1.508 -172.1	-172.1			
0			0				
4570E-04	3972E-01	-4570E-04-3972E-01-5601E-02 .2813E-03-3200E-02	.2813K-03	3200E-02			
.9843E-04	.9843E-04 .4441E-01	.1609E-02	.1609E-023257E-021735	1735			
7484E-044987E-01	4987E-01		.1398E-024821E-04	9601.			
1949E-04	1949E-04 1318E-01	.8159E-03	*0-36659°	.2436E-01			

P100 HODEL-0.9/45K, PLA=20, MIN PB PLAP=40, GROUP 4 POINT 3-2/13/76 PAA

.3784b-022147b-02 .7682b-01 1.578 -7.823 .2758b-016155b-022396b-02 .53229530b-01 2.234 .3295 125.6 -18.44 .5040 .180Rb-02 .44792083 4.372 .4759 231.1 3.043 -19.07 .9493b-02
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GROUP 4 POINT 4-2/12/76 FAB F100 HODEL-1.8/75K, PLA=83,

	81D-01-	10-01464 103	1712. 111	52720	D-02-,4603D	70	•	-20.1	-02 6		(7)	S)	-02	7	9.91 .8373b-01		10-0	•	i	3336 .2075	20-0	10-0	.2873 .1209		-01	-03	7	•	D-02 .	211. 115	•	2- 5
	-1.275 -73	•	i	•	1842D-01 .	•	.501	-30		02	7		10-0		-	-10.08 1.3	D-01.	•	i	.1560 .33	02 .83350-02 .91	61.	-01	108730	95	4D-04	•	6D-01	-02.	.58	90D-02-19	56 27
	6 -613.0	•						•	•	-022	'	-3	-1.	-		. 820.5	-		•	2 1.253	-03	9	5	-01-	9	9	7 -7.247				0-03 .8	9
	5.231 396.				.1116 1.945					1230-01	1.77		15	7	10-01	34.17 1091		.3232 .1889	-	.7536 .1392	.3847D-02 .5667D	0-01	_	0-01-			8.288	7	- 117.	-3	0-01	
HATRIX	.1297	9388	'	.2655D-02	8D-02	9118D-01	8	153	3D-02-	5463D-036069D-04-	3	2	10-01-	2569	30-01-	1.110	n	2.534 -	-1.955	.6461	-02 .49290-02-	-	-01 .5645	1	.3401D-01	6662	5.509	•	-,3816	-3.652	-03 .6802D-02	h hes
	-	1807	.5947	.1349	.9314D-	.8318	.7314	7206	3676D-	5463D-	-6.855	à	4008D-	2293	9	7400	.1341	1.383	2154	2097	.1455D-	.2116D-0	.6612D-0	6612D-	-50.00	9999	6433	-9.126	4056	-4.136	-7557D-	ORES

THE B MATRIX5152 -1.627 2.641	109.6 -109.6 1.317 29.90	-16.52 -1.783 11.32	4.841 -7.881 15.74	496.3 2802. 1060E+05	
-1.196 4516E-02-11.11 1536E-01 166.9	11.16 12-11.11 11 166.9	4.796 .1181 4.848	4.617 4361E-01 -2.921		
	-77.76 -3.933	-20.78 -9772	-9.074 -8.812 -1.169	-1280. -6438.	
62.94 62.94 .5537	-576.7 -14.23	-201.8 -227.4 -9722 -2.387 - 42558-01-1057	4	.6359E+05 -6064.	
.1041 1900E-0		.7064 1801 -63.04	-2.198 -4781 43.75	-2065. 43.84 8147.	
THE C HATRIX8017E-012613E-02 5933E-04 1105E-038666E-05	HI I	17.446 146397E-( 141012E-( 132416E-( 151030E-(	THE C HATRIX8017E-019197E-01 7.446 129.7 43.12 .2613E-02 .1968E-046397E-031606E-022781E .0 .0 .0 .0 .5933E-04 .3097E-041012E-022222E-024520E .1105E-03 .3259F-032416E-01 .3077E-02 .6236E8666E-055009E-051030E-025172E-0210374662E-05 .4523E-053523E-031299E-022591E	**************************************	**************************************
.9873E-01 .07476E-05- .7476E-05- .1043E-04- 1760E-04-	11 .3447 54251E-0 156459E-0 14 .8897E-0 141511E-0	2542  3045E-(  4933E-(  4933E-(  2653E-(	.4048E-01 .0.6213E-05- .059471E-05- .1310E-04 .2207E-04- .55479E-05-	55592E-048407 0.0 58524E-041252 4178E-031739 41987E-032880	.9873E-01 .34472542 .4048E-01 .3644 .5527 .5996 .5814E-014723E-054251E-043045E-056213E-055592E-048407E-052121E-049102E-050
-343734376523E-039768E-041160E-032034E-035139E-04	**************************************	747721203554 .1486 .0000 .00 .00 .00 .00 .00 .00 .00 .0		496.1 .5795E-03 .7870 .0 .9970E-031003 .4085E-032863 .1002E-036751E-01	

GROUP 4 POINT 5-2/12/76 FAA F100 HODEL-1.8/20K, PLA=83,

67 37 10 10 10 10 10	.4828D-02 .7716D-01 .1261 20.17 5.848	49 11 61 00 89D-01 86	1.982 5.182 7.602 1.855 .2045D-01 .9255D-01	.7065D-02 .9420D-04 1.524 .9891D-01 .4380D-02 .6147D-01
.3167 1.037 1296		37.49 8.911 8.911 3961 4.100 78891	1.982 5.182 -7.602 1.855 .2045 .9255 -2077	9889
.6154 5.740 -4.907	.8863D-02 .4406 -19.26 9939 3878D-01	-7.263 .1112 -01 .4964D-02 -4137D-01	1.535 6.935 -9.134 2.792 -01.2308 -01.3740 5204	-021803D-01 -042748D-03 -3.797 .1674 -02.7419D-02 -2298 -02-19.99
-9.167 17.63 -69.69	-1677 -18-77 2-278 26-81 -2894	45.42 -01 .6351 -02 .2827D- 1.062 9833D-	1.075 3.994 -6.047 1.340 .1059 .5629D-	99 9 9
-568.2 33.28 142.0 81.46	99	-312.9 4571D- 1828D- 118.4 -01.7680 92.52	25.50 32.72 -48.04 11.73 -01.1226 -01.5796 -1.311	-02445 -04594 -9.56 -9.56 -1.97 -2.86 -02 .222 -02 .222
434.0 -179.3 -84.85	2.059 2.059 .9885 -1.313 1.5411D	-9.795 152.0 6.756 16.86 1 .2886D	14.50 3.628 -5.337 1.295 1.1437D 1.6440D	4954D 6605D -1.063 -49.92 -1.996 -3.246 1.2477D
7.642 123.2 -183.4	2 .1506 1 .5699 .9963 17.32 2 .8963D—0	10.03 -40.80 1-1.813 -18.44 22076b-0	-4.637 -3.080 4.176 8.7914148D-0 -4744D-0	135.85 -48.9 39./1 1 38 1 323D-0
1.403 -4.913 5.166		- 32 - 14 - 22 - 25 - 25	10.89 39.81 -36.23 4.788 11.1085 11.7055	
THE A MAT -7.944 -2.225 8.756	.2358D-01 .8499 .5476 -1.099 4955D-01	-8.049 5937D-01 2636D-02 1554 -1963D-02	2.488 17.98 -4.063 -3.018 -2365D-0 -4666D-0	-50.00 6667 7671 -9.733 4326 -4.463 -1866D-02

																	.9407 .3303	85E-02-	0.	145E-03-	1286F-63 2724E-03 2020E-02 1563E-63 5006E-02-1649E-02 1163E-03	E-04 -1458E-03 -4670E-04		7 785 1 409	504-1	2770E-03-48/1E-07-3480E-03-1/430E-03-608/E-07-64 33E-03-26/9E-02-1060E-02	0.	1180E-041105E-031202E-041644E-0414/9E-031425E-045854E-042354E-04	-1228E-04 .1146E-03 .1250E-04 .1699E-04 .1529E-03 .1488E-04 .6069E-04 .2439E-04	2834E-042646E-032869E-043959E-043563E-033421E-041398E-035633E-04	6826E-056398E-046973E-059548E-058593E-048194E-053387E-041360E-04									
.1918E+05	.5612E+05	1560E+06	.2377E+05	173.2	1212.	2120.	-2655.	-5709.	-84.62	.5094E+05	-5203.	-231.4	-2393.	61-66	.1618E+05				0.	.4999E-0232401	5006E-02-1649	1422E-057563E-051387E-033602E-031164E-015169E-04	.2367E-054734E-048684E-042840E-024489E-05	8 948 7 058	0000	.668/E-026433	0.	.14/9E-031425	.1529E-03 .1488]	.3563E-0334211	.8593E-0481941		12600.06	-17.48	0	3415	.3194	7602	1855	
160.5		•		8519	-7.997			-2.982 -	E-01		•	E-01		1.003	93.26		137.7	194E-02-	••	.031506E-03-	02 1563E-03	03-3602E-03-	-048684E-04-	4907	1066	-03 14 30E-03-	0.	-041944E-04-	-04 -1699E-04	-40-3959E-04-	-029548E-05-			1051		-2214P-02-		-1346E-02-	-03 .1093E-031855	
-189.0	-16.12	154.9	39.16	1.272	8.613	15.45	-27.78	-1.266	-011863E		5934	2621E-	1.707	9820E-01	-43.52		10.32	-026982E-	0.	-04 1463E-	-032020E-	-051387E-	-05-4734E-	-2.717	1000	-02 2480E-	0000	-031202E-	-03 . 1250E-	-032869E-	-046973E-		2 1.13	1.5413		-025287E-	-02 3275E-	-02- 1011E-	-03 .2613E-03	
B BATRIX		•		1023E-01-132.1	5238E-01 7.528		-1075 -21.43	.3384E-01 1.122	•	1	10-3	.2014E-02 .7633	4474E-01-35-25	2036E-02 -6067		THE C HATRIX	1007299	.4606E-01 .1038E	0.	.1327E-03 .2335E	386P-63 2774E	122E-05-, 7563E	.4919E-06 .2367E	1,602 7,582	7000	220E-0348/1E	0.	180 E-04 1 105E	228E-04 . 1146E	334E-042646E	326E-056398E	*10*1* U 40*	TT SOS C	R-03-	0	1271F-04-4393F-02-5287F-03	1335E-04 .5211E-02 .3275E-02	135E-04 1313E	7488E-05-3030E-03	
THE B -1.119	-2.643	4.010	9272	t	52		2.	.33	64.	5.5	.45	.20	44.	20	7315	THE	-1.100	94.	•	.13	-12		54.	1		700	9	-	-	28	66	400	100	59	9	-12	-13	.31	.7	

-05-.1032D-01 .8171D-02-.5354D-02 -.3204D-01-.6973D-01-.4301D-01 2-.4855D-03-.1069D-02-.6483D-03 .70290-01-1-131 -.2460D-01 -02--37490-03 .8139D-01-.6748D-02 -.5905b-01 GROUP 4 POINT 6-2/12/76 FAA -15690--. 1511 6.715 -11.59 -1.603 2.980 -3.269 .1448 -19.75 --2263 -2.706 19.85 1.265 -1.192 -3.089 .3682 1.376 -2.845 -.9957 .2614D-01 .9043D 11.70 1.832 2.578 -4.310 .1247 98.49 -26.05 .5313 -19.53 29.79 .6108 -. 4389 -.8302 -1.203 -1.209 -.3682 -39.50 -7.933 . 1044 1.930 1.382 -.3367D --4702b-03--4232b -214.6 -17.29 -56.51 1.660 -10.08 21.68 -10.16 1.060 844. -3.071 --3120D-01-.280B 90.09 -2.513273.4 1207. 17.40 -9.317 235.5 -.2577 -.2165b-01-.3584b .1668D-03-.1259D . 1177 64649 -4.020 -591.6 -1.118 2.406 -.3413 .1251b-01-.8499 -1.129 -275.6 78.75 1.585 3.532 15.85 -10.93 -128.1 362.4 16.11 35.30 .6043 .5890D 1-.3002D .8506b .261 .3221 116.9 36.3 116.8 910-9 -98.60 -.2207D-02-.2045D-01-4.382 -2.538 2.428 -.2315 4.503 27.97 -2.951 4.548 39.32 P100 HODEL-0.3/20K, PLA=83, -45-22 6.592 .5826 -50.36 -- 1348D-01-. 1525D-01 --2030b-03-.2274b-03 .3756D-01 -4308D-02-.7050D-02 .7213D-02 .59590-01 .3183D-0 .1994b-02-.2116p .7966 3.370 3.300 .9486D-01-.2113 -6.880 --4967D-01-.4600 --2460 -3.050 1.947 -2.336 -2.164 -1.302 16.30 -9.223 --2109 -.6698 -27.25 THE A MATRIX --8687D-02 .8515D-01 -2.130 -1.423 -2.366 8044--- 1786 .3679 .7108 -. 1825 5.447 -2.988 1.030 1.793 -2.412 -.2534 -50.02 -.6670

4.515

-1.548

-.5648

-1.556

3,325

.4375

5.843

1001. 31.18

-2.628

.1543

-6479

-20.45 6.760

-2.880 5.949 -.8827D-01

.26 18D-01

.2039D-01-19.95

. 1899D

.3347D

.1283

.1426D

1.703

38.32

-10.50

-9.738 --4328 -4.485

--4664 -4.873

-3.228

-.54 18D-02-. 1450D-01-.7855D-02

-.3242

-. 1209

-1.147

-50.13 -2.006

-2.051 -3.770

-.1758

-5.554

-.4705

.6815

.1623

2.436 63.34 -21.08 -24.32 -3108.  2.534 448.1 -21.08 -24.32 -3108.  2.534 448.1 -21.08 -2.34 -7.6502-01-27.47  -2.534 448.1 -2.394 -7.6502-01-27.47  -2.534 448.1 -2.394 -7.6502-01-27.47  -2.534 448.1 -2.394 -7.6502-01-27.47  -2.534 2.30.40 -3.394 -7.6502 -2.30.40  -2.534 2.30.40 -3.394 -7.6502-01-27.47  -2.547 20.40 -3.599 -6.75.3  -2.592 -2.30.4 -3.599 -6.75.3  -2.592 -2.30.4 -3.599 -6.75.3  -2.506 -3.40.4 -3.30.4 -3.67.3  -2.501 -3.30.4 -3.50.4 -3.50.4 -3.40.4  -2.714 -3.00.4 -3.20.4 -3.30.4 -3.40.4  -2.714 -6.2.1 -3.00.4 -3.40.4 -3.50.9  -2.714 -6.2.1 -3.00.4 -3.00.4 -3.00.4  -2.714 -6.2.1 -3.00.4 -3.00.4 -3.00.4  -2.714 -6.2.1 -3.00.4 -3.00.4 -3.00.4  -2.714 -6.2.1 -3.00.4 -3.00.4 -3.00.4  -2.714 -6.2.1 -3.00.4 -3.00.4 -3.00.4 -3.00.4  -2.714 -6.2.1 -3.00.4 -3.00.4 -3.00.4 -3.00.4  -2.714 -6.2.1 -3.00.4 -3.00.4 -3.00.4 -3.00.4  -2.714 -6.2.1 -3.00.4 -3.00.4 -3.00.4 -3.00.4  -2.714 -6.2.1 -3.00.4 -3.00.4 -3.00.4 -3.00.4  -2.714 -6.2.1 -3.00.4 -3.00.4 -3.00.4 -3.00.4  -2.714 -6.2.1 -3.00.4 -3.00.4 -3.00.4 -3.00.4  -2.714 -6.2.1 -3.00.4 -3.00.4 -3.00.4 -3.00.4  -2.714 -6.2.1 -3.00.4 -3.00.4 -3.00.4 -3.00.4  -2.714 -6.2.1 -3.00.4 -3.00.4 -3.00.4 -3.00.4  -2.714 -6.2.1 -3.00.4 -3.00.4 -3.00.4 -3.00.4 -3.00.4  -2.714 -6.2.1 -3.00.4 -3.00.4 -3.00.4 -3.00.4 -3.00.4 -3.00.4  -2.714 -6.2.1 -3.00.4 -3.00.4 -3.00.4 -3.00.4 -3.00.4 -3.00.4 -3.00.4  -2.714 -6.2.1 -2.4.10 -2.2.2 -2.20.4 -3.00.4	263.4 -21.08 -992.6 46.77 448.1 -18.17 E-02-46.99 .2394 6.058 11.92	0
63.4 -21.08 -39.38 -8978. 6.39 -2394 -750E-01-27.47 6.99 -2394 -7550E-01-27.47 6.99 -2394 -7550E-01-27.47 12.92 .8290 1271. 12.92 .8290 -2649. 0.40 -5999 -2.579 -6755. 0.40 -5999 -2.579 -6755. 0.40 -29056E-02-8533E-02-100.2 3037 -9056E-02-8533E-02-100.2 1.12 -8214 -9720E-01-359.9 1.12 -8214 -9720E-01-359.9 1.12 -8214 -9720E-01-359.9 1.12 -8214 -13.06 -34.97. 1.12 -93.88 -03-3.518E-03-3148E-01-1729E-02. 34.128-04-2.5150E-02-3148E-01-3148E-01-31318E-04. 1.00 -0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	363.4 -21.08 -992.6 46.77 448.1 -18.17 -02-46.99 .2394 6.058 11.92	0
92.6 46.77 44.99 -9337. 48.1 -18.17 -9.3181103E+0.5 6.9923947550E-01-27.47 6.8923947550E-01-27.47 6.81 19.928290 1271. 6.81 18.52 4.368 4407. 12.9599356796755. 6.240591356796755. 76982.2386.22 4.464E+0.5 62.243199720E-01-359.9 62.1 -82.149720E-01-359.9 62.1 -82.149720E-01-359.9 62.1 -82.149720E-01-359.9 62.1 -82.149720E-01-359.9 62.1 -59.88 27.04 .1402E+0.5 62.1 -59.88 27.04 .1402E-0.3-2549E-0.4 64.07 -1.1741315 -1.168 1.779 62.1 -2.258E-04-2.374E-04-2136E-03-2549E-04 64.03 -3823E-04-2812E-04-4661E-03-5543E-04 62.98 -0.3 -3832E-05-1227E-04-1110E-03-1323E-04-68299E-04-8333E-05-1227E-04-1110E-03-1323E-04-6829E-03-3543E-04-3682E-04-3682E-04-3682E-04-3693E-03-3543E-04-3682E-04-36	292.6 46.77 448.1 -18.17 2-02-46.99 .2394 6.058 11.92	8-0-8
48.1 -18.17 -9.3181103E+05 6.99 .23947650E-01-27.47 11.92 .6290 1271. 12.9 .7.213 7.310 -2649. 12.9 .7.213 7.310 -2649. 12.9 .2056E-02-8533E-02-100.2 76959895679 -6755. 13079056E-02-833E-02-100.2 76921049720E-01-359.9 1.1282149720E-01-359.9 1.1282149720E-01-359.9 1.1282149720E-01-369.6 1.1282149720E-01-3497. 1.1282149720E-01-369.6 1.12821493.52 -7.016 5.851 1.12821493.63E-02-3801E-01-1129E-02 1.1299.88 27.04 .1402E+05 1.29E-03-455E-02 .1807E-02-3801E-01-1129E-02 1.29E-03-455E-02 .1807E-02-3801E-01-3109E-04 1.000 0 0 1.000 0	E-02-46.99 .2394 .6.058 11.92	E-01
6.99	E-02-46.99 .2394 .6.058 11.92	E-0
31.4 18.52 4.368 4407.  12.9 -7.213 7.310 -2649.  0.4059995679 -6755.  30379056E-028532E-02-100.2  76982.23 -86.22 .4164E+05  4.67 -2.4105184 -8096.  2.2024319 .427 -9720E-01-359.9  1.128214 .427 -3.420 -1.1243E-03  3705 4.127 93.52 -7.016 5.851  0.0 0.0 0.0  1716E-055150E-031651E-023580E-011243E-03  1716E-055150E-031651E-023580E-013407E-03  3412E-042613E-031651E-023580E-013407E-03  3412E-042613E-033951E-038455E-02  5750 -1.1743351E-042374E-042136E-035543E-04-  5750 -1.1743318E-044661E-035543E-04-  3508E-033295E-044885E-044661E-035249E-04-  3508E-033295E-044885E-044899E-035234E-04-  2203E-013050E-024161E-035481  2203E-013050E-024161E-035481  3351E-013959E-01161E-027522E-034807  3351E-013959E-014685E-034807  3351E-013959E-034807  3351E-013950E-024161E-035481	6.058 11.92	•
31.4 18.52 4.368 4407. 12.9 -7.213 7.310 -2649. 0.40598956795679. 0.40598956795679. 30379056E-028533E-02-100.2. 76982.410 -2.184 -0096. 33010749720E-01-359.9 1.128214 -1.306 -34.97. 2.2024319 27.04 .1402E+05 0.0 -0.0 -0.3283E-038258E-032190E-011243E-03- 1716E-055150E-031651E-0235801E-011129E-03- 1716E-055150E-031651E-0235801E-011129E-03- 1716E-055150E-031651E-0235801E-011129E-03- 1716E-055150E-031651E-0235801E-011129E-03- 1716E-055150E-031651E-0235801E-011129E-03- 1716E-055150E-031651E-0235801E-011129E-03- 1716E-055150E-031651E-0235801E-011129E-032549E-04- 1298E-033283E-045131E-042136E-035543E-04- 15000 -0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	_131 h 10 K2	
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5.259 132.6 -185.6 -185.6 -185.6 -185.6 1.900 29.07 1.900 29.07 1.900 29.07 1.900 29.07 1.900 29.07 1.900 25.08 -78.89 10-01-35.06 4.533 -2.441 -2.441 -3.540 -1.442 -1.442 -1.442 -1.442 -1.442 -1.442 -1.442 -1.442 -1.442 -1.442 -1.442 -1.442 -1.442 -1.442 -1.442 -1.442 -1.443 -1.752 -1.6470-01 -36.170-01 -36.170-01 -36.170-01 -36.170-01 -36.170-01 -36.170-01 -36.170-01	GROUP 4 POINT	9-8 -558.0 -8 77.6 57.68 1	140.2 -9.689 20.72 37.16 -45.07 -2.936	-414.5 1.468 .6562b-01 251.2 01 1.312	12
	HODEL-1.8/40K, FLA=83.	5.259 4 132.6 -1 5 -185.6 -9	146.4 10-0127290-01 1.014 1.900 29.07 50-01.1037 30-03.15080-02	5.08 8.89 .506 5.72 6409D-01	5.0441 1.752 1.752 3.017 5.017 5.017 5.017 5.017 1.647D-01 2.2842D-01 2.2842D-01 2.3647D-01 3.3673 -48.27 -48.27 1.725 1.725 1.725 1.725

2378E-01 2378E-01 388 2378E-01 3787E-01 3787E-01 3787E-01 3895E-01 2595E-02 2595E-02 2595E-02 2595E-04 2595E-04 2595E-04 2596E-04 2596E-04 2596E-04 2038E-05 2032E-05 2032E-05 2033E-05 2033E-05 2033E-05	37 100 47 00 3 64	1 200	1.202 13.72	256.0 43.80 31.94 25.28+05	1740 - 3783 -	179.8 5.893 -2.037	17.09 -4.035	-22.07 -9.422 -	-1.4036251	-01-2074E-01-9157E-02	-203.7	6.582 -1.118 .2744 -	.2972 4939E-01 .1244E-01	78.83 1.339 -	.7166 1930 .2608	16.55 -58.23 26.50	PIX5953 8.176 133.7 32.60 7.67744451663 1 .43678-0222388-0273618-0175908-0218148-0295248-04		.4328E-042427E-031518E-034990E-025142E-031236E-03-	.2839E-033929E-02 .1878E-03 .6179E-022045E-02	4 .1018E-041312E-032894E-039520E-021530E-03 .4897E-041272E-04	**************************************	-5861E-045592E-035853R-047769E-046992R-038827R-042791E-031103E-03 -0.0.0.0.1.000 -0.130E-053824E-043969E-055505E-054954E-046317E-051923E-047818E-05	.5082E-US .4/71E-U4 .4979E-US .6895E-US .6145E-U4 .7774E-US .2391E-U4 .9792E-US7917E-US .4771E-U4 .9792E-US7917E-U4 .9792E-US7477E-U4 .9792E-US7477E-U4 .9347E-U4 .9347E-U4 .9347E-U4 .9792E-US2033E-U57410E-U47477E-U5 .9827E-U52033E-U51904E-U481924E-U52695E-U52425E-U43188E-U5 .9179E-U53827E-U524033E-U53033E-U53180E	THE D HATRIX3447
	THE B MATRIX	0.740	0.001	256.0	23781-01-63.57	4326E-01 179.8		.3817E-01-75.92	.7334E-01-3.296	124	-684.1	.5895E-01 6.582					THE C MATRIX 41355953 1549E-01 .4367E-0	0	.5290E-04 .4328E-0				1.939 045592E-0 .0 053824E-0	057410E-0 057410E-0 051904E-0	THE D HATRIX3447 180.91040E-02 1.5350 .0 .0108E-04 .1708E-0

HODEL-2.5/65K	K, PLA=83, B	INX HN,	GROUP 4	POINT 8-2/	12/76 PAA	
5	6.702	458.8	-484.3	-4.796	+094-	.1093
3	109.7	-239.9	-18.21	8.043	-1.131	8014
.318	-152.2	27.52	62.75	-18.26	3.873	1.723
74	103.0	0.607-	-214.6	-4.637	-3.061	-2.452
010	-02 .1606	2.187	-8.300	-4820D-C	147110-02	-2008D-02
92	2.407	4.586	10.46	-18.43	.6387	.7566D-01
4738	5.468	666.9	15.96	4.485	-18.49	.1154
52	66.89	-13.90	-31.68	23,90	-1.929	-20.23
.9550D	-02 .6614D-	-018493	-1.937	.3079	80790-01	964.5
1398D	-03 .1176D-	-051246D-	012841D-	-01 .4572b-(	321172b-02	.8143D-C1
.246	29.54	-94.19	-214.8	62.01	-13.10	38.55
.3326	-189.0	685.5	.5810	8864D-	0-0	
.1478D	-01-8-400	30.47	-3699D-	-013919D-C	123802D-03	
.2624	-84.57	61.83	562.8	1.134	1568	3.888
.d967	-011764D-	-01 .3058	2769.	1017	19.95	.7005n-01
945	26.04	415.0	142.3	-4.437	4.622	1.101
564	9156	2.983	5.604	-d1686.	1 .5403	.4567
.459	-1.271	1204	-1.084	1269	2227	1686
.826	1.932	.4177	3,759	.4374	.7678	.5656
12.89	.5386	-1.760	-15.87	-1.839	-3.265	-2.424
.3069D	-018423D-	9	-02 .3718D	-01 .2358b-(	10-0600-10	.54790-02
.4710			-01 -6227	-7291D-(	11 .1280	.9485D-01
98	-7035D-	-01 . 1071	0996	.1112	.1953	.1468
17.	1367	2056	-1.887	2209	3877	2853
.8721D	-01 40.03	1281D-	011153	1350D-C	112370b-01	1756D-01
6679	.5930	1	031691D-	-021980D-0	3-	2576D-03
-9.672	-49.25	-1.425	-12.82	-1.497	-2.628	-1.948
979	39.38	-50.00	1153D-	-01 .4050D-0	2 .7109D-0	.15810-0
4277	1.750	-2.000	-2.000	-2160b-0	17	•
960-1-	18.00	-2.748	-2,362	-19.69		.7166b-0
3140D	-01 .1212b-	-01 .4 100D-	02 .3690D-	-01 .4860D-C		.5621D-02
21	-1.069	1.295	11.66	24.36	29.59	-48.21

-1.103 3 -1.75 -8 -1.70 1 5.965 -3	32.56 -85.14 139.0 -376.5	-25.12 -10.24 35.13	35.31 -55.26 111.5 -19.80	3984. 1379. 1429E+05
-2482 12.02 -3707 13.45 -8895 9.449 -1756 -5.236 -2589E-02-8054 -4024 -65.17 -1783E-01-2.886 -8431E-01 17.24 -1381E-01 -9814	2482 12.02 3707 13.45 8895 9.449 1756 -5.236 2589E-028054E-0 26.13 4024 -65.17 1783E-01-2.886 8431E-01 17.24 1381E-01 9814	2.422 12.422 12.46 -16.99 6340 117.3 9.653 4372E -3888 -4349E	-6.660 -22.652 -21.68 -21.68 -2.053 -274.4 -6.093 -012706 -8.113 -02 .6756 58.04	869.3 1208. -3539. -4914. 01-72.69 -4849E+05 -5670. -251.6 -1834. 58.24
THE C HATRIX -,40911289E-01 . 01256E-041368E-032444E-04 .	-3646 -3646 1 -1116E -0 4 -5478E 3 -1355E 4 -3544E	13.01 021805E-0 048625E-0 038094E-0 041614E-0	163.1 .0.0 13.1678E-( 12.1534E-( 12.4844E-( 13.1136E-(	### C MATRIX  -40913646 13.01 163.1 175.1 4.4399885E-02 .1789  -1289E-01 .1116E-021805E-013419E-017795E-011177E-014782E-025639E-03  -0
.4632 .3973E-0 .1926E-0 .1564E-0	2.531 33510E- 41723E- 44974E- 44974E- 4167E-	5911 023450E-0 031675E-0 031675E-0 034833E-0 031132E-0	3378 35253E-( 0.0 142551E-( 147359E-( 147359E-(	.4632 2.5315911 .3378 3.044 1.747 2.276 .46993973E-033510E-023450E-035253E-034728E-025434E-039538E-037202E-030 0 1.000 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0
-1.162 -1.162 -18278-02 -0 -8838E-04- -7088E-04- -2465E-03- -5792E-04-	89.02 2.9726E- 0.04-4150E- 4-4150E- 4-3416E- 4-3416E-	-1.162	-15.71 .5040E-( .0 .2.2537E-( .2.3658E-( .2.1164E-(	15.71 4239. .5040E-01-5.876. 0. .2537E-02-3309. .3658E-02.2620. .164E-02-8691. .3993E-03-2036.

-.5799D-01 .7615p-01 -.7789D-01 -.9070b-01 -.4847D-02 02--3496D-01--1951D-01 .8872D-01 --4379b-01-.6992b-01-.6578b-01 .6924D-02 .9232D-04 -.2033D-02-.7932D-02-.3462D-02 .7039D-02-.8419D-02-.1936D-01 4.013 -. 1236 -. 1048D-01 5.989 8.943 -. 1869 -. 1426 .1592 1.245 -19.87 39.31 .3974 -.8135 1.167 2.533 -- 8905P-01-.8025 -1.432 -.2319 --4657 -1.482 . 1044D-01 -05-. 1746D-03 .1391D-03 -01-.9429D-02 GROUP 4 POINT 9-2/12/76 PAA -2.559 .1670 -1.619 -.3237 -.2122 -.2139 -.9280D-01-.1440 1.362 1-.3645D-02-20-01 .1362 -19.64 -.2064 -. 5867 -1.258 1.931 -.4562D-01-.1774 3.106 19.90 -2607D-02 .3476D-04 .4563D .1887D .4693b .8172 -19.79 24.37 2.127 52.95 .3085 -2.847 -.1151 .9150 -. 1069 . 1022 1.846 -13.61 -1.654 -19.23 27.11 .4252 .4262 -8.957 -.4901 -.6256 .14850 -02--2673D .9717b-03-.6781b 2.100 67.76 37.73 -11.03 95.66 7.809 -2.029 -7.864 57.75 T-699--.9045 .9536 -.6572 -3.462 -151.9 357.7 -88.79 -7288D-01-4.594 -17.72 -.7875 482.5 1.359 -5.146 7.552 -.9357 1-.21570-01-.13800-01-.1254 -.3493D-01-.4703D-01-.4099 .4951D-02 .1003 .66010-04 .9057b-02-.2970p -272.0 .8317 -.7537b-01-.1020 .8281b-01 .1139 .8899 -2.879 1-2.216 41.76 2.805 52.33 -605.9 1.126 199-1-5.306 606.3 26.95 55.07 -. 2041 -.5817 -. 1312 -50.06 -2.002 .4077b-02 P100 HODEL-2.15/58.5K, PLA=83, .7644D 1.728 2.392 38.89 16.66 133.0 69.80 .2866 39.78 .5962 .5893 -47.41 -146.6 -02-.4848D-02-.1285 60.10 -165.1 -.3740D-02-.1789D-01-7.339 -75.55 -.2288D-02 .9077D-02-.1376 -1.305 -1.706 1.840 -10.09 -3906b-02 -.7723b-01 .5704D-04 -.2248D-02-.2023D-0 .3372b --7327b-02-.9077b -4.278 -.2181 -.2252 9469 -- 8405D-01-4025 5.733 -1.475 1.642 -. 1231 .5195 -.8430 -.2529 -1.888 -.3447D-01-.3069 -.7568D-01-.6778 .8355b-01 .7520 -.6662 5.867 -.9272b-01-5.159 -9.948 -.4421 -4.805 THE A MATRIX -.5835D-03 --3929b-01 .3481D -6.684 -.2326 -1.847 -. 7525 2.250 .3029 .5983 .5633 -.8500 -1.928 .1705 3.259 -1.056 66.64--.6665 •6566 -9.553 -.4246 -4.379 .6350

-1435. -4527. -1301E+05 -126.9 -126.9 -197. -1845. -86.69 -8058E+05 -6706. -298.0 -3187. -26.99	-37532777 1.859 118.8 .5250E-01 2.1805600560056005610E-02 .4741E-0
20.74 -1435. -97.98 -4527. 167.2 -1301. 9.699 -112.4. -9.292 -361.9. -25.77 -1197. -10.30 1245. -2.638 -5845. -2.638 -5845. -3903E-01-86.69. -458.8 -6706. -7.912 -6706. -7.912 -6706. -7.912 -6706. -7.912 -6706.	1.859 .6211E-03 .1649E-013105 .0 .0 .2445E-04 .7060E-031188E-01 .4532E-03 .1505E-022513E-01 .4532E-03 .1505E-022513E-01 .7185E-04 .3744E-036211E-02 1.063
-43.70 -19.22 51.01 2.326 .8414E-01 3.796 .4.73 -16.18 -7725 -7725 -2.161 -2.161 -2.161 -6.44	2777 1.859 .6857E-036211E-03 .0 0 .2617E-042445E-04 .27328E-054532E-03 .7328E-054532E-03 .5421E-057185E-04 1.735 -1.663 .2592E-022861E-03 .0 1.000 .9941E-041077E-04 .9771E-041077E-04 .5154E-045698E-05 X 107.6 -8.041 .5154E-045698E-05 .0 0 .3407E-021999E-02 .1200E-012120E-03
01.4 93.4 93.4 93.4 15.1 15.1 15.1 15.1 16.0 95.2 95.2 74.2	37532777 37532777 11692-01 -68572-03-00 52982-05 -26172-04-11062-03 -73282-05-05-17382-05-05-05-05-05-05-05-05-05-05-05-05-05-
TRE B HATRIX  3011 -4 1.769 -2 2.291 4 4.350 -5 4.196E-01-3 1.689 4 1.3891 1 -4.185 -7 -1.3891 1 -4.186 -5 1.3891 1 -4.2883 -4 2.395E-01-1 4.208 4.208 2.415 -2	.375337531769E-01 .5298E-05 .1106E-04 .1006E-04 .2903E-03-1 .2903E-03-1 .2903E-04 .2357E-04 .2357E-04 .2357E-04 .2357E-04 .2357E-05 .4208E-04 .5071E-04 .5071E-04